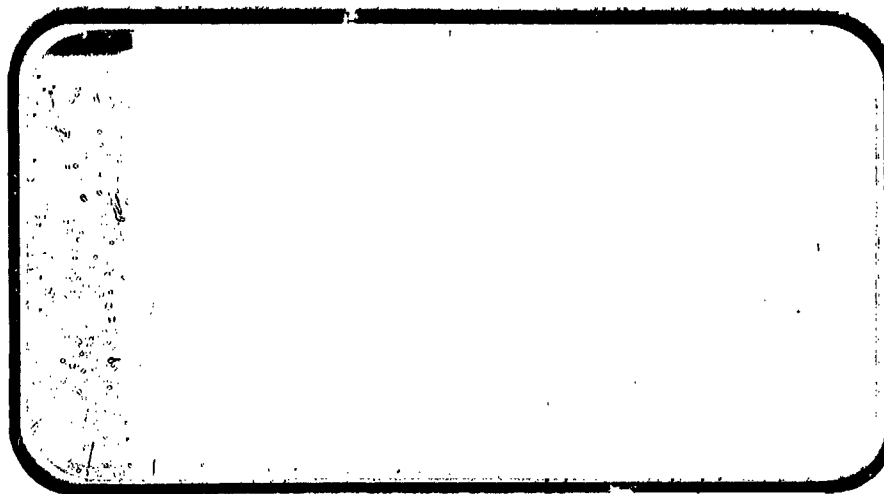


CR 134070



NASA-CR-134070) EFFECTS OF REACTION  
CONTROL SYSTEM JET SIMULATION ON THE  
STABILITY AND CONTROL CHARACTERISTICS OF  
A 0.015 SCALE SPACE SHUTTLE (Chrysler  
Corp.) 120 p HC \$9.00  
SPACE SHUTTLE

N74-20547

Unclas  
G3/31 34402

## AEROTHERMODYNAMIC DATA REPORT

JOHNSON SPACE CENTER

HOUSTON, TEXAS

DATA MANAGEMENT services

SPACE DIVISION



CHRYSLER  
CORPORATION

February, 1974

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NASA CR-134,070

EFFECTS OF REACTION CONTROL SYSTEM JET SIMULATION ON  
THE STABILITY AND CONTROL CHARACTERISTICS OF A 0.015  
SCALE SPACE SHUTTLE ORBITER MODEL TESTED IN THE  
LANGLEY RESEARCH CENTER UNITARY PLAN WIND TUNNEL  
(ØA70)

By

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Shuttle Aero Science  
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Prepared under NASA Contract Number NAS9-13247

By

Data Management Services  
Chrysler Corporation Space Division  
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center  
National Aeronautics and Space Administration  
Houston, Texas

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WIND TUNNEL TEST SPECIFICS:

Test Number: UPWT 1043  
NASA Series No.: 0A70  
Test Date: July 22 - July 27, 1973  
Model No.: 47.0

FACILITY COORDINATOR:

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EFFECTS OF REACTION CONTROL SYSTEM JET SIMULATION ON THE STABILITY AND  
CONTROL CHARACTERISTICS OF A 0.015 SCALE SPACE SHUTTLE ORBITER MODEL  
TESTED IN THE LANGLEY RESEARCH CENTER UNITARY PLAN WIND TUNNEL

By

J. J. Daileda and John Marroquin  
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ABSTRACT

An experimental investigation was performed in the Langley Research Center Unitary Plan Wind Tunnel (Test OA70) to obtain the detailed effects that RCS jet flow interactions with local orbiter flow field have on supersonic stability and control characteristics of the space shuttle orbiter. Six-component force data were obtained through an angle-of-attack range from 15 to 35 degrees at angles of sideslip of 0, +5, and -5 degrees. The test was conducted with yaw jet simulation at free-stream Mach numbers of 2.5 and 4.6, simulating SSV re-entry flight conditions at these Mach numbers. In addition to the basic force measurements, fuselage base pressures and pressures on the non-metric RCS pods were obtained. Model 42-0 was used for this test.

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Effect of RCS on Orbiter Aerodynamic Characteristics		
( $\beta = 0^\circ$ , Mach = 2.5, $\delta_e = 0^\circ$ )	A	1-2
( $\delta_e = -20^\circ$ )	A	3-4
( $\delta_e = -40^\circ$ )	A	5-6
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( $\delta_e = -20^\circ$ )	A	9-10
( $\delta_e = -40^\circ$ )	A	11-12
( $\beta = -5^\circ$ , Mach = 2.5, $\delta_e = 0^\circ$ )	A	13-14
( $\delta_e = -20^\circ$ )	A	15-16
( $\delta_e = -40^\circ$ )	A	17-18
(Mach = 4.6, $\delta_e = 0^\circ$ )	A	19-20
( $\delta_e = -20^\circ$ )	A	21-22
( $\delta_e = -40^\circ$ )	A	23-24
( $\beta = 5^\circ$ , Mach = 2.5, $\delta_e = 0^\circ$ )	A	25-26
( $\delta_e = -20^\circ$ )	A	27-28
( $\delta_e = -40^\circ$ )	A	29-30
(Mach = 4.6, $\delta_e = 0^\circ$ )	A	31-32
( $\delta_e = -20^\circ$ )	A	33-34
( $\delta_e = -40^\circ$ )	A	35-36

## SCHEDULE OF PLOTTED COEFFICIENTS:

A) CN, CLM, CA, CY, CYN, CBL vs. ALPHA



## INTRODUCTION

An experimental investigation was performed to determine interaction effects of the Reaction Control System (RCS) exhaust flow on the aerodynamic characteristics of the Space Shuttle Vehicle (SSV) orbiter. The test article was an 0.015 scale representation of the SSV orbiter configuration 3 (model 42-0). The tests were performed in the Langley Research Center Unitary Plan Wind Tunnel to simulate two re-entry trajectory points. Nominal test conditions are given below.

<u>Mach</u>	<u><math>q_{\infty}</math> PSF</u>	<u>Re/ft</u>	<u>Stagnation Temperature, °F</u>
2.5	374	$1.72 \times 10^6$	150
4.6	202	$1.72 \times 10^6$	175

Complete simulation of the RCS jet/free-stream interaction would require duplication of the trajectory free-stream conditions as well as mass flow ratio, momentum pressure, thrust and plume shape of the RCS jets. However, utilizing the Secondary Injection Momentum Principle for the injection of a jet perpendicular to the free-stream flow, only two dominant parameters significantly affect the interaction forces; jet momentum and jet pressure. Mass flow ratio and jet plume shape are considered to be less important parameters. Thus, design of the model nozzles was based entirely on matching jet to free-stream pressure ratio and momentum ratio.

RCS flow was simulated by blowing jets of cold air from non-metric nozzles attached to the model support sting in proximity to the fuselage base. Momentum ratio and pressure ratio simulation was obtained by regulating nozzle plenum pressure (as specified by the nozzle bench calibration.)

Nozzle thrust was measured using a single component strain gauge balance. The nozzle was calibrated at near vacuum conditions (because of its high expansion ratio) and corrected to total vacuum conditions. Mass flow rates were measured using a calibrated orifice meter. A plot of both measured and theoretical thrust as a function of model plenum pressure is presented in Figure 2.f. The nozzle, which simulated an RCS yaw control firing configuration, was tested in conjunction with various elevon and body flap control settings.

Six-component force data were measured on the complete model using the LaRC 1.125-inch diameter balance number 834, mounted on LaRC sting No. 77. Wind-off balance data (at tunnel operating pressure) were recorded with RCS jets on for each elevon/body flap configuration tested to determine if direct impingement effects existed.

With the tunnel flowing, data were recorded thru an angle of attack range from  $15^{\circ}$  to  $35^{\circ}$ , in  $5^{\circ}$  increments, with RCS flow either off or on for each run. An RCS-on run was made immediately after each RCS-off run, for a given configuration, to obtain an RCS increment independent of any balance shifts due to temperature.

The model had six pressure taps, located in the following positions: One tap in the RCS plenum chamber wall, three taps on the nozzle surfaces, one tap on the nozzle base, and one on the fuselage base. Pressure tap locations are shown in Figure 2.d.

Model surface flow pattern resulting from combined tunnel and RCS flows were obtained using black light oil flow techniques. Schlieren

photographs were taken during all runs with  $\delta_p = -20$  and  $\delta_p = -40$  at angles of attack of 15, 25, and 35 degrees.

Two oil-flow runs (Figures 3.d. and e.) and 62 valid force runs were made during the test period of July 22 thru 27, 1973. A summary of configurations tested and test conditions for each run is given in table II.

# NOMENCLATURE General

<u>SYMBOL</u>	<u>SADDAAC SYMBOL</u>	<u>DEFINITION</u>
$a$		speed of sound; m/sec, ft/sec
$C_p$	CP	pressure coefficient; $(p_i - p_o)/q$
$M$	MACH	Mach number; $V/a$
$p$		pressure; $N/m^2$ , psf
$q$	$Q(NCM)$ $Q(PCF)$	dynamic pressure; $1/2\rho V^2$ , $N/m^2$ , psf
$V$		velocity; m/sec, ft/sec
$\alpha$	ALPHA	angle of attack, degrees
$\beta$	BETA	angle of sideslip, degrees
$\psi$	PSI	angle of yaw, degrees
$\phi$	PHI	angle of roll, degrees
$\rho$		mass density; $kg/m^3$ , slugs/ft <sup>3</sup>

## Reference & C.G. Definitions

$A_b$		base area; $m^2$ , $ft^2$
$b$	BREF	wing span or reference span; m, ft
c.g.		center of gravity
$\frac{l}{c}$	LREF	reference length or wing mean aerodynamic chord; m, ft
$S$	SREF	wing area or reference area; $m^2$ , $ft^2$
	MRP	moment reference point
	XMRP	moment reference point on X axis
	YMRP	moment reference point on Y axis
	ZMRP	moment reference point on Z axis

## SUBSCRIPTS

b	base
l	local
s	static conditions
t	total conditions
$\infty$	free stream

# NOTATION (continued)

## Body-Axis System

SYMBOL	SYMBOL	DEFINITION
$C_H$	CH	normal force coefficient; $\frac{\text{normal force}}{qS}$
$C_A$	CA	axial force coefficient; $\frac{\text{axial force}}{qS}$
$C_Y$	CY	side force coefficient; $\frac{\text{side force}}{qS}$
$C_{A_b}$	CAB	base force coefficient; $\frac{\text{base force}}{qS}$ $-A_b(p_n - p_n)/\pi$
$C_{A_f}$	CAP	forebody axial force coefficient; $C_A - C_{A_b}$
$C_m$	CM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS l_{ref}}$
$C_n$	CYN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS b}$
$C_l$	CL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS b}$

## Stability-Axis System

$C_L$	CL	lift coefficient; $\frac{L}{qS}$
$C_D$	CD	drag coefficient; $\frac{\text{drag}}{qS}$
$C_{D_0}$	CDB	base-drag coefficient; $\frac{\text{base drag}}{qS}$
$C_{D_f}$	CDP	forebody drag coefficient; $C_D - C_{D_0}$
$C_Y$	CY	side-force coefficient; $\frac{\text{side force}}{qS}$
$C_m$	CM	pitching-moment coefficient; $\frac{\text{pitching moment}}{qS l_{ref}}$
$C_n$	CIN	yawing-moment coefficient; $\frac{\text{yawing moment}}{qS b}$
$C_l$	CL	rolling-moment coefficient; $\frac{\text{rolling moment}}{qS b}$
$L/D$	L/D	lift-to-drag ratio; $C_L/C_D$
$L/D_f$	L/DF	lift to forebody drag ratio; $C_L/C_{D_f}$

# NOMENCLATURE (CONTINUED)

## ADDITIONAL NOMENCLATURE

<u>SYMBOL</u>	<u>DATAMAN SYMBOL</u>	<u>DEFINITION</u>
$A_{bm}$		OMS pod base area, in <sup>2</sup>
$e_B$		orbiter reference body length, in.
$P_J$	PO-JET	nozzle plenum chamber pressure, psia
$P_{static}$		static or ambient pressure, psia
$Re/ft$	RN/L	unit Reynolds number, per foot
$X_{cp}/c_B$	XCP/L	longitudinal center of pressure location, fraction of body length
$\Delta AF$		incremental axial force due to model RCS flow direct impingement on metric model components, lbs.
$\Delta NF$		incremental normal force due to model RCS flow direct impingement on metric model compo- nents, lbs.
$\Delta PM$		incremental pitching moment due to model RCS flow direct impingement on metric model com- ponents, in-lb.
$\Delta RM$		incremental rolling moment due to model RCS flow direct impingement on metric model compo- nents, in-lb.
$\Delta SF$		incremental side force due to model RCS flow direct impingement on metric model components, lbs.
$\Delta YM$		incremental yawing moment due to model RCS flow direct impingement on metric model compo- nents, in-lb.
$\delta_a$	AILRON	aileron, total aileron deflection angle, degrees, (left aileron - right aileron)/2.

# NOMENCLATURE (CONTINUED)

<u>SYMBOL</u>	<u>DATAMAN SYMBOL</u>	<u>DEFINITION</u>
$\delta_{BF}$	BDFLAP	body flap deflection angle, degrees
$\delta_e$	ELEVTR	elevator deflection angle, degrees
$\delta_R$	RUDDER	rudder deflection angle, degrees
$\delta_{RF}$	RUDFLR	rudder flare angle, degrees
$\delta_{bf}$	BDFLAP	flap, surface deflection angle, positive deflection, trailing edge down; degrees
$C_{p16}$	CPB	model base pressure coefficient
$C_{p17}$	CP17	right half nozzle surface pressure coefficient
$C_{p18}$	CPBM	right half nozzle base pressure coefficient
$C_{p19}$	CP19	left half nozzle horizontal surface pressure coefficient
$C_{p20}$	CP20	left half nozzle lower surface pressure coefficient
$C_{psc}$	CPSC	sting cavity pressure coefficient
$P_\infty$	PINF	freestream static pressure, psia

## CONFIGURATION INVESTIGATION

The test article (provided by Rockwell) was a 0.015 scale model (42-0) of the VL70-000139B definition of the SSV orbiter Configuration 3. A three-view drawing of the model showing the principal dimensions, photographs of the model installation in the tunnel, and the RCS hardware are shown in Figures 2 and 3.

The model was constructed of Armco 17-4 stainless steel and was comprised of the following parts: fuselage, canopy, wing and cuff, vertical tail and orbital maneuvering system (OMS) pods. Elevon deflections of  $0^\circ$ ,  $-20^\circ$  and  $-40^\circ$ , body flap deflections of  $0^\circ$  and  $-14.25^\circ$ , and a rudder with a  $40^\circ$  speed brake deflection were tested.

The RCS plenum was attached to the sting at the base of the model (termed non-metric installation); air loads acting on it and forces produced by the RCS jet were not measured by the balance. One nozzle (N<sub>19</sub>), simulating the RCS firing for yaw control was mounted on the left side of the plenum; a nozzle with a plug inserted was mounted on the right side of the plenum chamber. The yaw nozzle defined in Figure 2.a., was built and calibrated by General Dynamics Convair of San Diego. The nozzle blocks were mounted in proximity to the OMS pods as shown in Figure 2.a.

The following nomenclature was used to designate the model components:

<u>COMPONENT</u>	<u>DEFINITION</u>
B <sub>19</sub>	Near vehicle configuration 3 (139B) fuselage of the Rockwell International SSV orbiter configuration (VL70-000139B)
C <sub>7</sub>	Basic vehicle configuration 3 (139) canopy (VL70-000139)



<u>COMPONENT</u>	<u>DEFINITION</u>
E <sub>23</sub>	Eleven on vehicle configuration 3 (139B) wing (VL70-000139B)
F <sub>5</sub>	Basic vehicle configuration 3 (139) body flap (VL70-000139B)
M <sub>6</sub>	Modified OMS-RCS pod for the Rockwell International SSV configuration 3 (VL70-000139B)
N <sub>19</sub>	Twin LH yaw nozzle sized to simulate the center two prototype 3 configurations (VL70-000140A) RCS yaw engines when tunnel Mach No. equals M for prototype trajectory.
O <sub>139B</sub>	Complete orbiter configuration consisting of B C F M V R W E 19 7 5 6 7 5 107 23
R <sub>5</sub>	Basic vehicle configuration 3 (139) rudder for vertical tail (VL70-000139)
V <sub>7</sub>	Basic vehicle configuration 3 vertical tail (VL70-000139)
W <sub>107</sub>	Vehicle configuration 3 (139B) wing (VL70-000139B)

A general arrangement configuration drawing is shown in Figures 2.a.

#### RCS NOZZLE CALIBRATION

Calibration of the RCS nozzle was performed in the vacuum chamber at Convair Aerospace Division, San Diego from July 2 to July 9, 1973, to establish nozzle thrust and mass flow characteristics as a function of nozzle plenum pressure.

The nozzle assemblies were mounted on a 12 pound capacity single component strain gauge force balance in the 5-foot vacuum chamber to obtain direct measurement of thrust. The measured nozzle thrust data were corrected to total vacuum conditions.

Dial gauge pressure readings and regulator settings for selected flow rates were recorded and used as an operating guide during the tunnel test.

Approximate flow rates were set by selecting increments of RCS plenum pressure from a curve of  $P_j$  versus estimated weight flow. The actual flow rate was calculated using the measured pressure drop across an orifice plate built to ASME specifications for which the flow coefficient had been determined by prior calibration.

Calibration results for the RCS yaw nozzle ( $N_{19}$ ) are shown in Figure 2.f.

#### TEST FACILITY DESCRIPTION

The Langley Unitary Plan Wind Tunnel is under the direction of the High-Speed Aircraft Division at NASA LaRC. The tunnel is used for force, moment, pressure-distribution, and heat-transfer studies. The test medium is air. Model mounting consists of various sting arrangements with axial and lateral movement, and side-wall support. The tunnel is of the continuous-flow, asymmetric sliding-block type. There are two test sections, Nos. 1 and 2, each 4 feet square and 7 feet long. Test section No. 2 used for these tests has a Mach number range of 2.29 to 4.63. The dynamic pressure (PSF) and unit Reynolds number (per foot) range for the lower Mach number are 120 to 1260 and  $0.76 \times 10^6$  to  $5.5 \times 10^6$ , respectively. For the maximum Mach number these ranges are 95 to 905 and  $0.83 \times 10^6$  to  $7.78 \times 10^6$ , respectively. Normal operating temperature for the tunnel is 150°F, except at Mach numbers above 3.75 where it is 175°F.

## DATA REDUCTION

Force and moments measured by the orbiter internal strain gauge balance were transferred to body and stability axes and reduced to dimensionless coefficients using standard data reduction methods. Corrections applied to the data include model static weight tare, balance and sting deflection and tunnel flow inclination. No adjustments were made to axial or drag coefficients for model base drag.

The following reference dimensions and constants were used for data reduction:

<u>SYMBOL</u>	<u>DEFINITION</u>	<u>VALUE</u>
b	Span, wing	14.050 in.
$x_{CG}$	Reference C.G.	12.58 in.
$z_{CG}$	Reference C.G.	FRL ( $z = 6.00$ )
CL BAL X	Center, balance force, measured from $X_0 = 0$ , See Figure 2.a.	17.279 in
CL BAL Z	Centerline, balance	W.L. 5.85 in.
$\bar{c}$	MAC, wing	7.122 in.
$l_B$	Reference body length	19.35 in.
S	Area, wing (ref.)	0.605 ft <sup>2</sup>

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Holdaway, George H.; Polek, Thomas E; and Kemp, Joseph H. Jr.: Aerodynamic Characteristics of a Blunt Half-Cone Entry Configuration at Mach Numbers of 6.2, 7.4, and 10.4, NACA Report TMX-682.

Orbiter Configuration Control, Rockwell Drawing No. VL70-000140A.

Lines Control Aft Body and OMS Pod, Rockwell Drawing No. VL70-000094A.

Orbiter Lines, Rockwell Drawing No. VL70-000139B.

Model Assembly and Details 139 and 139B Lines SSV Orbiter, Rockwell Drawing No. SS-A-00106.

Details and Assembly Wing and Vertical 0.015-Scale SSV, Rockwell Drawing No. SS-A-00107.

3.5-Foot Hypersonic W.T. Open Throat Model Support 1.0-Inch Balance Sting, NASA/ARC Drawing No. A13911D60.

Tunnel Installation 0.015-Scale RCS Power Orbiter - Ames RC 3.5-Foot, G/D Convair Drawing No. WT-72-108101.

Assembly and Details RCS Power Orbiter Force Model 0.015-Scale, G/D Convair Drawing No. WT-72-108102.

Pretest Information for Test OA70 of the 0.015-Scale Space Shuttle Orbiter Configuration 3A in the Langley Research Center UPWT to Determine Effects of RCS Jet Flow Field Interactions on the Aerodynamic Characteristics, Rockwell International Report SD73-SH-0191, July 1973.

TABLE I

[illegible]

TABLE II.

TEST : LARC UPWT-1043

DATE :

DATA SET/RUN NUMBER COLLATION SUMMARY

DATA SET IDENTIFIER	CONFIGURATION	SCHD.		PARAMETERS/VALUES				NOMINAL RCS				PLENUM PRESSURE ~ P3 ~ P51A				TEST RUN NUMBERS
		$\alpha$	$\beta$	M	Re/ft	$\delta e$	$\delta Re$	$\delta Re$	0	64	70	164	226	415		
	B4C7F5M8V4R3W107F23N19	-	-	-	-	0	0	40°		21	21	21	21		} DIRECT IMPINGEMENT } FORCE DATA (SEE TABLE IV)	
		-	-	-	-	-20	-14.2	T		62	62	62	62			
RPV001		A	-5	25	1.72	0	0		02	03			04			
02		T	0		T	T	T		05	06			07			
03			5		T	T	T		08	09			10			
04			-5		-20	-14.2	T		53	54			55			
05			0		T	T	T		56	57			58			
06			5		T	T	T		59	60			61			
07			-5		-40	T	T		22	23			24			
08			0		T	T	T		28	29			30			
09			5		T	T	T		25	26			27			
10			-5	4.6	0	0	0		11		12	13				
11			0		T	T	T		14		15	16	17			
12			5		T	T	T		18		19	20				
13			-5		-20	-14.2	T		44		45	46				
14			0		T	T	T		47		48	49				
15			5		T	T	T		50		51	52				

75 78

67

61

55

49

43

37

31

25

19

13

7

BETA

ICN

ICA

ICL

ICM

ICN

ICP

ICQ

ICR

ICS

ICT

ICU

ICV

ICW

ICX

ICY

ICZ

ICD

ICE

ICF

ICG

ICH

ICI

ICJ

ICK

ICL

ICM

ICN

ICO

ICP

ICQ

ICR

ICS

ICT

ICU

ICV

ICW

ICX

ICY

ICZ

ICD

ICE

ICF

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ICI

ICJ

ICK

ICL

ICM

ICN

ICO

ICP

ICQ

ICR

ICS

ICT

ICU

ICV

ICW

ICX

ICY

ICZ

ICD

$\alpha$  OR  $\beta$

SCHEDULES

A:  $\alpha = 15, 20, 25, 30, 35$  DEGS.

\* COEF. FOR APPROX. THROUGH DATA IN APPENDIX  
S/REF. CPB.CPSC.CPBM.CP17.CP19.CP20.FINE

ICVAR (1)

ICVAR (2)

ICVAR (3)

ICVAR (1)

ICVAR (2)

ICVAR (3)

TABLE II. (Concluded)

[illegible]

TABLE III. MODEL DIMENSIONAL DATA

MODEL COMPONENT: BODY - B19GENERAL DESCRIPTION: Fuselage, Configuration 3, per Rockwell Lines  
VL70-000139B.NOTE: Identical to B17 except forebody.Model Scale = 0.015DRAWING NUMBER: VL70-000139B

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Length - IN.	<u>1290.3</u>	<u>19.35450</u>
Max. Width - IN.	<u>267.6</u>	<u>4.01400</u>
Max. Depth - IN.	<u>244.5</u>	<u>3.66750</u>
Fineness Ratio	<u>4.82175</u>	<u>4.82175</u>
Area - FT <sup>2</sup>		
Max. Cross-Sectional	<u>386.67</u>	<u>0.08700</u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>



TABLE III. MODEL DIMENSIONAL DATA (Cont Inued)

MODEL COMPONENT: Canopy - C7

GENERAL DESCRIPTION: Configuration 3 per Rockwell Idnos VL70-000139

Model Scale = 0.015

DRAWING NUMBER VL70-000139

<u>DIMENSION:</u>	<u>FULL SCALE</u>	<u>MODEL SCALE</u>
Length ( $X_0 = 433$ to $X_0 = 670$ ) - in. FS	<u>237</u>	<u>3.555</u>
Max Width	<u>                    </u>	<u>                    </u>
Max Depth ( $Z_0 =$ to $Z_0 = 501$ ) - in FS	<u>                    </u>	<u>                    </u>
Fineness Ratio	<u>                    </u>	<u>                    </u>
Area	<u>                    </u>	<u>                    </u>
Max Cross-Sectional	<u>                    </u>	<u>                    </u>
Planform	<u>                    </u>	<u>                    </u>
Wetted	<u>                    </u>	<u>                    </u>
Base	<u>                    </u>	<u>                    </u>

TABLE III. MODEL DIMENSIONAL DATA (Continued)

MODEL COMPONENT: ELEVON - E23GENERAL DESCRIPTION: Configuration 3 per W107 Rockwell LinenVL70-000139B, data for (1) of (2) sidesModel Scale = 0.015DRAWING NUMBER:VL70-000139BDIMENSIONS:FULL-SCALEMODEL SCALEArea - FT<sup>2</sup>205.520.04624

Span (equivalent) - IN.

353.345.30010

Inb'd equivalent chord

114.781.72170

Outb'd equivalent chord

55.000.82500Ratio movable surface chord/  
total surface chord

At Inb'd equiv. chord

.208.208

At Outb'd equiv. chord

.400.400

Sweep Back Angles, degrees

Leading Edge

0.000.00

Trailing Edge

-10.24-10.24

Hingeline

0.000.00Area Moment (Normal to hinge line) - FT<sup>3</sup>  
Product of Area Moment1548.070.34832

TABLE III. MODEL DIMENSIONAL DATA (Continued)

MODEL COMPONENT: F5 Body Flap

GENERAL DESCRIPTION: 3 Configuration per Rockwell Lines VI70-000139

Scale Model = 0.015

DRAWING NUMBER

VI70-000139

DIMENSION:

FULL SCALE

MODEL SCALE

Length - in

84.70

1.27050

Max Width - in

267.6

4.01400

Max Depth

Fineness Ratio

Area - Ft<sup>2</sup>

Max Cross-Sectional

Planform

142.5

0.03206

Wetted

Base

38.0958

0.00857

TABLE III. MODEL DIMENSIONAL DATA (Continued)

MODEL COMPONENT : NOZZLE - N19

GENERAL DESCRIPTION : Basic configuration 3A (VH70-000139B OMS Nozzles  
with Cold Jet Simulation of Yaw Control (Lateral Thrust) at Mach 10.3

Entry Condition

MODEL SCALE = 0.015

DRAWING NUMBER : \_\_\_\_\_

DIMENSIONS :	FULL SCALE	MODEL SCALE
Freestream Mach No. <u>10.3</u>		
No. of nozzles (Left Side Only)	<u>2</u>	<u>2</u>
Expansion Ratio	<u>--</u>	<u>10.81</u>
Diameter ~ in.		
Exit	Direct Scaling No Applicable	<u>0.1440</u>
Throat	<u>--</u>	<u>0.0437</u>
Area ~ IN <sup>2</sup> .		
Exit		<u>.01629</u>
Throat	<u>↓</u>	<u>.00151</u>
Thrust Centerline		
X	<u>1533.0</u>	<u>22.995</u>
Y	<u>--</u>	<u>--</u>
Z	<u>472.5</u>	<u>7.087</u>

TABLE III. MODEL DIMENSIONAL DATA (Continued)

MODEL DIMENSIONAL DATA

MODEL COMPONENT : OMS POD M<sub>6</sub>

GENERAL DESCRIPTION : BASIC CONFIGURATION 3A OMS PODS WITH DETACHED  
RCS NOZZLE

NOTE: HOUSING TO MAKE THE RCS NON METRIC.

DRAWING NUMBER : VL70-000139B

MODEL SCALE 0.015

DIMENSIONS :	FULL SCALE	MODEL SCALE
Length IN.	<u>346.0</u>	<u>5.1900</u>
Max Width	<u>108.0</u>	<u>1.620</u>
Max Depth	<u>113.0</u>	<u>1.695</u>
Fineness Ratio	<u>          </u>	<u>          </u>
Area	<u>          </u>	<u>          </u>
Max. Cross-Sectional	<u>          </u>	<u>          </u>
Planform	<u>          </u>	<u>          </u>
Wetted	<u>          </u>	<u>          </u>
Base	<u>          </u>	<u>          </u>

TABLE III. MODEL DIMENSIONAL DATA (Continued)

MODEL COMPONENT: RUDDER - R5GENERAL DESCRIPTION: 2A, 3 and 3A Configuration per Rockwell LinesVI70-000095Model Scale = 0.015DRAWING NUMBER: VI70-000095

<u>DIMENSIONS:</u>	<u>FULL-SCALE</u>	<u>MODEL SCALE</u>
Area - FT <sup>2</sup>	<u>106.38</u>	<u>0.02394</u>
Span (equivalent) - IN.	<u>201.0</u>	<u>3.01500</u>
Inb'd equivalent chord	<u>91.585</u>	<u>1.37377</u>
Outb'd equivalent chord	<u>50.833</u>	<u>0.76249</u>
Ratio movable surface chord/ total surface chord		
At Inb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
At Outb'd equiv. chord	<u>0.400</u>	<u>0.400</u>
Sweep Back Angles, degrees		
Leading Edge	<u>34.83</u>	<u>34.83</u>
Tailing Edge	<u>26.25</u>	<u>26.25</u>
Hingeline	<u>34.83</u>	<u>34.83</u>
Area Moment (Normal to hinge line)- FT <sup>3</sup>	<u>526.13</u>	<u>0.00178</u>
Product of Area and Mean Chord		

TABLE III. MODEL DIMENSIONAL DATA (Continued)

MODEL COMPONENT: VERTICAL - V 7

GENERAL DESCRIPTION: Centerline vertical tail, double-convex airfoil with rounded leading edge.

NOTE: Same as V5, but with manipulator housing removed.

Model Scale = 0.015

DRAWING NUMBER:

VI.70-000139

## DIMENSIONS:

## FULL-SCALE

## MODEL SCALE

## TOTAL DATA

Area (Theo) $\text{Ft}^2$	425.92	0.09583
Planform		
Span (Theo) In	315.72	4.7358
Aspect Ratio	1.675	1.675
Rate of Taper	0.507	0.507
Taper Ratio	0.404	0.404
Sweep Back Angles, degrees		
Leading Edge	45.000	45.000
Trailing Edge	26.249	26.249
0.25 Element Line	41.130	41.130
Chords:		
Root (Theo) WP	268.50	4.02750
Tip (Theo) WP	108.47	1.62705
MAC	199.81	2.99715
Fus. Sta. of .25 MAC	1463.50	21.95250
W. P. of .25 MAC	635.522	9.53283
B. L. of .25 MAC	0.00	0.00
Airfoil Section		
Leading Wedge Angle Deg	10.000	10.000
Trailing Wedge Angle Deg	14.920	14.920
Leading Edge Radius	2.0	0.030
Void Area - $\text{Ft}^2$	13.17	0.00296
Blanketed Area	0.00	0.00

TABLE III. MODEL DIMENSIONAL DATA (Concluded)

MODEL COMPONENT: WING-W 107GENERAL DESCRIPTION: Configuration 3 per Rockwell Lines V170-0001391NOTE: Same as W103, except cuff, airfoil and incidence angle.

Model Scale = 0.015

TEST NO.DWG. NO. V170-0001392DIMENSIONS:FULL-SCALEMODEL SCALETOTAL DATAArea (Theo.)  $\text{Ft}^2$ 

Planform

2690.00

0.60525

Span (Theo) In.

936.68

14.05020

Aspect Ratio

2.265

2.265

Rate of Taper

1.177

1.177

Taper Ratio

0.290

0.290

Dihedral Angle, degrees (@ TE of Elevon)

3.500

3.500

Incidence Angle, degrees

0.500

0.500

Aerodynamic Twist, degrees

+3.000

+3.000

Sweep Back Angles, degrees

45.000

45.000

Leading Edge

-10.24

-10.24

Trailing Edge

0.25 Element Line

35.209

35.209

## Chords:

Root (Theo) B.P.O.O.

689.24

10.33860

Tip, (Theo) B.P.

137.85

2.06775

MAC

474.81

7.12215

Fus. Sta. of .25 MAC

1136.89

17.05335

W.P. of .25 MAC

299.20

4.48800

B.L. of .25 MAC

182.13

2.73195

EXPOSED DATAArea (Theo)  $\text{Ft}^2$ 

1752.29

0.39426

Span, (Theo) In. BP108

720.68

10.81020

Aspect Ratio

2.058

2.058

Taper Ratio

0.2451

0.2451

## Chords

Root BP108

562.40

8.43600

Tip 1.00  $\frac{b}{2}$ 

137.85

2.06775

MAC

393.03

5.89545

Fus. Sta. of .25 MAC

1185.31

17.77965

W.P. of .25 MAC

300.20

4.50300

B.L. of .25 MAC

251.76

3.77640

Airfoil Section (Rockwell Mod NASA)  
XXXX-64Root  $\frac{b}{2}$  =

0.10

0.10

Tip  $\frac{b}{2}$  =

0.12

0.12

Data for (1) of (2) Sides

Leading Edge Cuff  $\text{Ft}^2$ 

118.333

0.02662

Planform Area

500

7.5000

Leading Edge Intersects Fus M. L. @ Sta

1083.4

16.2510

Leading Edge Intersects Wing @ Sta

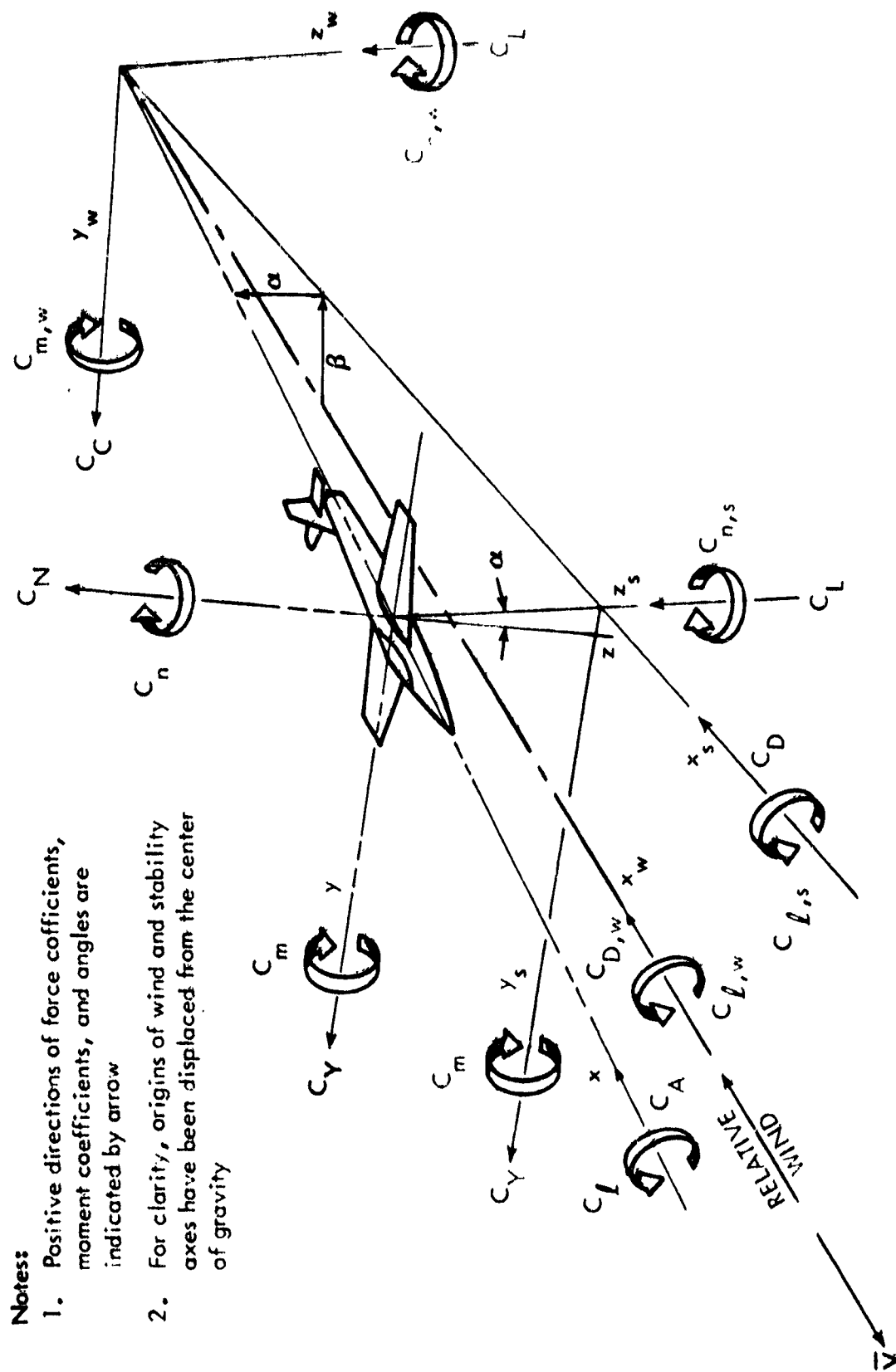


Table IV RCS Direct Impingement Force Data

RUN NO.	P <sub>t</sub> (PSIA)	ΔWF (LB)	ΔAF (IN-LB)	ΔPM (IN-LB)	ΔPM (IN-LB)	ΔYK (IN-LB)	ΔSF (IN-LB)	P <sub>STATIC</sub> (PSIA)
1	224.3	0.0	0.0	0.0	0.0	0.0	- 0.20	233.3
	65.0	0.0	0.0	0.0	0.0	0.0	0.0	234.5
	166.8	0.0	0.0	0.0	0.0	0.0	0.0	235.4
	71.7	0.0	0.0	0.0	0.0	0.0	0.0	235.8
21	64.5	0.37	0.0	0.0	0.0	0.0	0.0	371.9
	70.1	0.37	0.0	0.0	0.0	0.0	0.0	364.8
	164.8	0.37	0.0	0.0	0.43	0.0	0.0	357.5
	227.5	0.37	0.0	0.0	0.43	0.0	0.0	355.6
	402.1	0.37	0.0	0.0	0.43	0.0	0.0	353.0
43	62.5	0.0	- 0.04	0.91	0.0	0.0	0.0	245.3
	73.7	0.0	- 0.04	0.0	0.0	0.43	0.0	241.1
	164.3	0.0	- 0.04	0.0	0.0	0.43	0.0	240.5
	225.6	0.0	- 0.04	0.0	0.0	0.43	0.0	240.5
62	63.8	0.0	0.0	0.0	0.0	0.0	0.0	229.2
	70.1	0.0	0.0	0.0	0.0	0.0	0.0	230.0
	164.8	0.0	0.0	0.0	0.0	0.0	0.0	231.5
	226.3	0.0	0.0	0.0	0.0	0.0	0.0	232.5

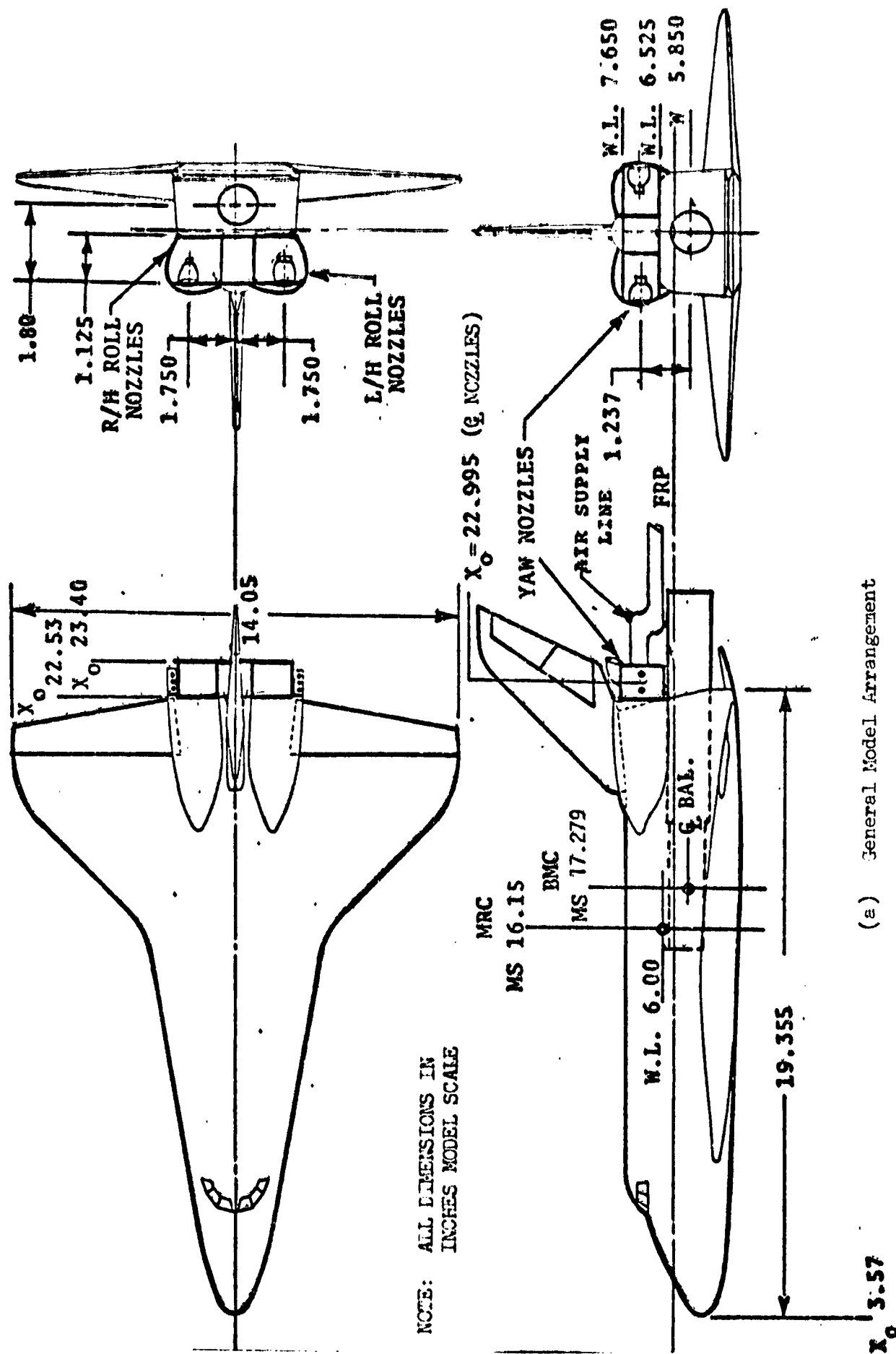
NOTE: INCREMENTAL BALANCE QUANTITIES (ΔNA, ΔAF, ETC.) ARE EQUAL TO (VALUE, AIR ON) - (VALUE, AIR OFF).  
THE MOMENTS ARE REFERRED TO THE BALANCE MOMENT CENTER.

MODEL FIGURES



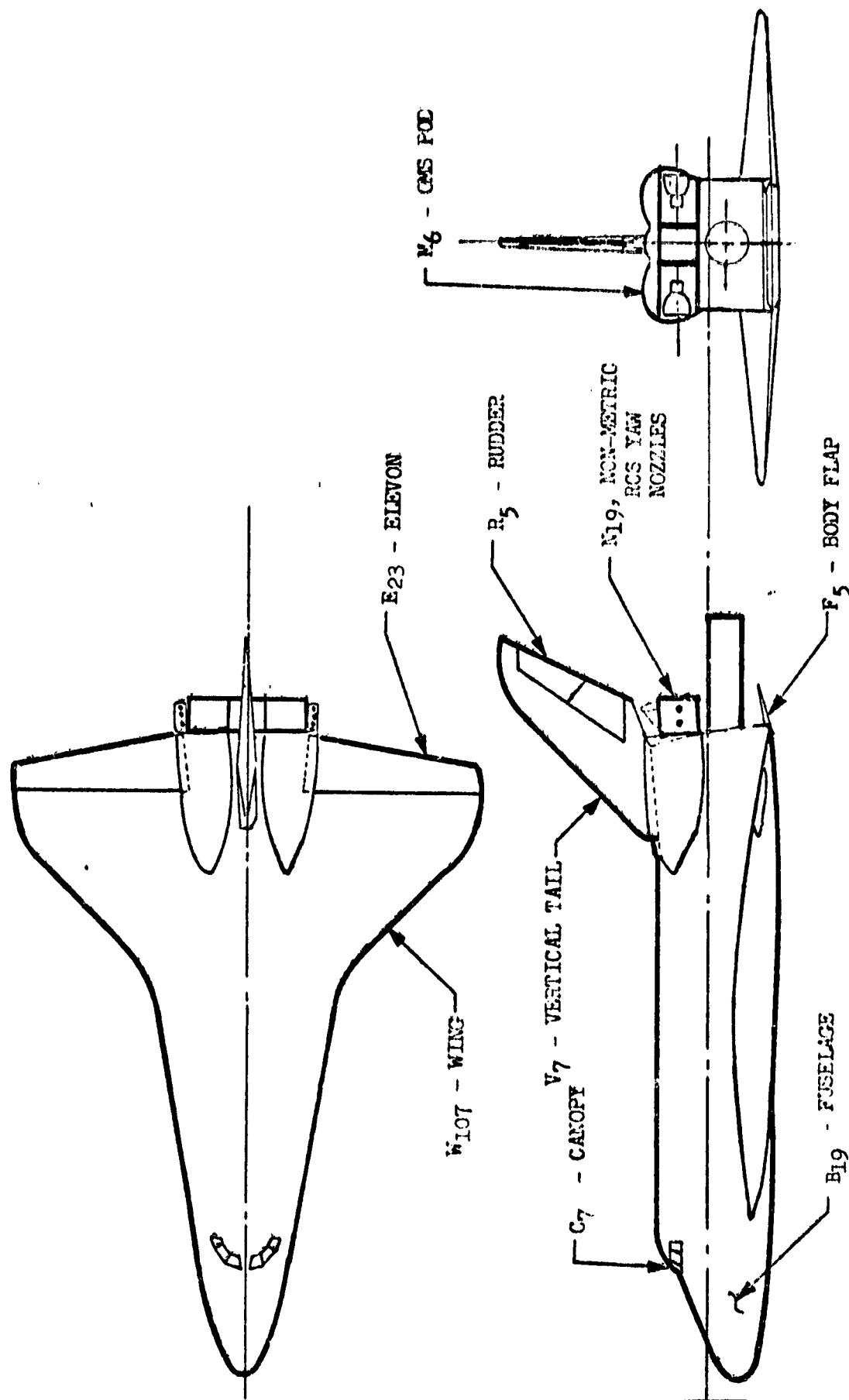
- Notes:**
1. Positive directions of force coefficients, moment coefficients, and angles are indicated by arrow
  2. For clarity, origins of wind and stability axes have been displaced from the center of gravity

Figure 1. - Axis Systems.



(a) General Model Arrangement

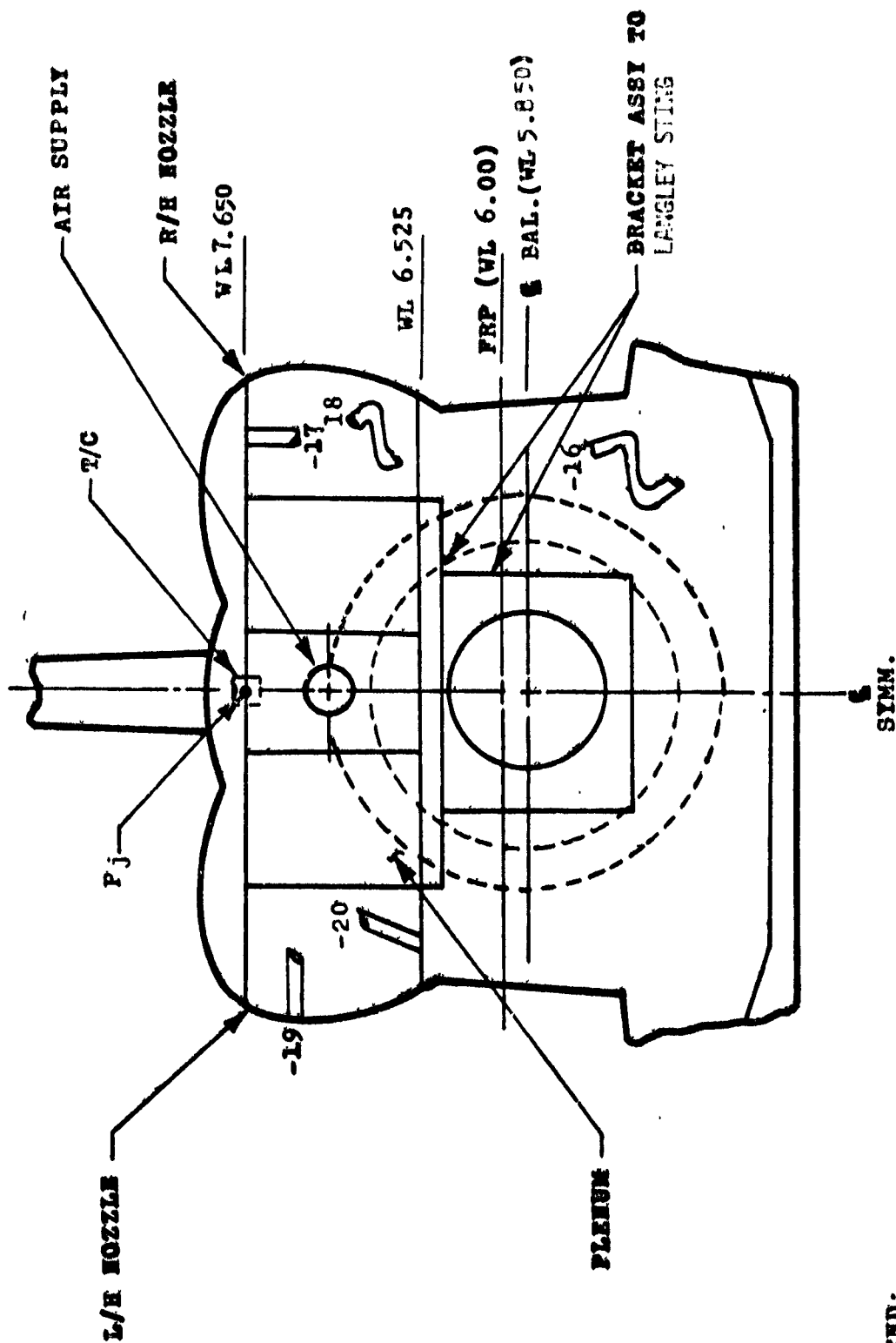
Figure 2. - Model Information.



(b) SSV Orbiter VL70000139 Model Nomenclature

Figure 2. - Continued.





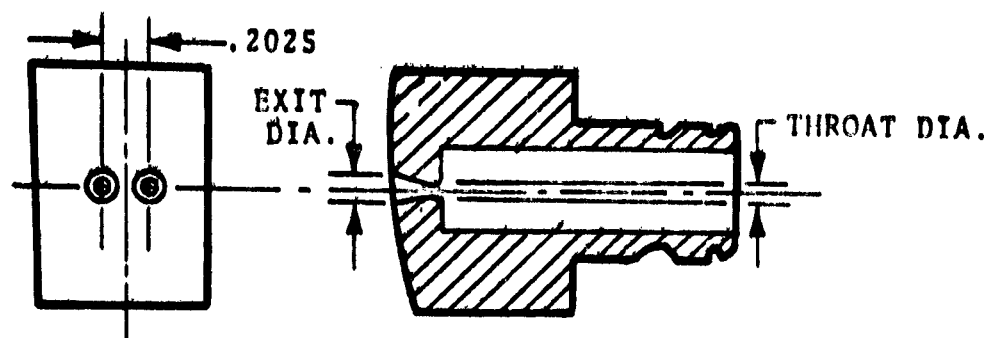
**LEGEND:**

- 16 MODEL BASE PRESSURE
- 17 R/H NOZZLE SURFACE PRESSURE
- 18 R/H NOZZLE BASE PRESSURE
- 19 L/H NOZZLE HORIZONTAL SURFACE PRESSURE
- 20 L/H NOZZLE LOWER SURFACE PRESSURE

P<sub>j</sub> RCS PLENUM PRESSURE

(d) Arrangement of Fuselage and RCS Plenum Base Pressures

Figure 2. - Continued.



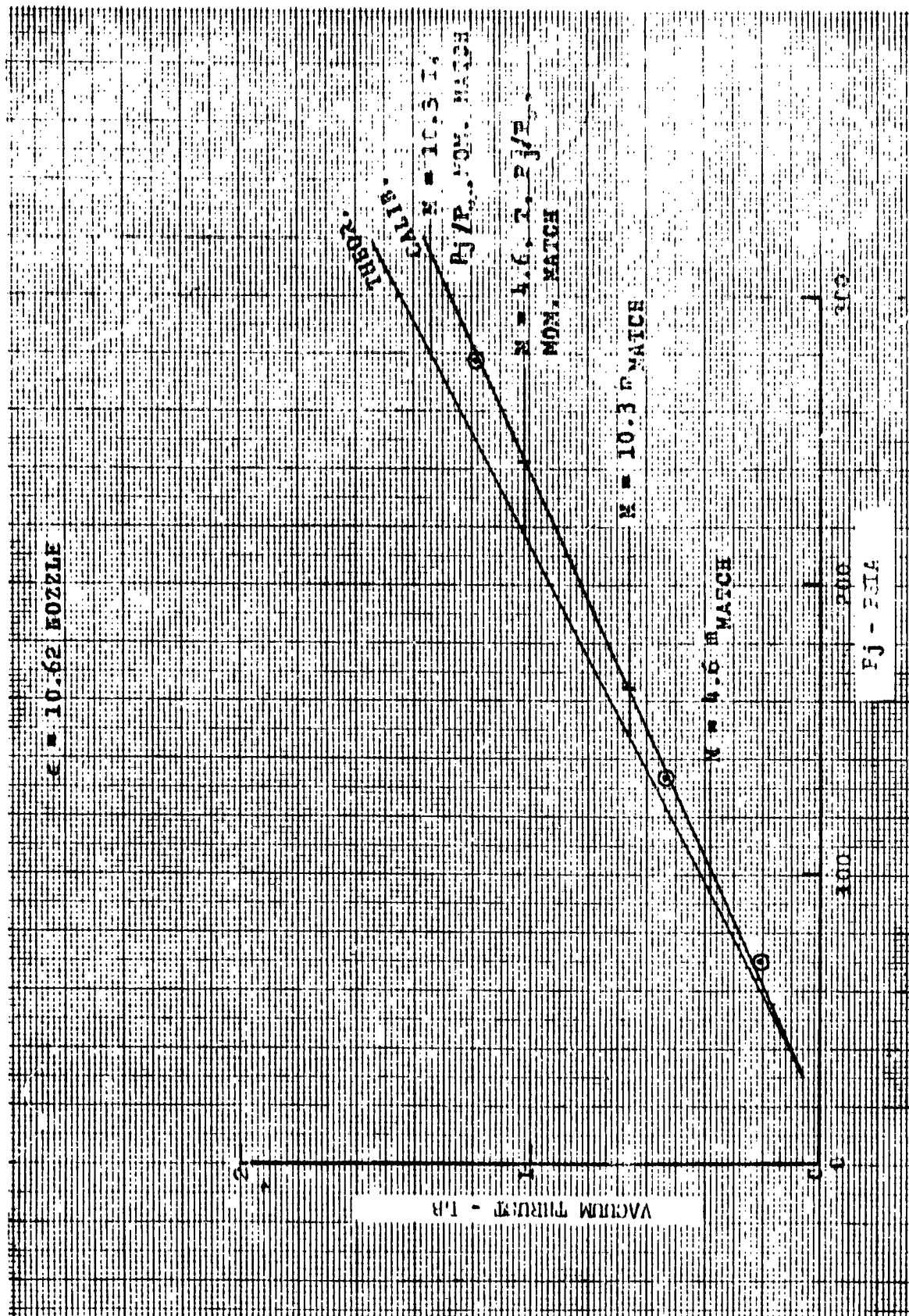
- NOTES:
- ALL DIMENSIONS IN INCHES MODEL SCALE
  - TWIN NOZZLE CONFIGURATION
  - LEFT-HAND NOZZLE FIRING ONLY  
(FOR YAW CONTROL SIMULATION)
  - THROAT DIA. = 0.0437 IN.  
     THROAT AREA = 0.00151 IN<sup>2</sup>  
     EXIT DIA. = 0.1440 IN.  
     EXIT AREA = 0.01629 IN<sup>2</sup>  
     EXPANSION RATIO = 10.81
  - PROVIDES SIMILITUDE WITH ORBITER  
     YAW RCS PLUME GEOMETRY AT MACH 10.3  
     ENTRY CONDITION
  - DESIGNATED AS N<sub>19</sub>

FIGURE 2. (Cont'd)

e) RCS Nozzle Details

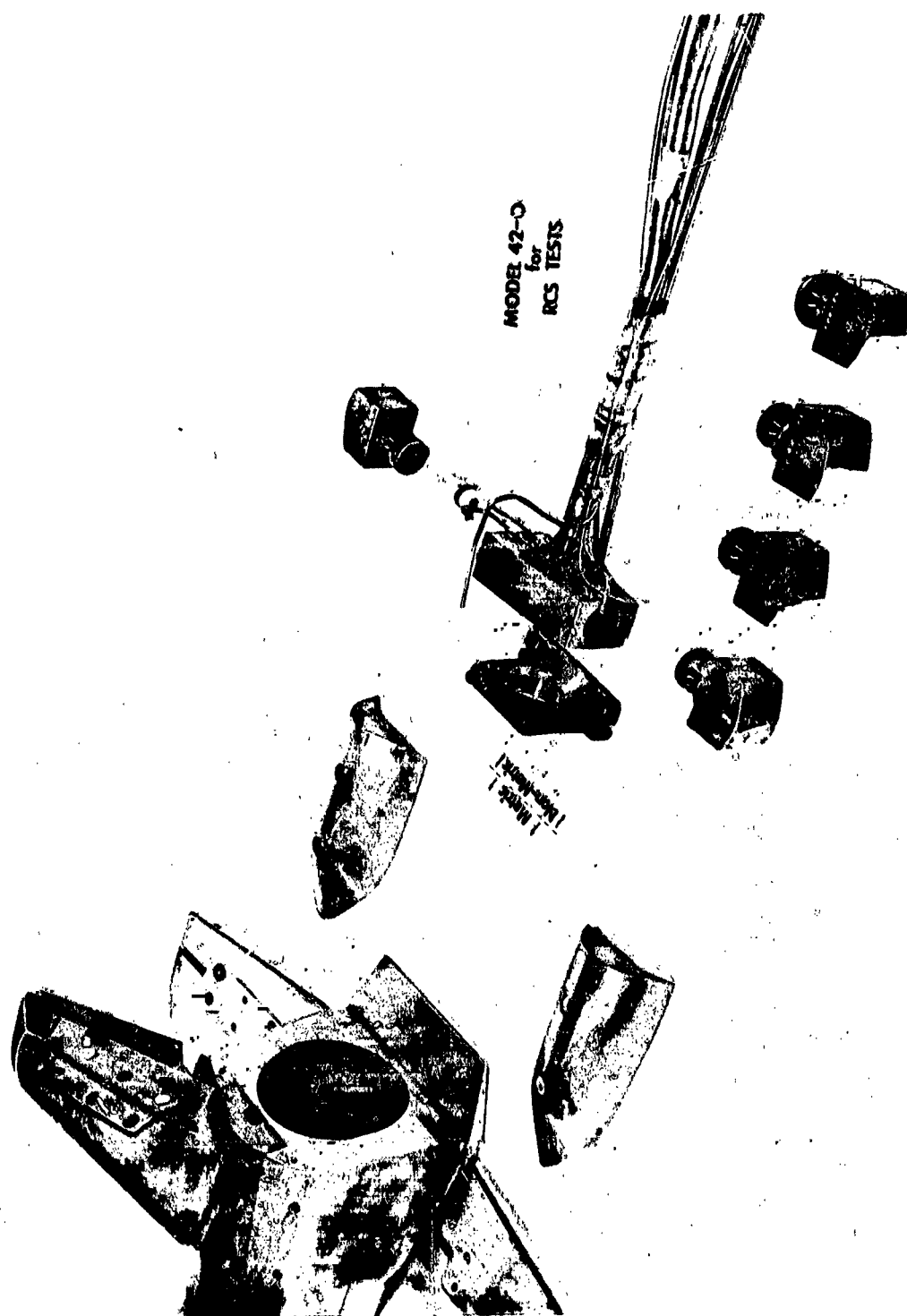


REPRODUCIBILITY OF THE ORIGINAL PAGE IS POOR.

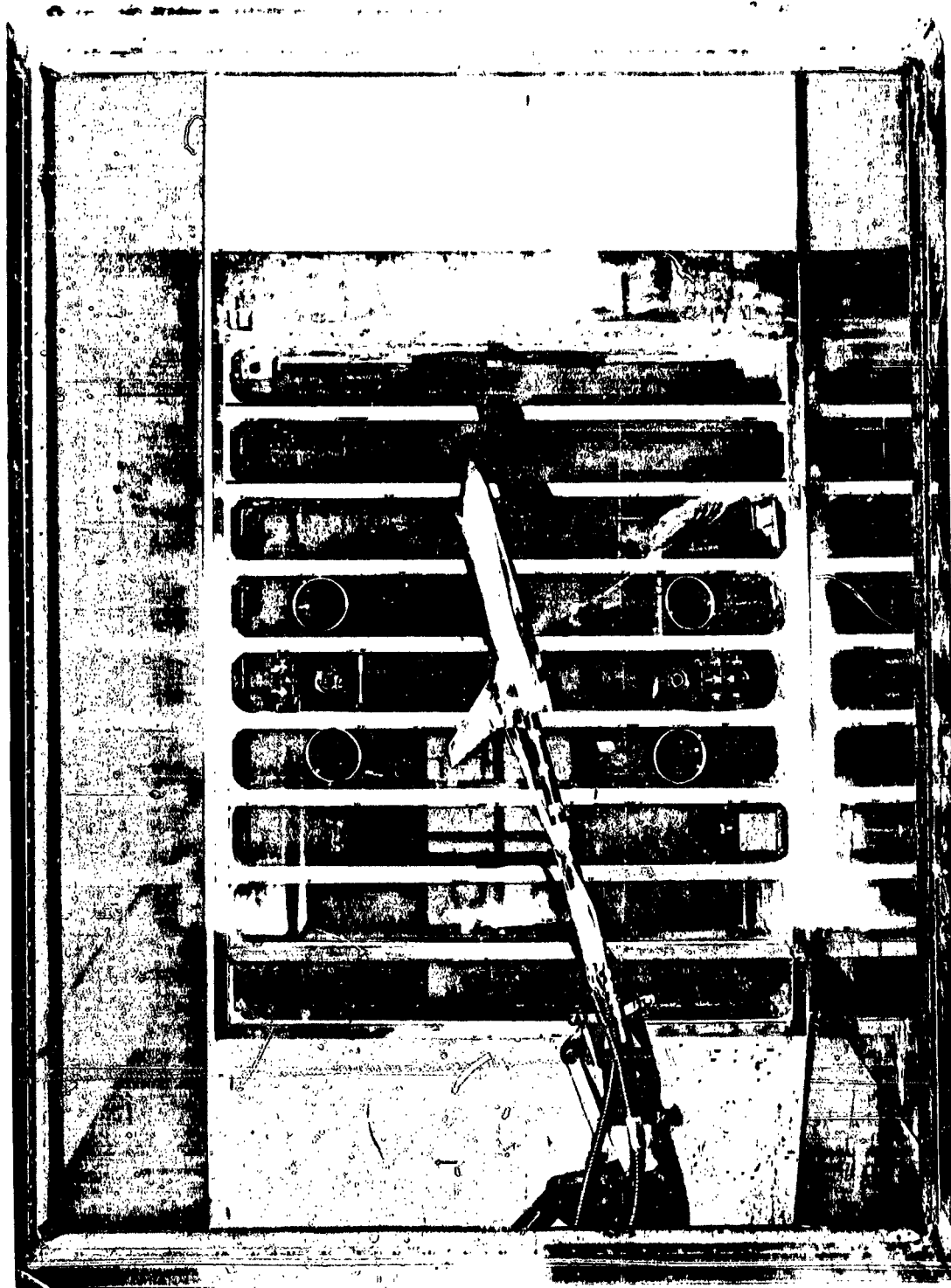


(f) K<sub>19</sub> Nozzle Calibration

Figure 2. - Concluded.

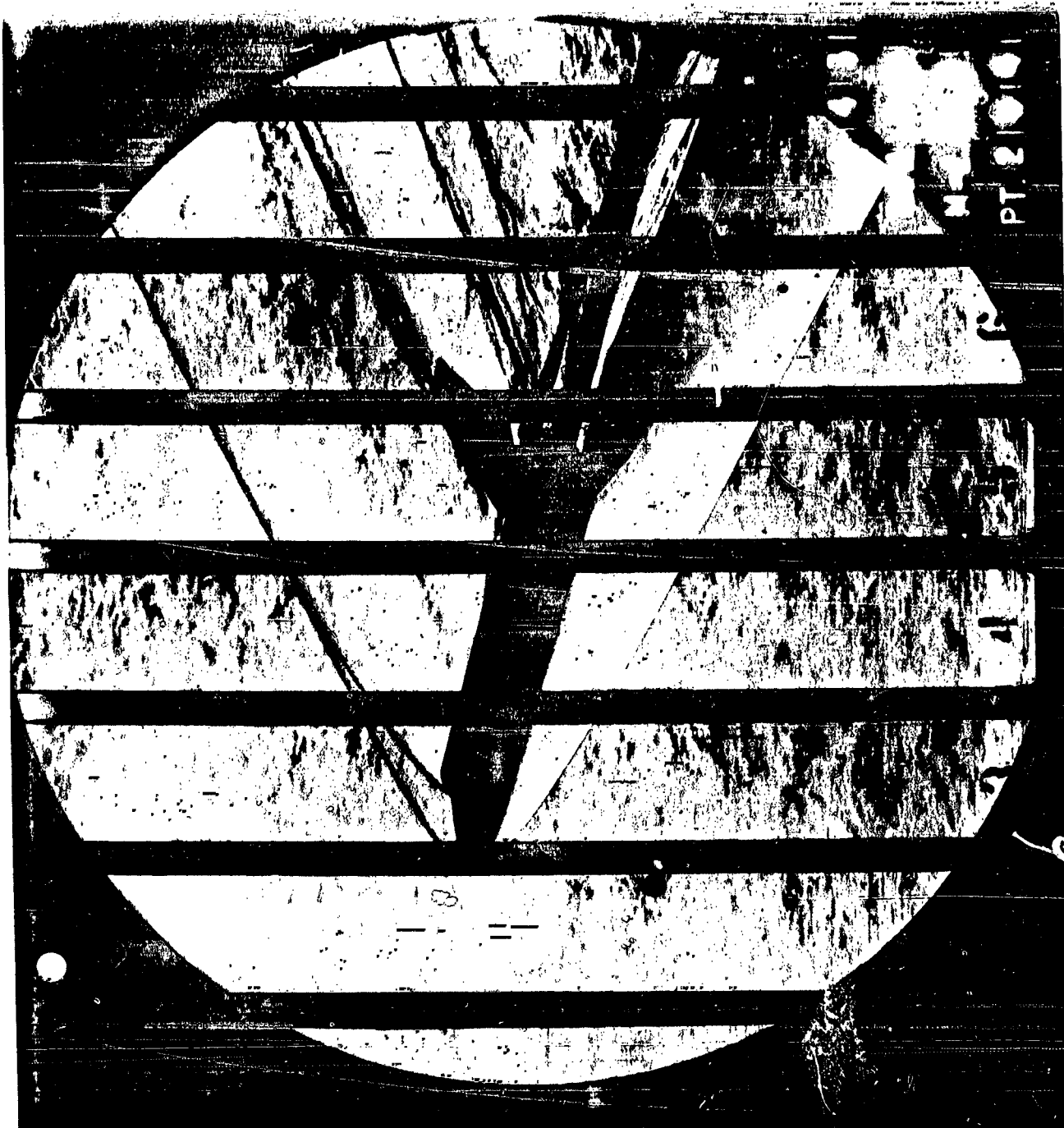


(a) Model RSC Nozzle Hardware  
Figure 3. - Model Photographs.



(b) General Installation Photograph

Figure 3. - Continued.



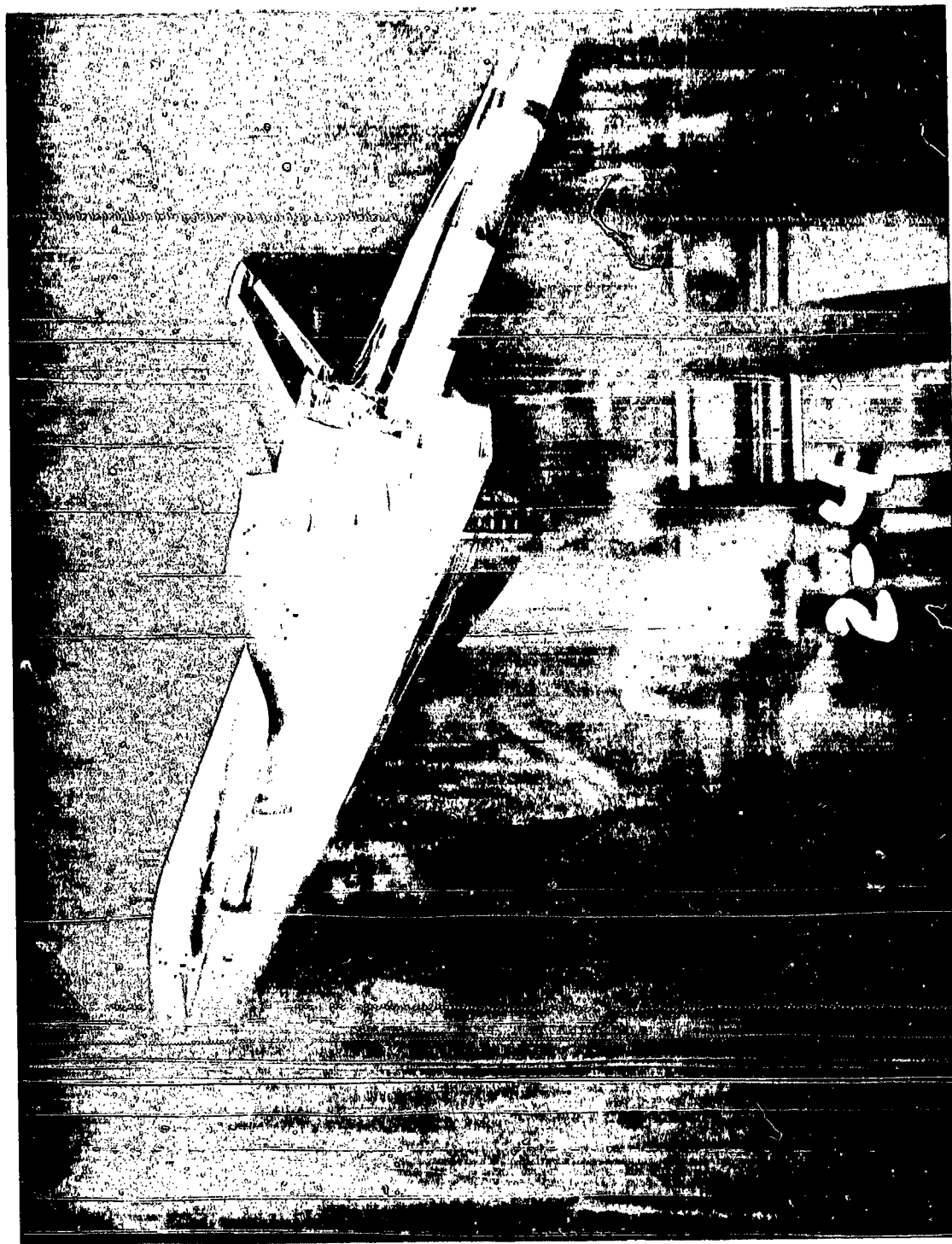
(c) Typical Schlieren Photograph

Figure 3. - Continued.



(a) Typical Oil-Flow Photograph (RCS Operating)

Figure 3. - Continued.



(e) Typical Oil-Flow Photograph (RCS Not Operating)

Figure 3. - Concluded.

DATA FIGURES



0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV00C2)

SYMBOL  
O  
◇

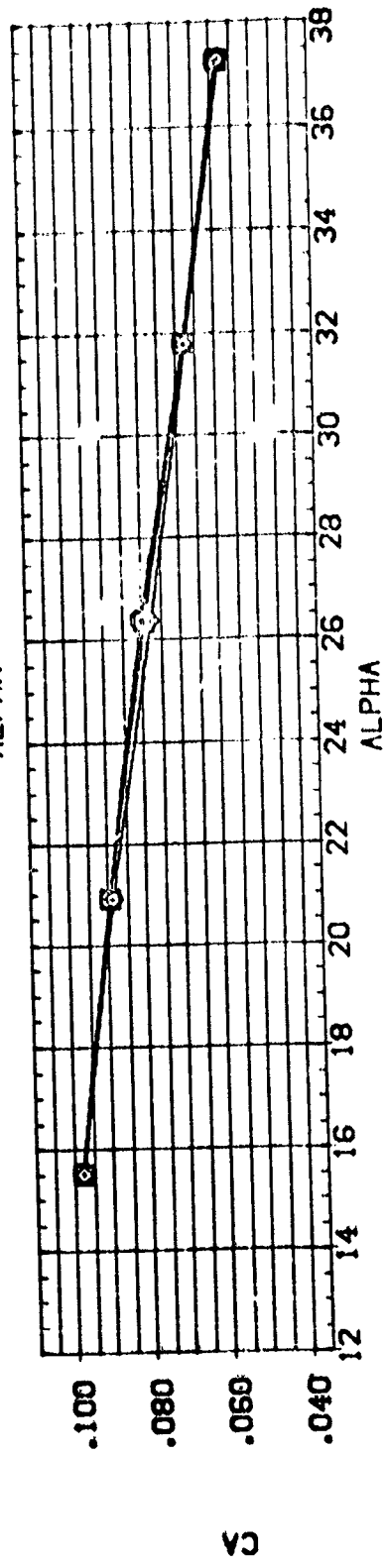
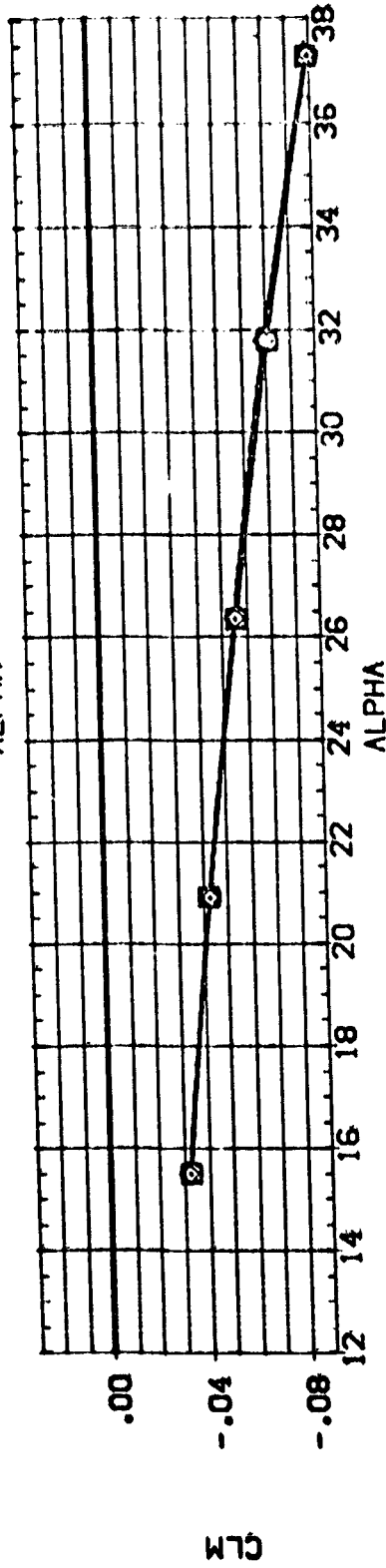
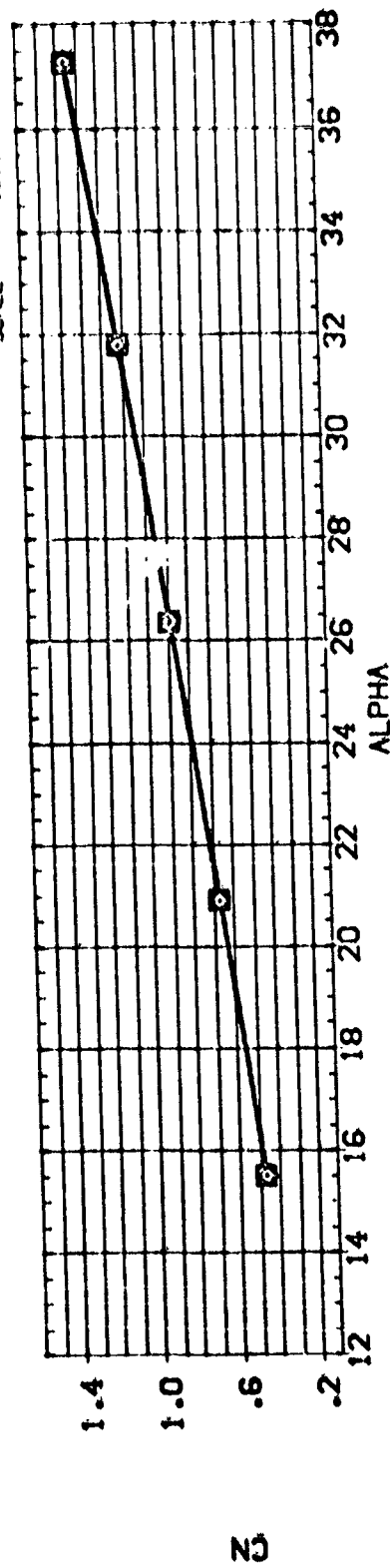
PG-JET  
.092  
69.464  
223.200

BETA  
RNVL  
AILRON  
P.OFLR

PARAMETRIC VALUES  
.000 MACH  
1.720 ELEVTR  
.000 BOFLAP  
40.000 R.OOER

2.500  
.000  
.000  
.000

REFERENCE INFORMATION  
SREF 07.1560  
LREF 7.1222  
BREF 14.0500  
XREF 12.5800  
YREF 0.0000  
ZREF 6.0000  
SCALE 0.150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=2.5)

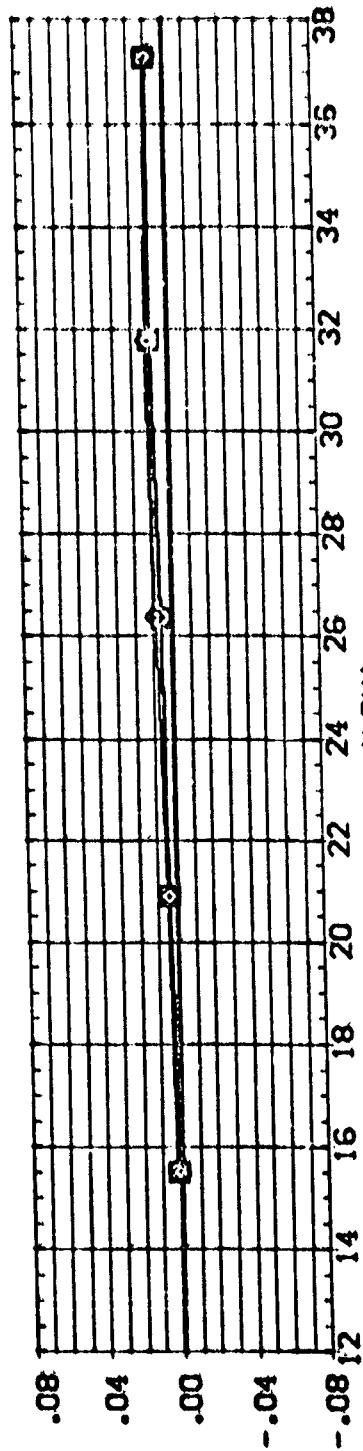


CA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV002)

SYMBOL  
 ○ □ ◇

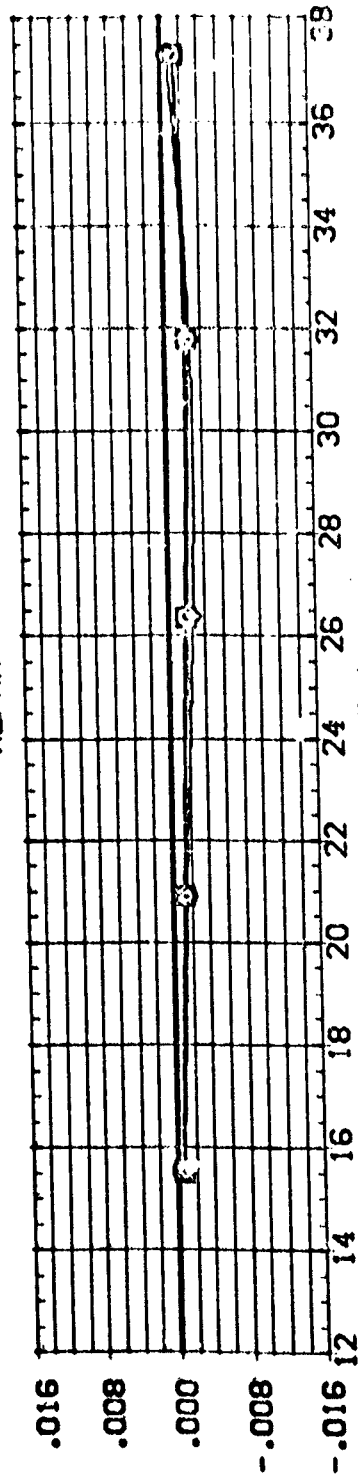
PG-JET BETA PVAL AILRON ROLF R  
 .092 .000 MACH 2.500  
 69.464 1.720 ELEVTR .000  
 223.200 .000 BFLAP .000  
 40.000 RUDDER .000

REFERENCE INFORMATION  
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 BREF 14.0500  
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 YREF .0000  
 ZREF 6.0000  
 SCALE .0150



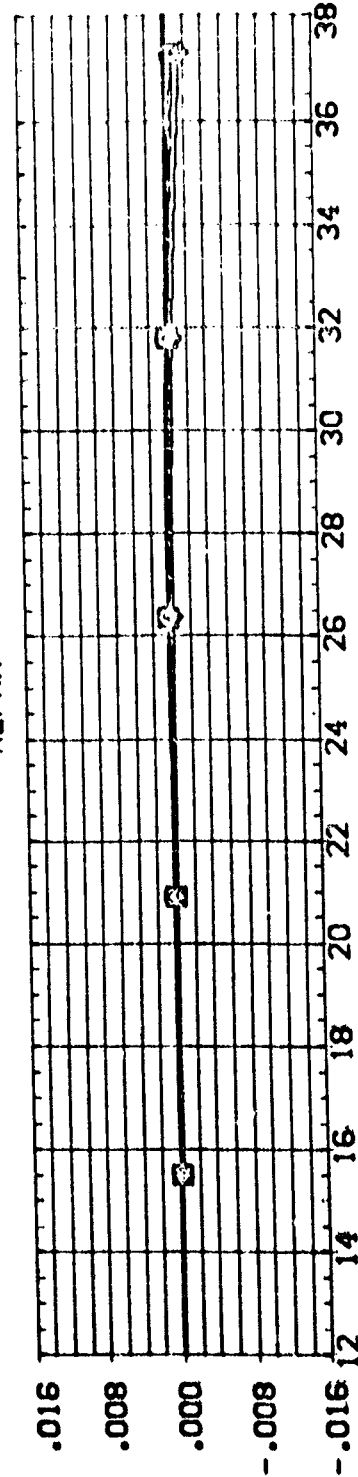
CY

ALPHA



CYN

ALPHA



CBL

ALPHA

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=2.5)

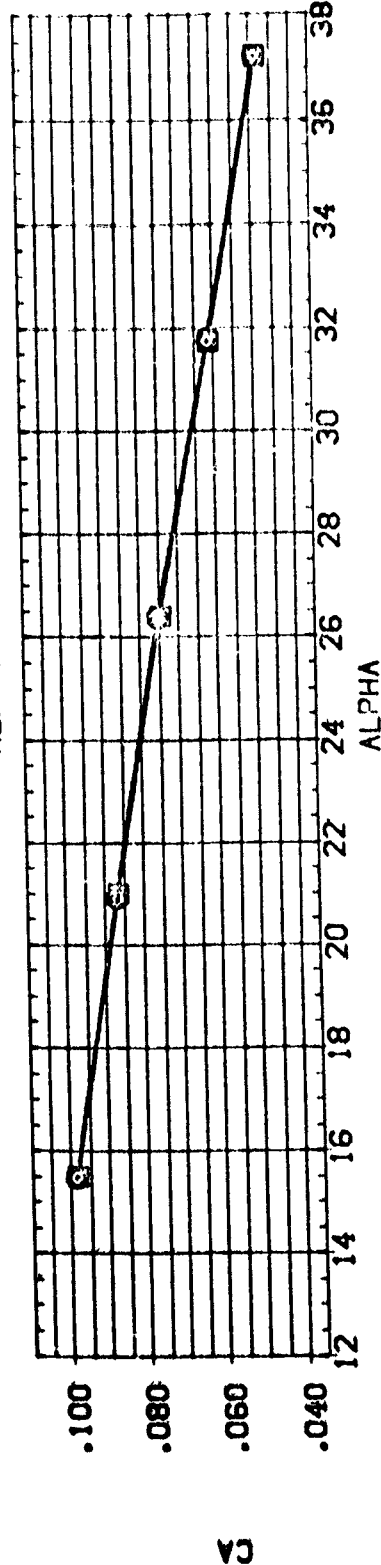
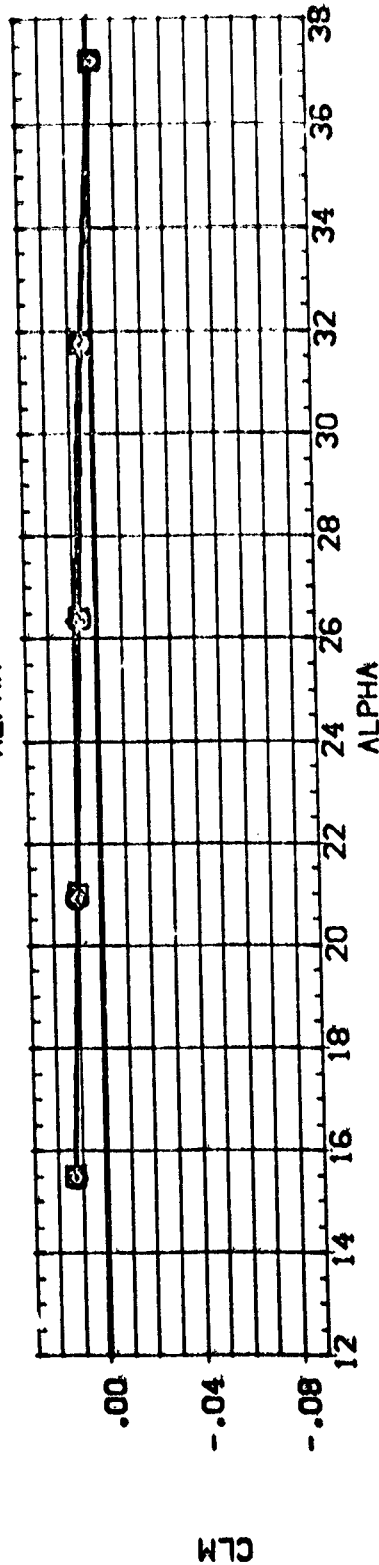
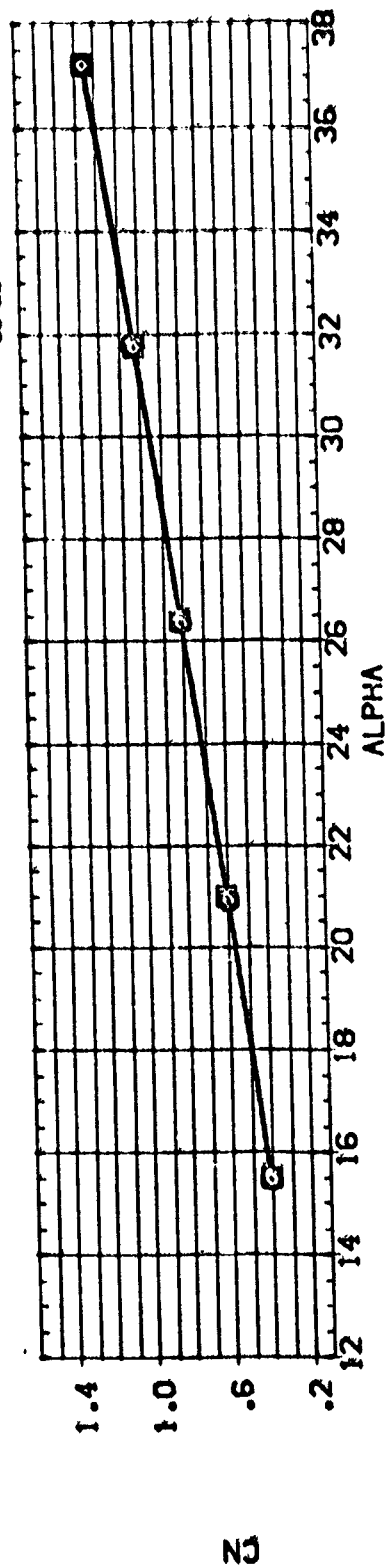


0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV005)

SYMBOL  
○  
□  
◇

PARAMETRIC VALUES  
PB-JET .277 BETA .000 MACH 2.500  
62.711 RNVL 1.720 ELEVTR -20.000  
226.743 AILRON .000 BOFLAP -14.250  
RUDFLR 40.000 RUDDER .000

REFERENCE INFORMATION  
SREF 87.1560 SQ-IN  
LREF 7.1222 IN-ES  
BREF 14.2500 IN-ES  
X-REF 2.5800 IN-ES  
Y-REF .0000 IN-ES  
Z-REF 6.0000 IN-ES  
SCALE .0150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=2.5)

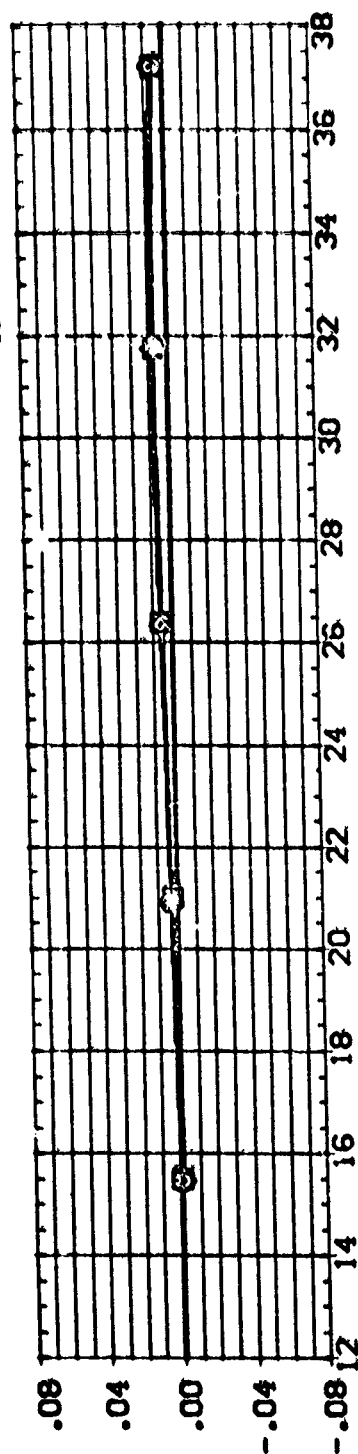
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SYMBOL  
○ □ ◇

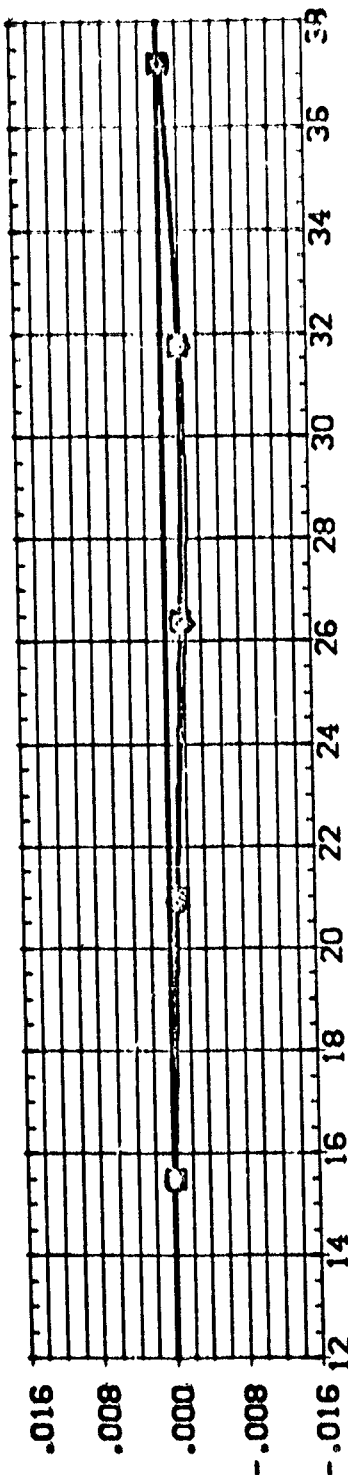
PG-JET BETA  
.277  
62.711  
226.743

PARAMETRIC VALUES  
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ELEVTR -20.000  
BDFLAP -14.250  
RUDDER .000

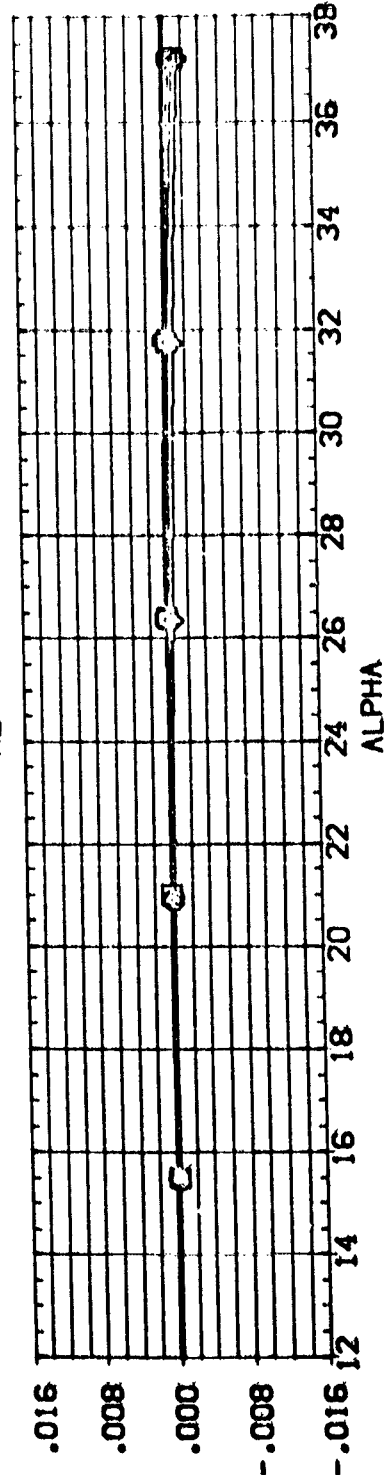
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CY



CYN



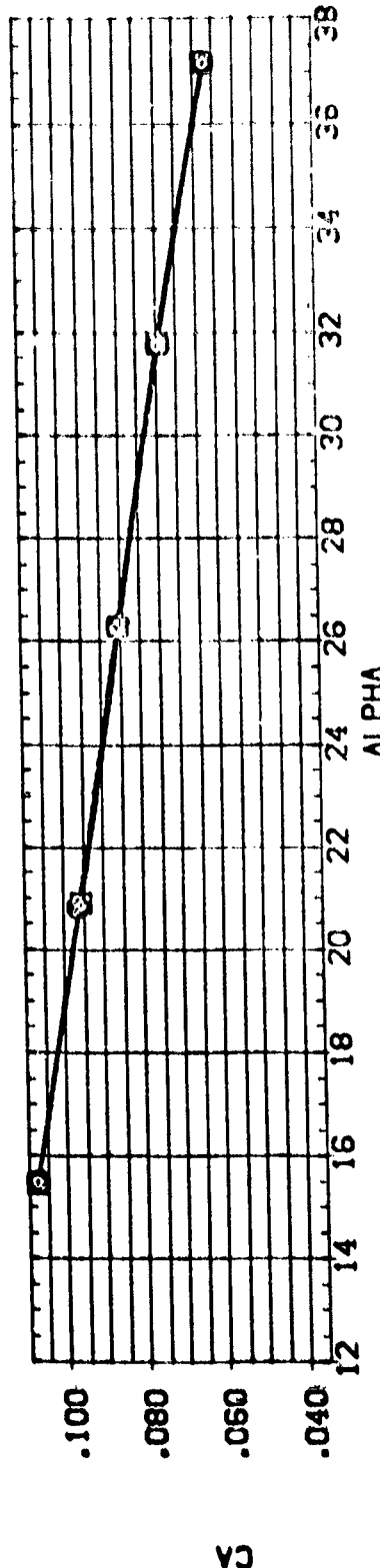
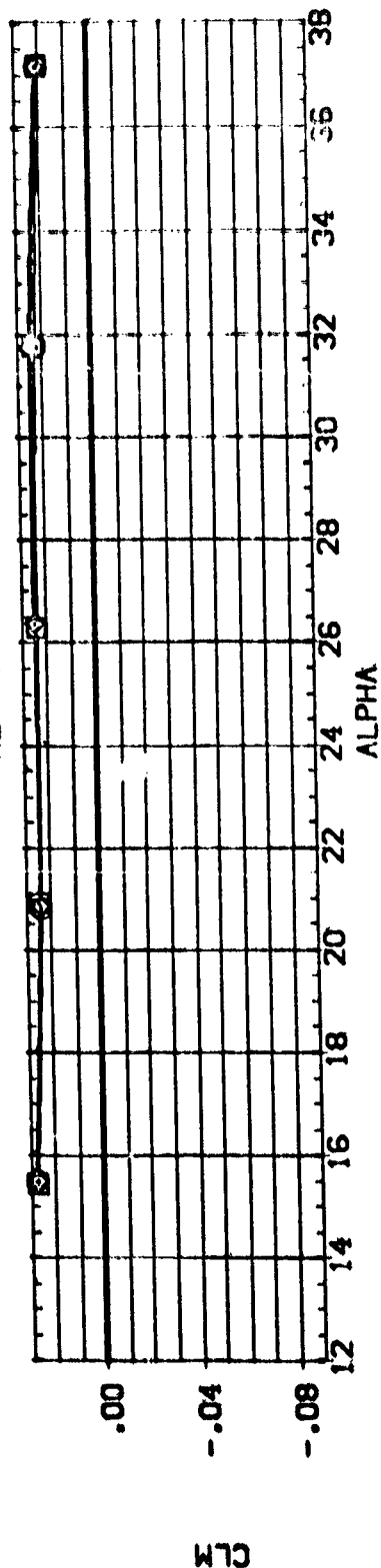
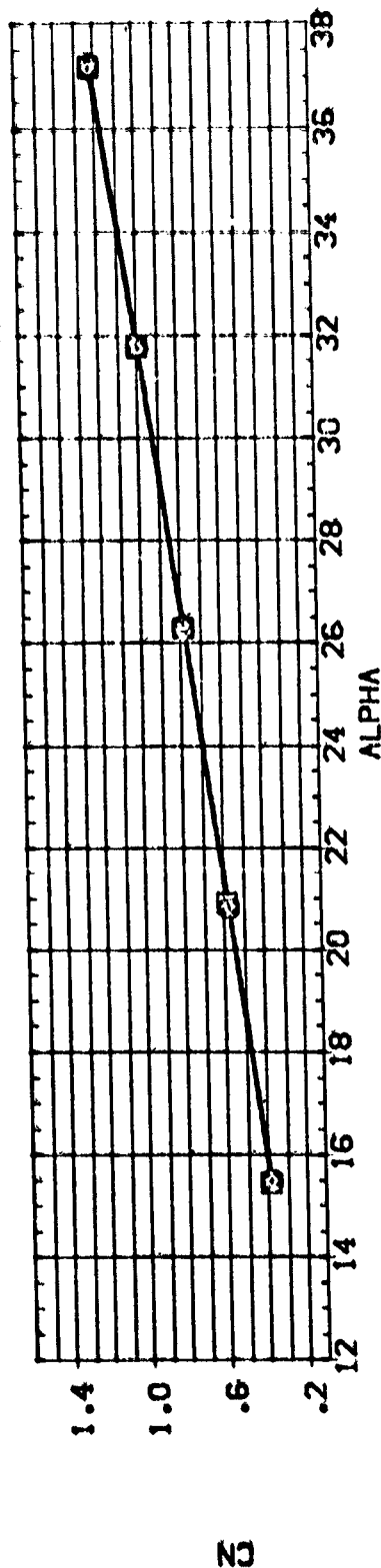
CBL

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=2.5)



0A-70. UPWT1043.CRB(819C7F5M6N19)(W107E23)(V7R5)(RPV008)

SYMBOL	PO-JET	PARAMETRIC VALUES				REFERENCE INFORMATION			
		BETA	MACH	ELEVTR	BOFLAP	RUDER	SPEC	DEF	SC.IN.
□	.337	.000	2.500	-48.000	-14.250	.000	87.1560	7.1222	1.0000
□	64.337	1.720	-48.000	-14.250	.000	.000	14.0500	1.0000	1.0000
◇	228.593	.000	-14.250	.000	.000	.000	2.5000	1.0000	1.0000
		40.000	RUDER				6.0000	1.0000	1.0000
							1.0150		



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=2.5)

CA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV008)

SYMBOL  
□  
□  
◇

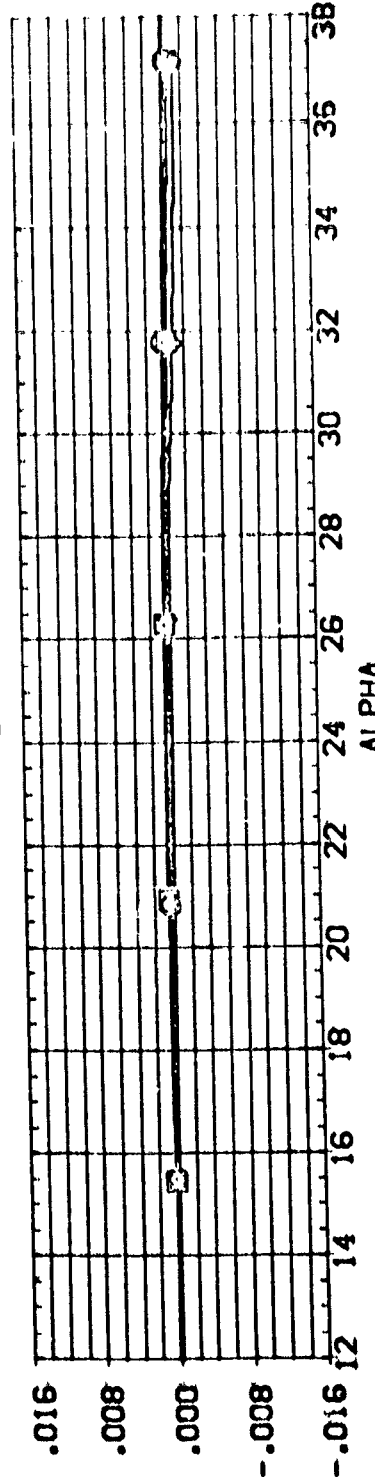
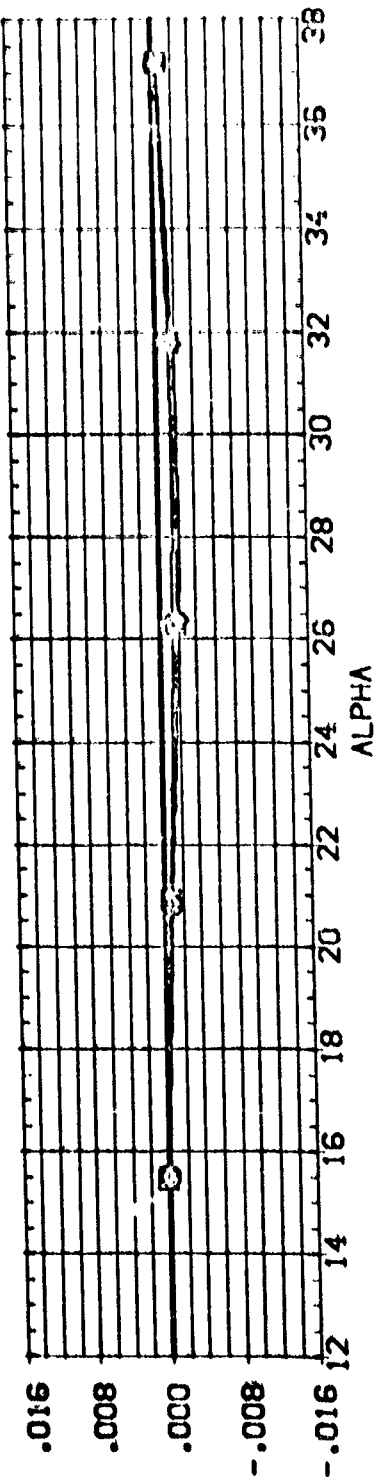
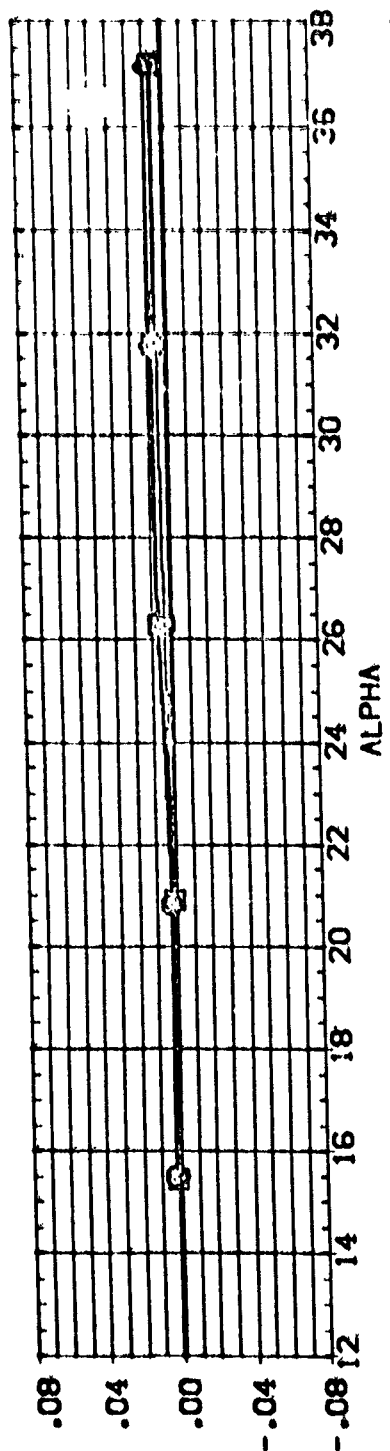
PG-JET  
.387  
64.387  
228.593

BETA  
RVAL  
A1URON  
RLOFLR

PARAMETRIC VALUES  
MACH  
ELEVTR  
BDFLAP  
RUDDER

2.500  
-40.000  
-14.250  
.000

REFERENCE INFORMATION  
SPEC 87.1560  
SFC 7.1272  
SPEC 14.5000  
XREF 12.5000  
YREF 1.0000  
ZREF 6.0000  
SCALE 0.0150



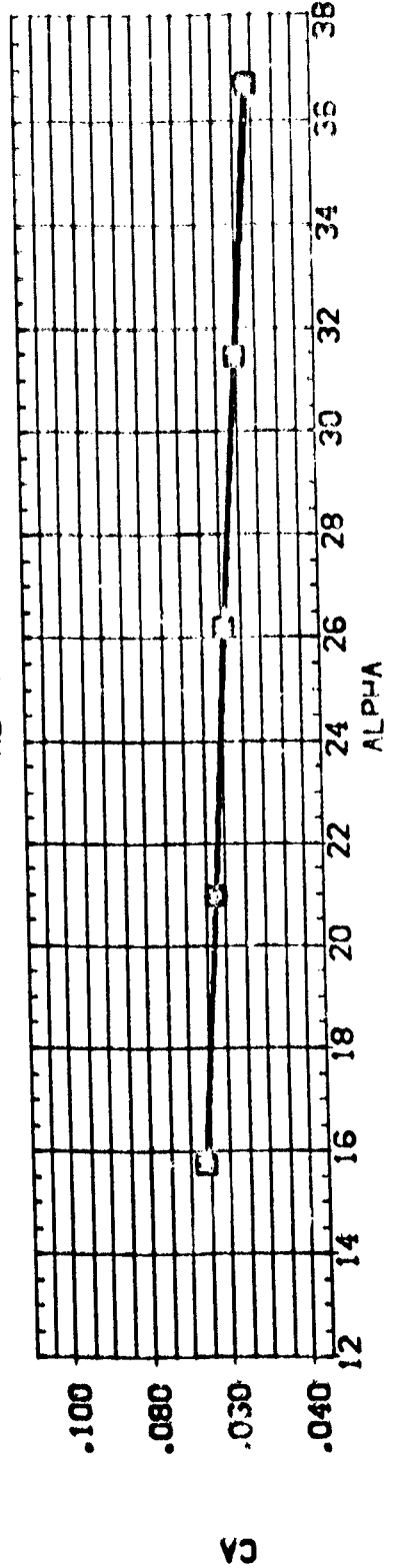
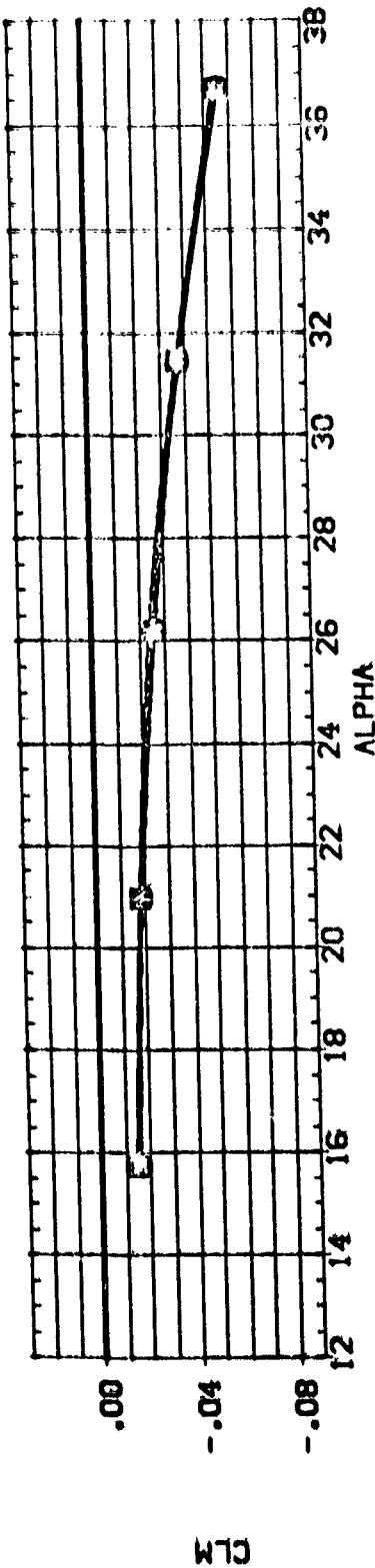
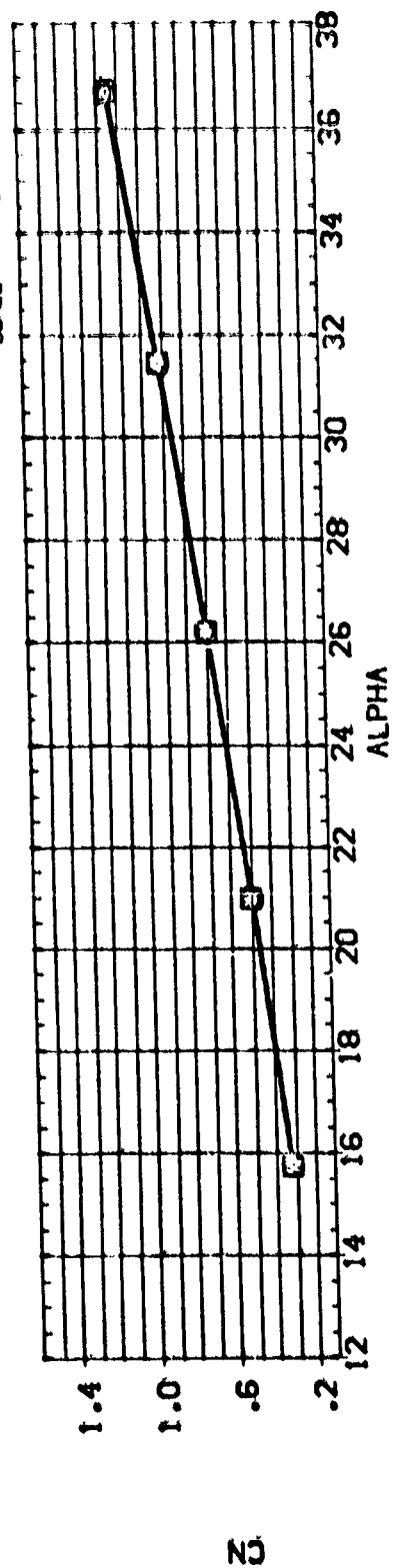
EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=2.5)



CA-70. UPWT1043.0R8(B19C7F5M6N19)(W107E23)(V7R5)(RPV011)

SYMBOL	PO-JET	BETA	PARAMETRIC VALUES	MACH	4.600
○	.317	RVA	1.720	ELEVTR	.000
□	70.137	ATLORN	.000	BOFLAP	.000
◇	161.887	RJOFIR	40.000	RODER	.000
△	397.752				

REFERENCE INFORMATION	SCALE
SREF	87.1560
LREF	7.1372
BREF	14.2500
XPRP	12.5800
YPRP	6.0000
ZPRP	6.0000
SCALE	1.150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=4.6)

GA-70. UPWT1043.0R8(319C7F5M6N19)(W107E23)(V7R5)(RPV011)

SYMBOL  
□  
◇  
△

PG-JET  
.317  
70.137  
161.887  
357.752

PARAMETRIC VALUES  
BETA  
RV/L  
ATLRN  
RUDFLR

.000  
1.720  
.000  
40.000

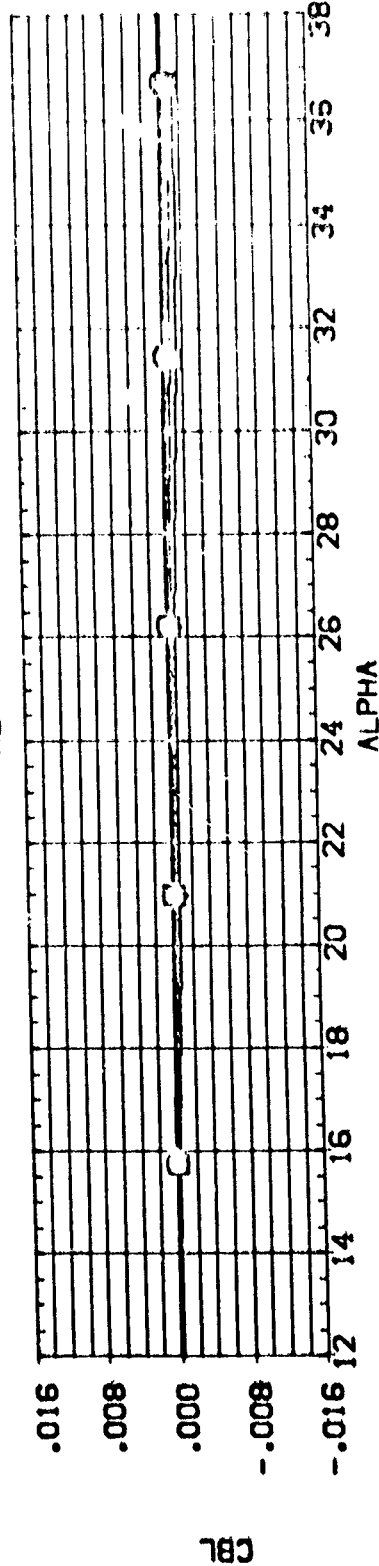
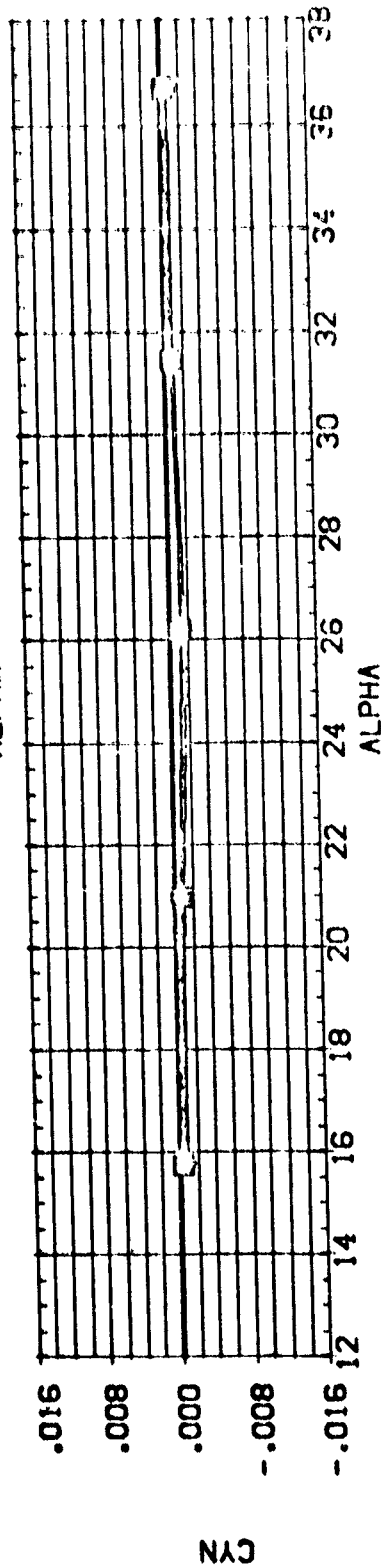
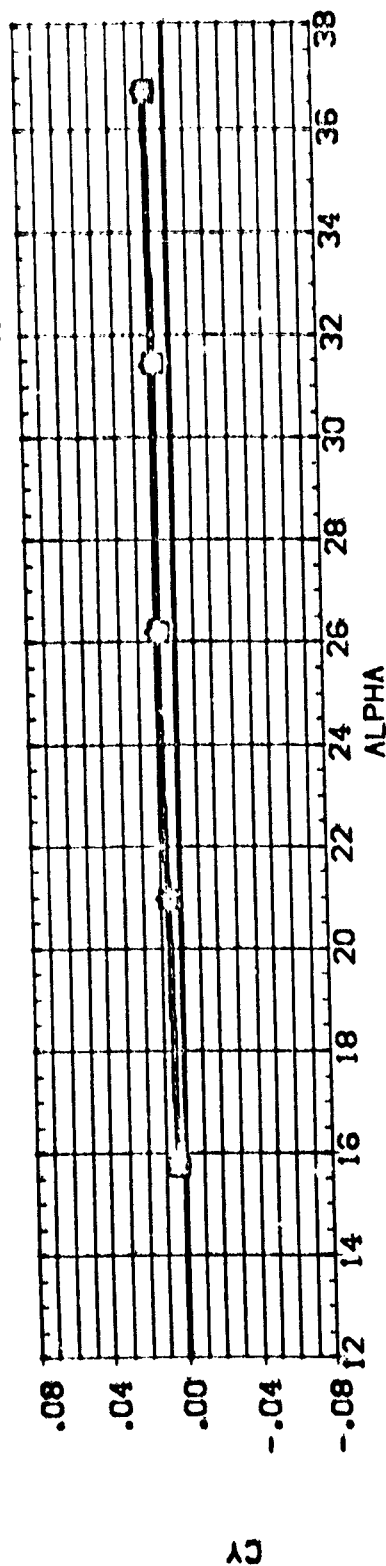
MACH  
ELEVTR  
BOFLAP  
R-DOER

4.600  
.000  
.000  
.000

SREF  
REF  
SREF  
XMOO  
VMOO  
ZMOO  
SCALE

87.1560  
7.1227  
4.2500  
12.5000  
10.0000  
5.0000  
1.0000

REFERENCE INFORMATION  
SQ-IN  
C-4-S  
C-4-S  
C-4-S  
C-4-S  
C-4-S  
C-4-S

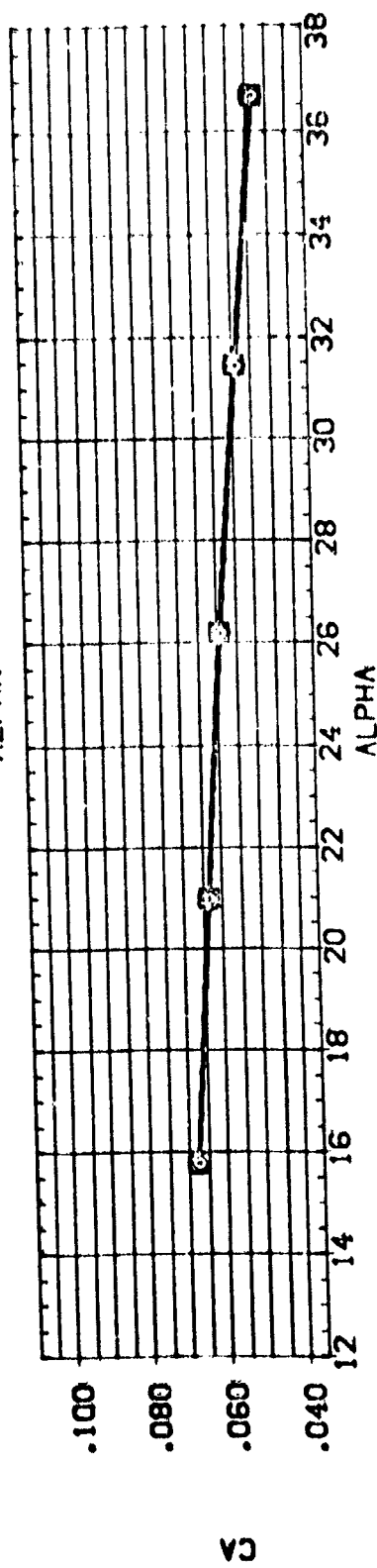
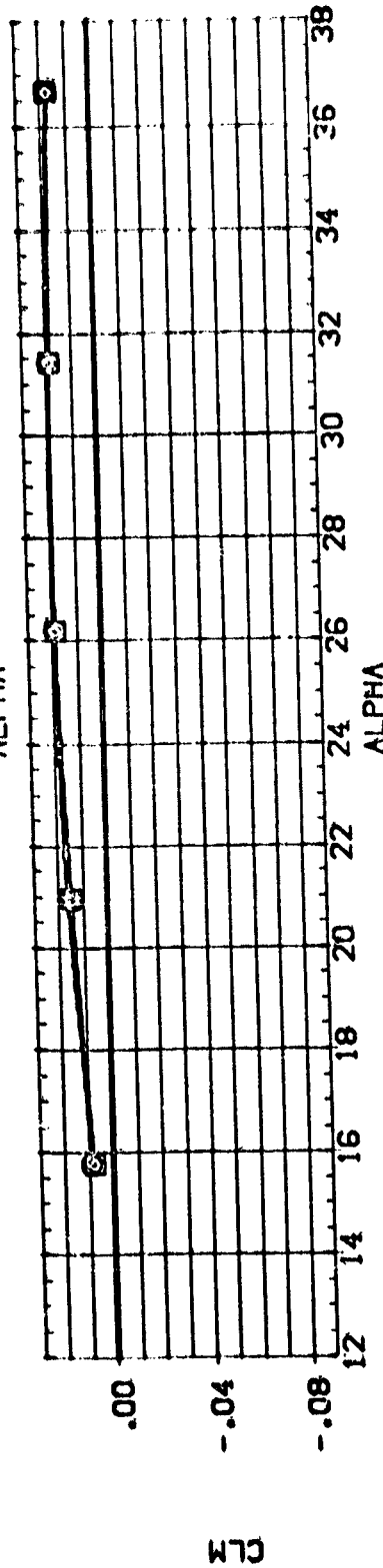
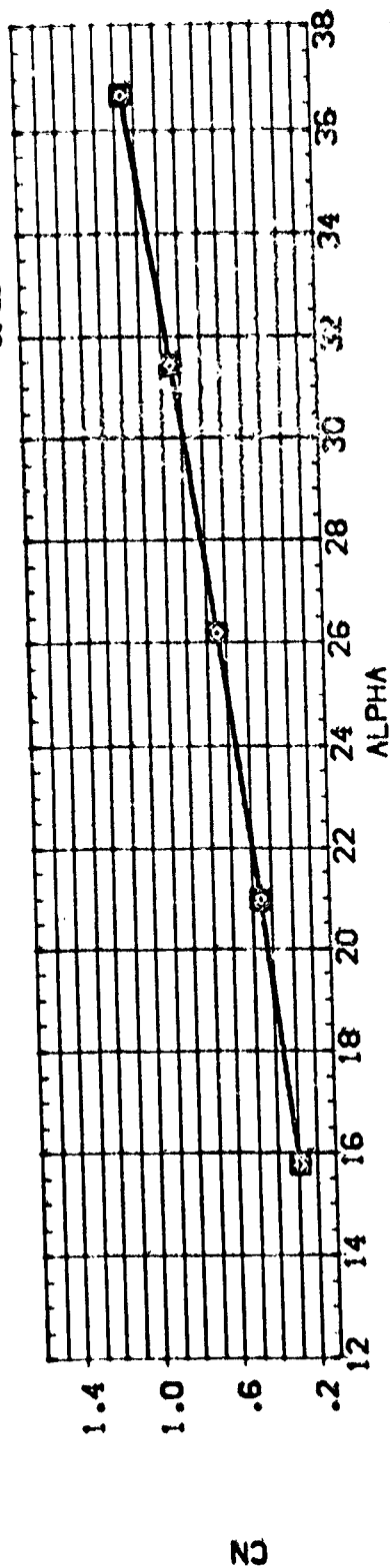


EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=4.6)



GA-70. UPWT1043.0R8(B19C7F5M6N19)(W107E23)(V7R5)(RPV014)

PARAMETRIC VALUES		REFERENCE INFORMATION	
PG-SET	BETA	SPREF	87.1560
.042	.000	LREF	7.1222
66.28	1.720	YREF	4.2500
163.65	.000	XREF	2.5800
	40.000	YREF	6.0000
	RUDER	ZREF	0.0000
		SCALE	0.050



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=4.6)



0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV014)

SYMBOL  
□  
◇

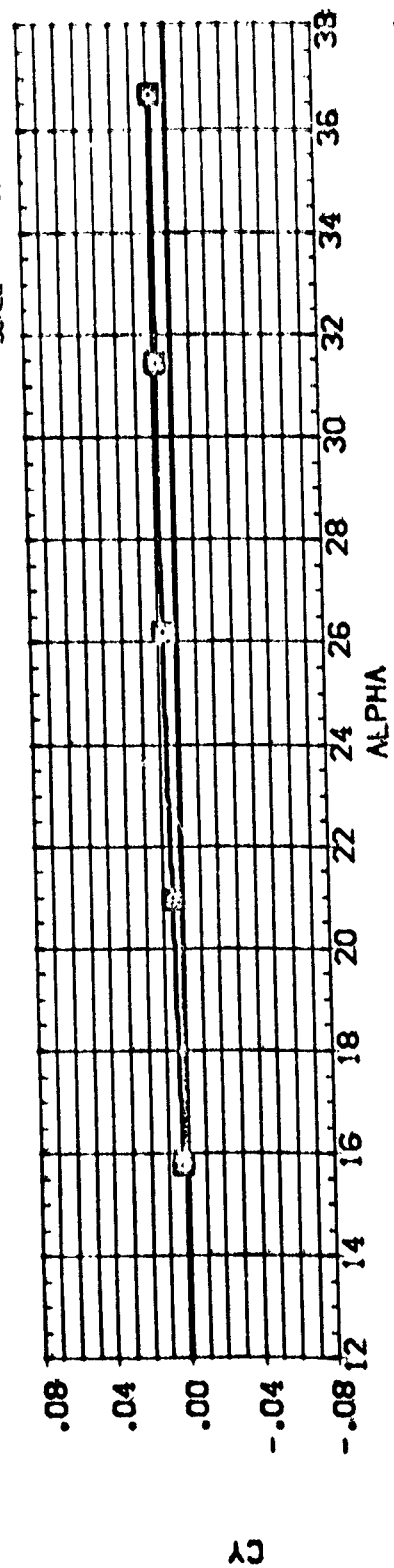
PG-JET .042  
66.281  
163.651

BETA  
RVL  
AILRON  
RUFLR

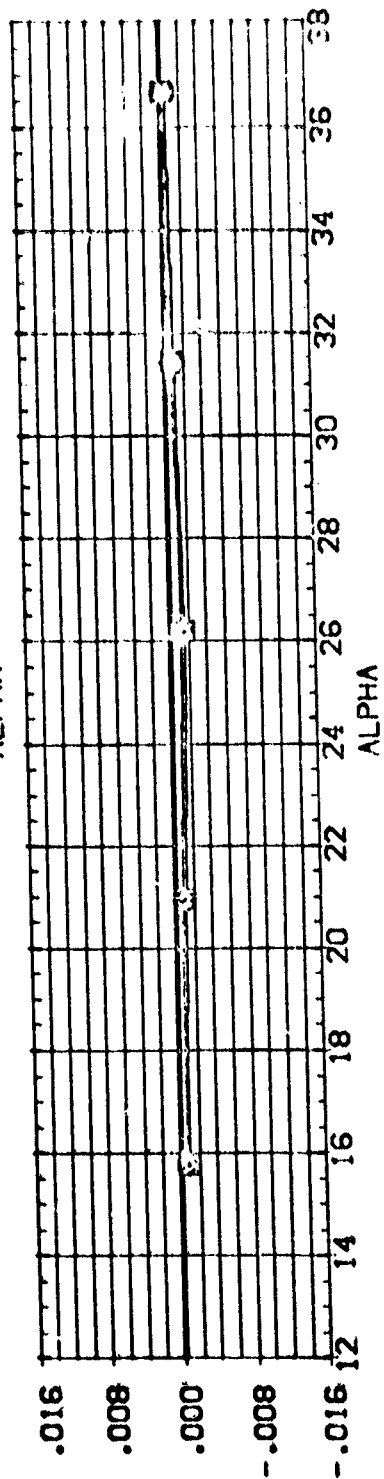
PARAMETRIC VALUES  
.000 MACH  
1.720 ELEVTR  
.000 BOFLAP  
40.000 RUDDER

4.600  
-20.000  
-14.250  
.000

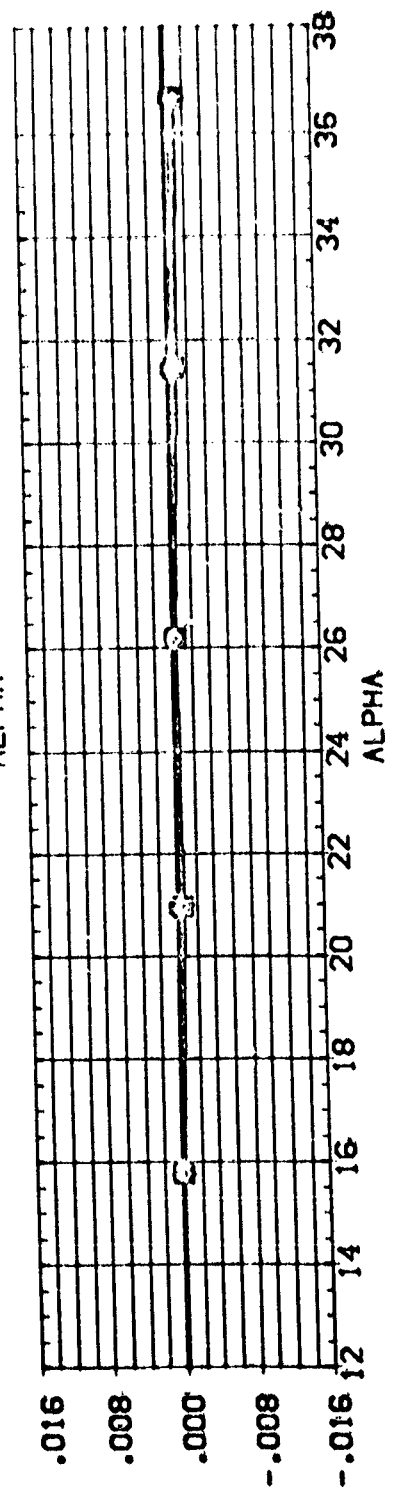
REFERENCE INFORMATION  
SREF 87.1560  
LREF 7.1222  
BREF 14.5500  
KREF 2.5800  
YREF 2.0000  
ZREF 6.3000  
SCALE 10.50



CY



CYN



CBL

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=4.6)



0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV017)

SYMBOL  
□ □ ◇

PG-JET  
-0.117  
71.998  
165.028

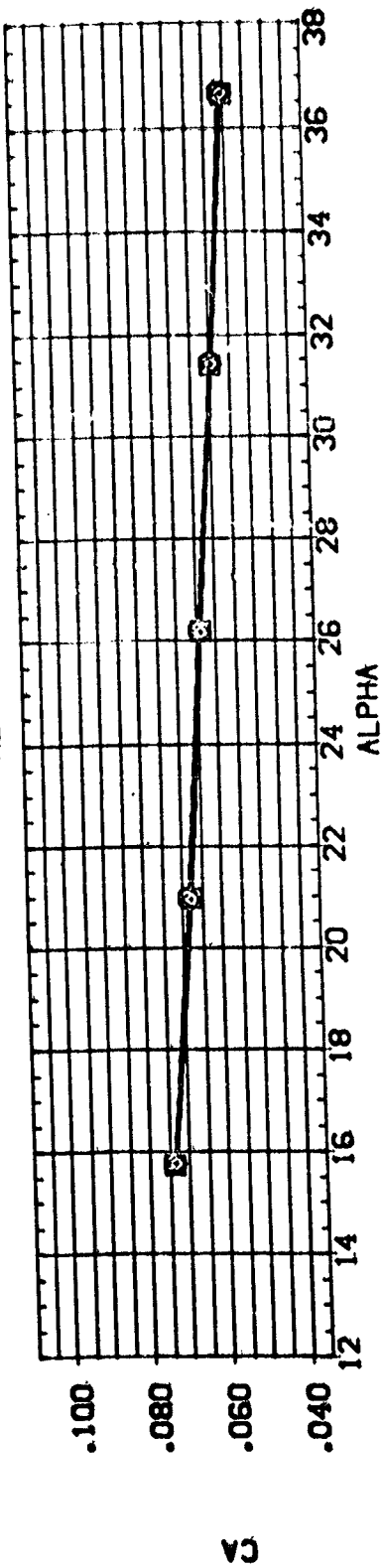
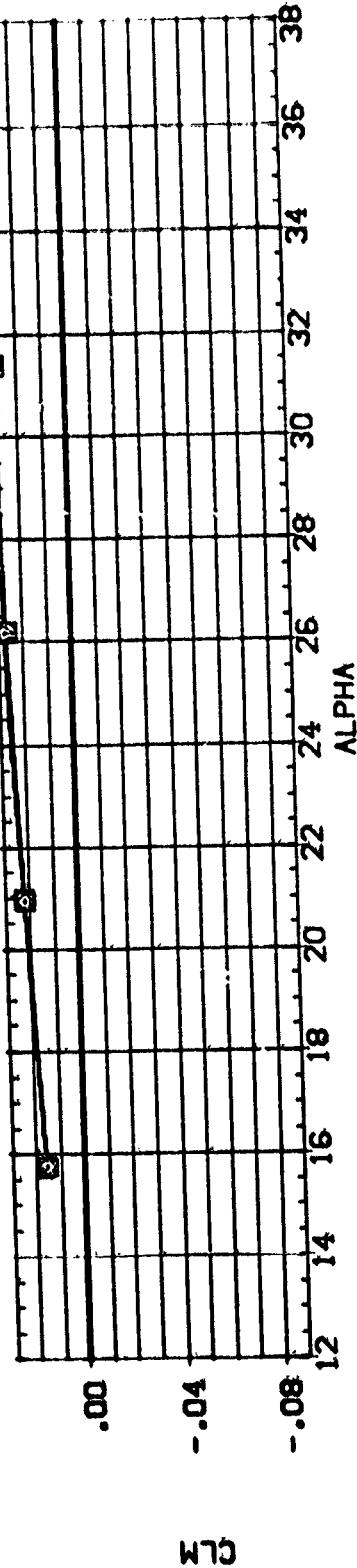
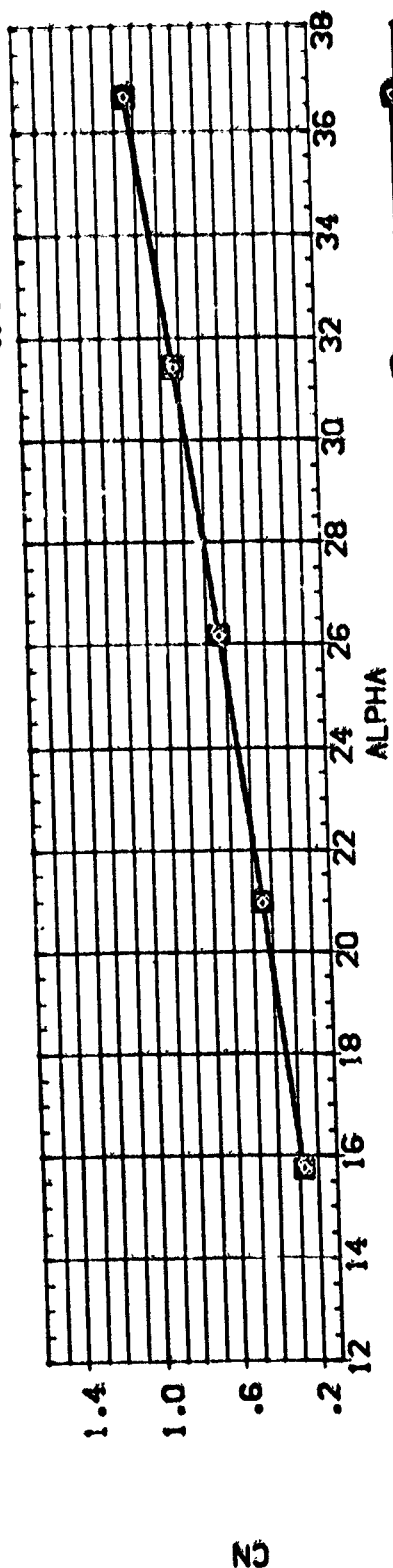
BETA  
RV/L  
AILRON

40.000  
40.000  
RUDER

4.600  
-40.000  
-14.250  
.000

PARAMETRIC VALUES  
MACH  
ELEVTR  
BEFLAP  
RUDER

REFERENCE INFORMATION  
SREF 87.1560  
LREF 7.1222  
BREF 14.0500  
XGRP 12.5800  
YGRP .3000  
ZGRP 6.0000  
SCALE .0150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=4.6)

GA-70. UPWT1043.ORB(B19C7F5M6N19)(W107E23)(V7R5)(RPV017)

SYMBOL

PG-JET

BETA

RVL

AILRON

RLOFLR

PARAMETRIC VALUES

.000

1.720

.000

40.000

4.600

-40.000

-14.250

.000

REFERENCE INFORMATION

SREF

87.560

LREF

7.222

BREF

14.550

XPRP

12.500

YPRP

.500

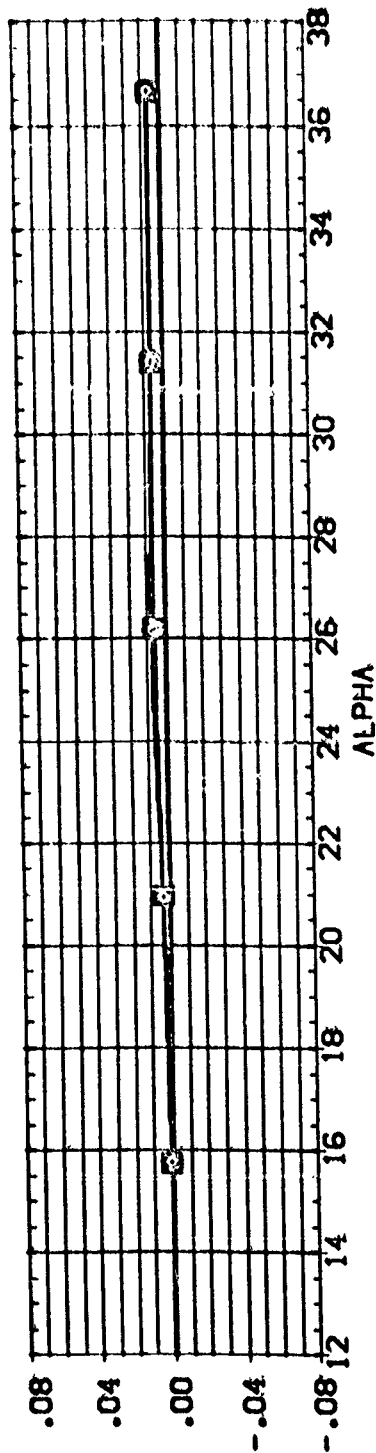
ZPRP

6.000

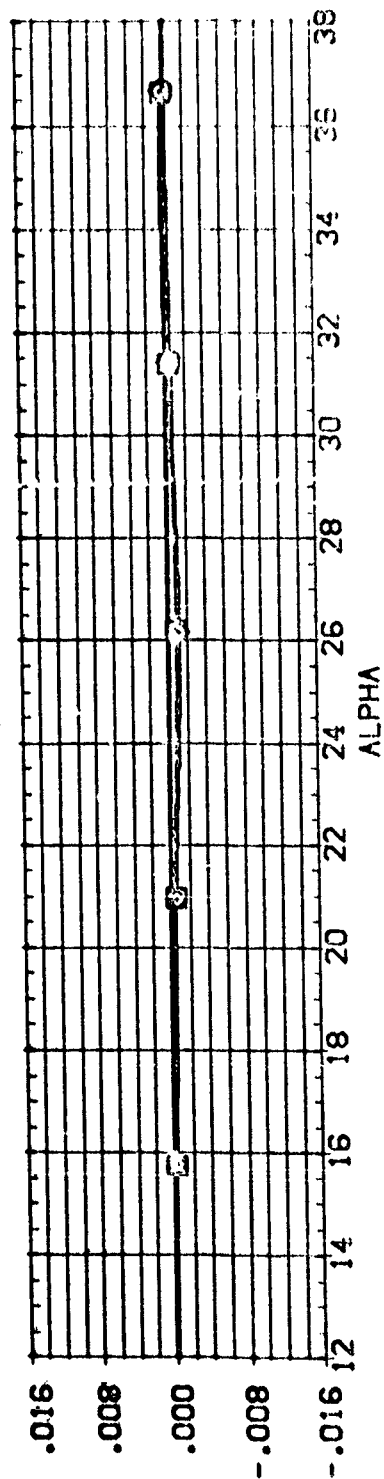
SCALE

.0150

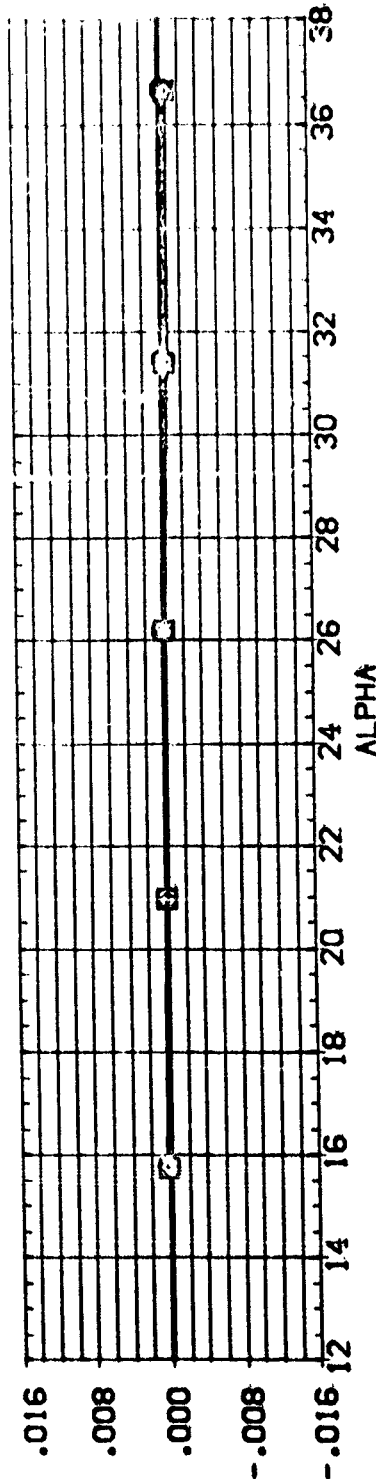
CY



CYN



CBL



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=0, MACH=4.6)



CA-70. UPWT1043.CRB(B19C7F5M6N19)(W107E23)(V7R5)(RPV001)

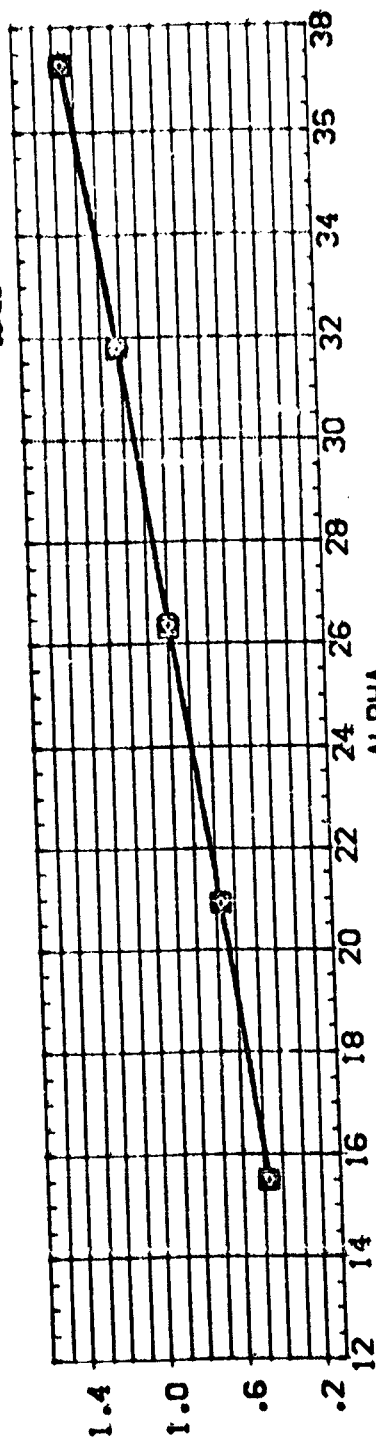
SYMBOL  
□ □ ◇

PG-JET  
.316  
63.198  
227.005

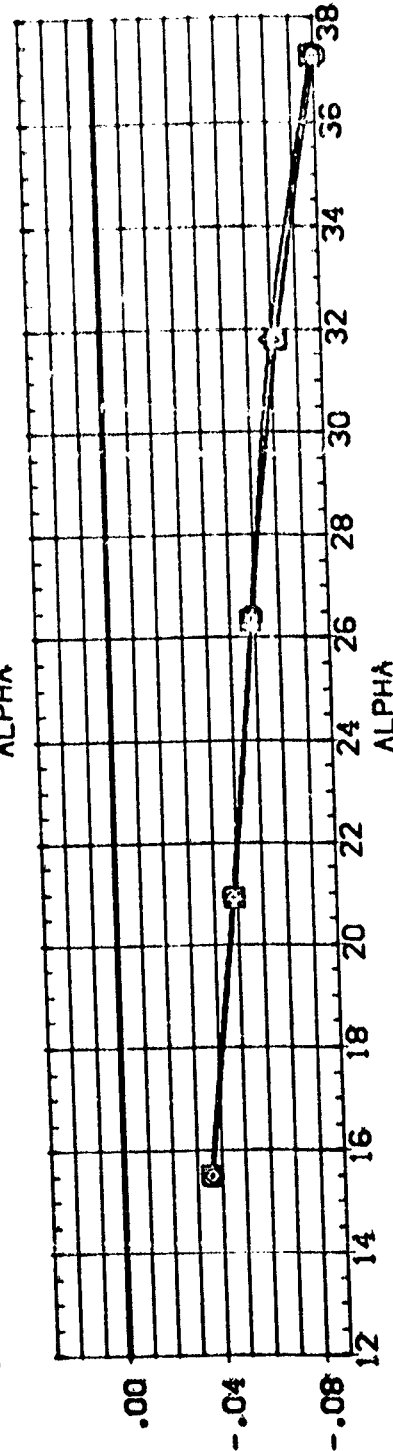
BETA  
RV/L  
AILRON  
RJOF/LR

PARAMETRIC VALUES  
-5.000 MACH 2.500  
1.720 ELEVTR .000  
.000 BOFLAP .000  
40.000 P-DOOR .000

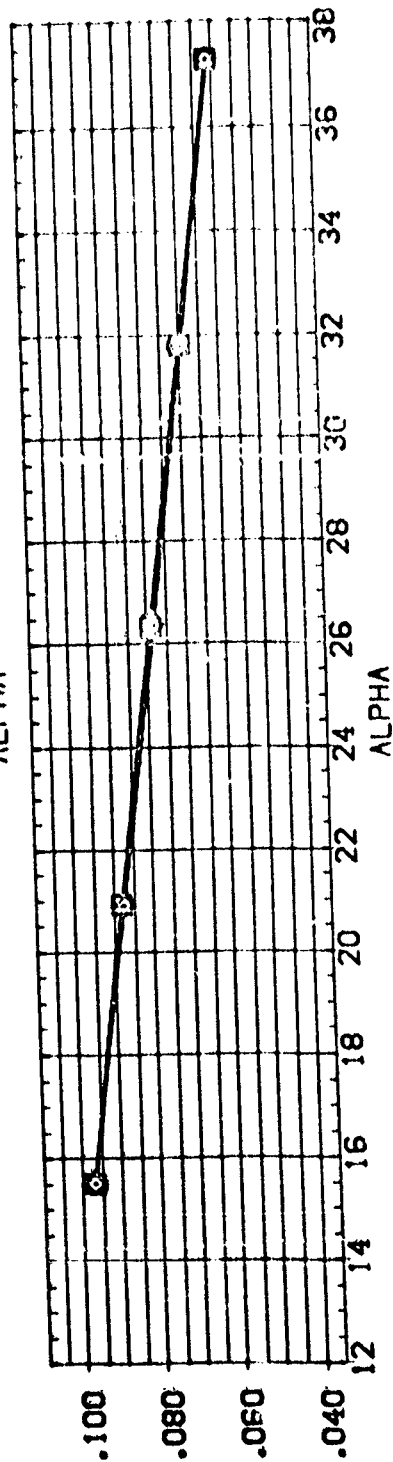
REFERENCE INFORMATION  
SREF 87.1560  
LREF 7.1222  
BREF 14.0500  
MAPP 12.5800  
CAPP 6.0000  
ZAPP 6.0000  
SCALE .0150



Z



CM



CS

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=2.5)

CA-70. UPWT1043.0R8(B19C7F5M6N19)(W107E23)(V7R5)(RPV001)

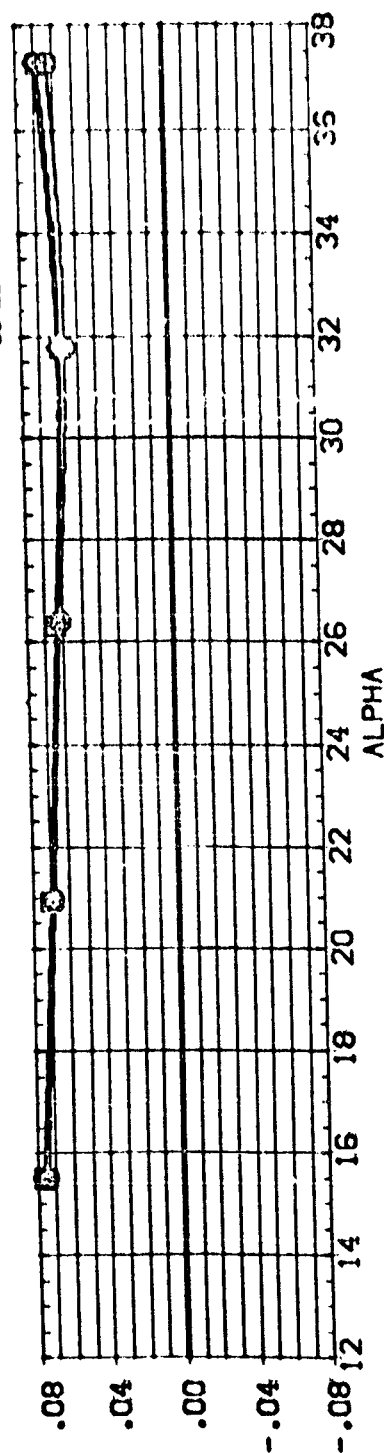
SYMBOL

PG-JET  
.316  
63.198  
227.005

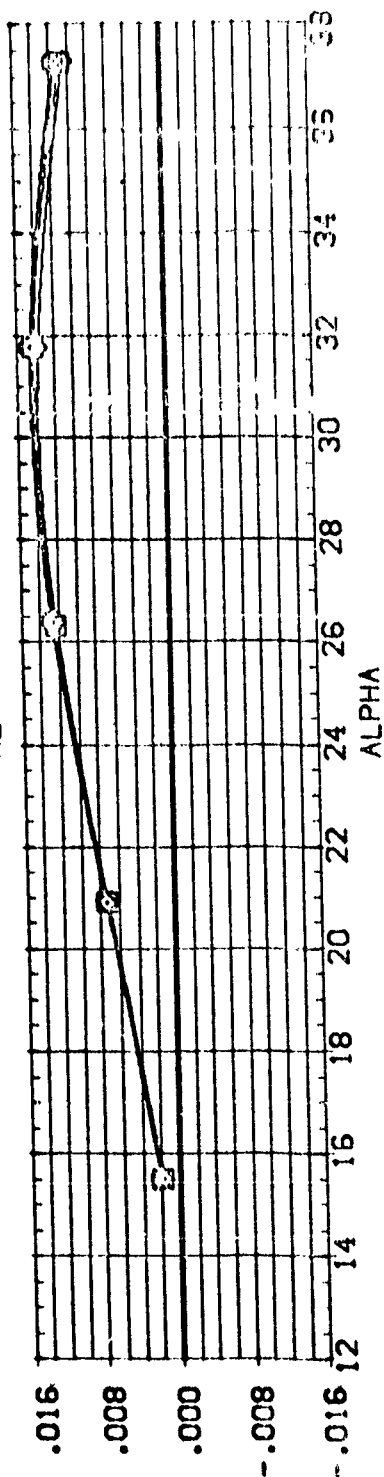
BETA  
RVL  
AILRON  
RUJFLR

PARAMETRIC VALUES  
-5.000 MACH 2.500  
1.720 ELEVTR .000  
.000 BOFLAP .000  
40.000 RUDDER .000

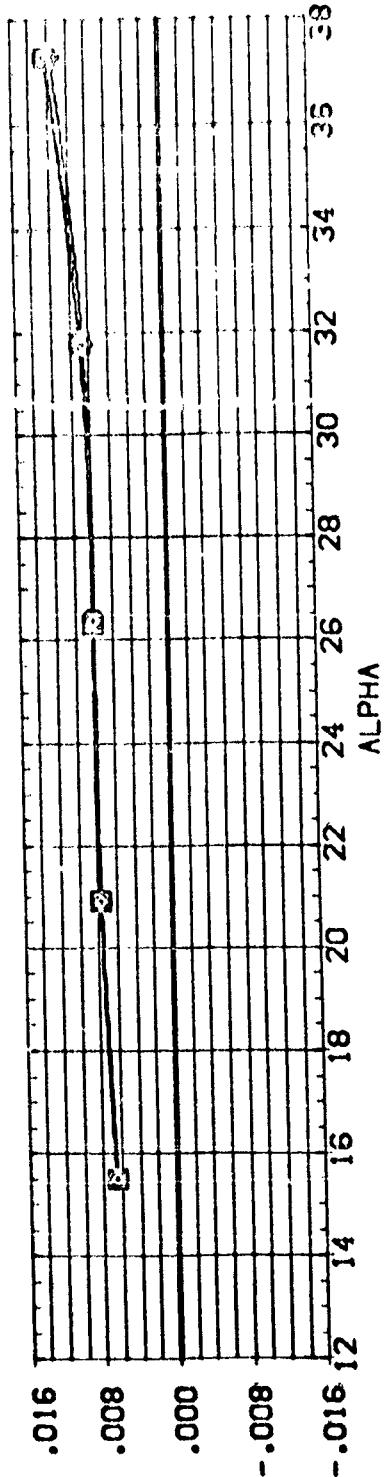
REFERENCE INFORMATION  
SPEC 87.1560  
LREF 7.1222  
BREF 14.0500  
XMRP 12.5000  
YMRP .0000  
ZMRP 6.0000  
SCALE 10.150



CY



CYN



CBL

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=2.5)



0A-70. UPW-1043.0R8(B19C7F5M6N19)(W107E23)(V7R5)(RPV004)

SYMBOL  
 ○ □ ◇

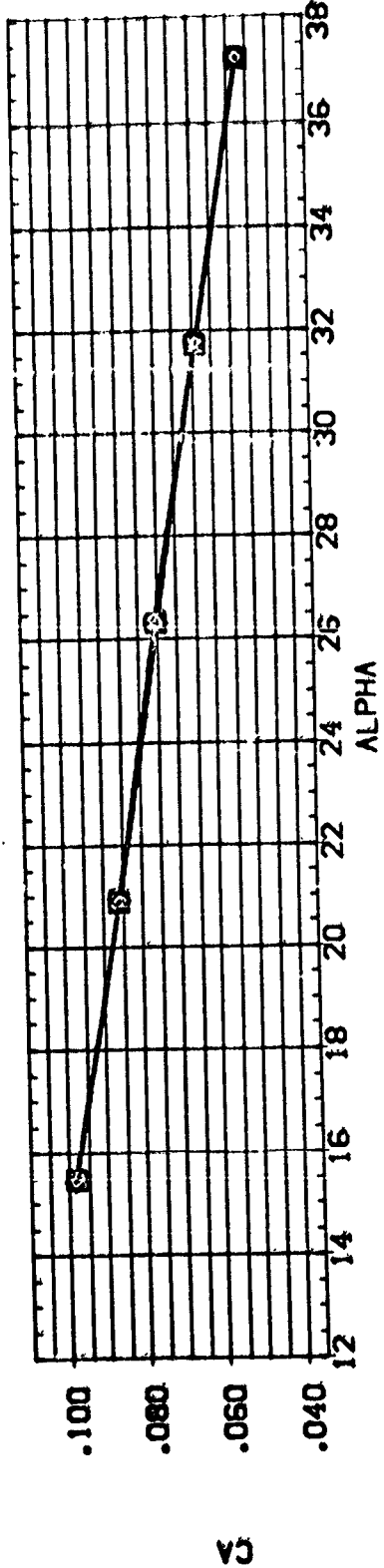
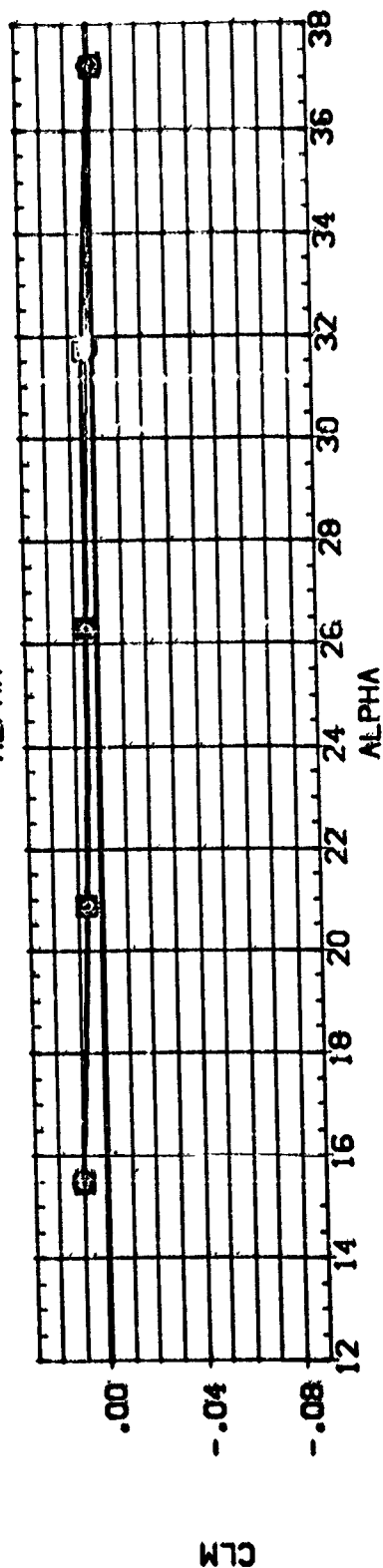
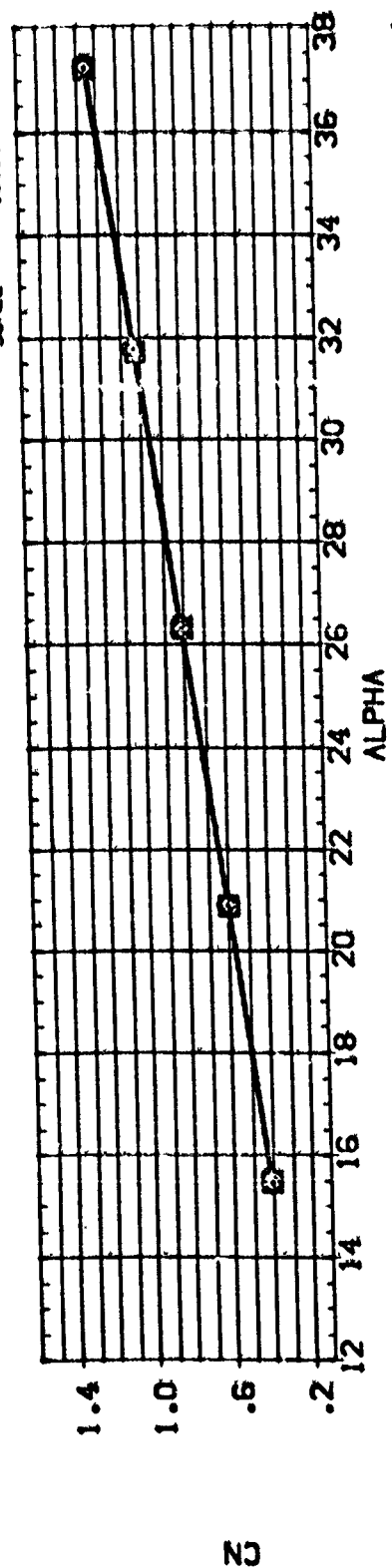
PO-JET  
 .272  
 64.051  
 276.069

BETA  
 .772  
 RVAL  
 1.720  
 AILRON  
 .000  
 ROLFLR  
 40.000

PARAMETRIC VALUES  
 MACH 2.500  
 ELEVTR -20.000  
 BOFLAP -14.250  
 RODER .000

REFERENCE INFORMATION  
 SREF 82.1560  
 LREF 7.1222  
 BREF 14.6500  
 XMRP 12.5800  
 YMRP .0000  
 ZMRP 6.0000  
 SCALE .0150

SQ. IN.  
 INCHES  
 INCHES  
 INCHES  
 INCHES  
 INCHES



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=2.5)



CA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV004)

SYMBOL



PG-JET

.272  
64.051  
276.063

PARAMETRIC VALUES

BETA  
RVL  
AILRON  
RDLR

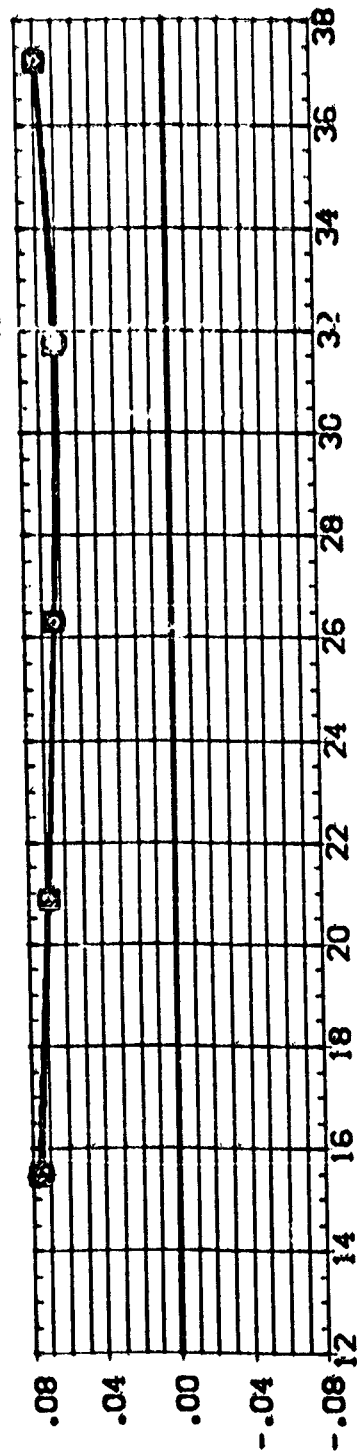
-5.000  
1.720  
.000  
40.000

MACH  
ELEVTR  
BDLAP  
RDOER

2.500  
-20.000  
-14.250  
.000

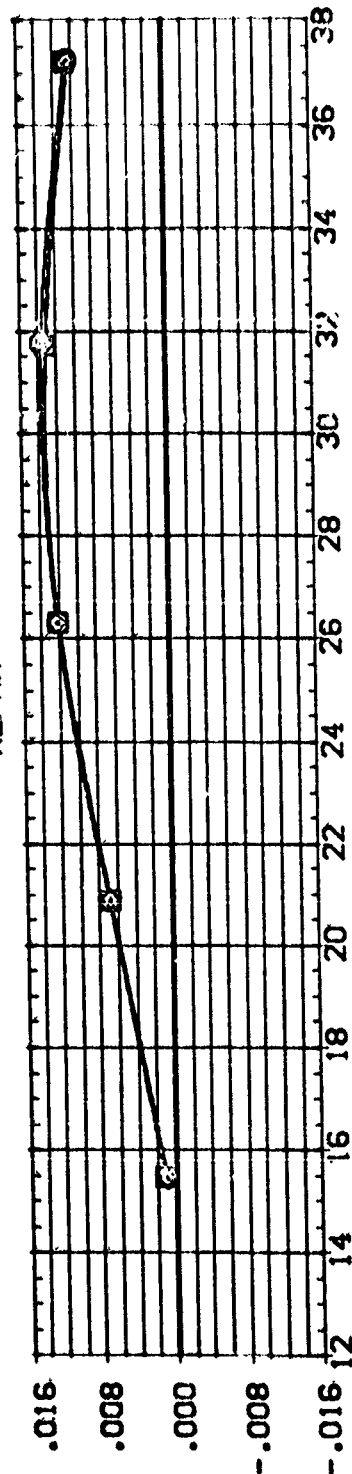
REFERENCE INFORMATION

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BREF 14.0500  
XREF 12.5800  
YREF .0000  
ZREF 6.0000  
SCALE .0150



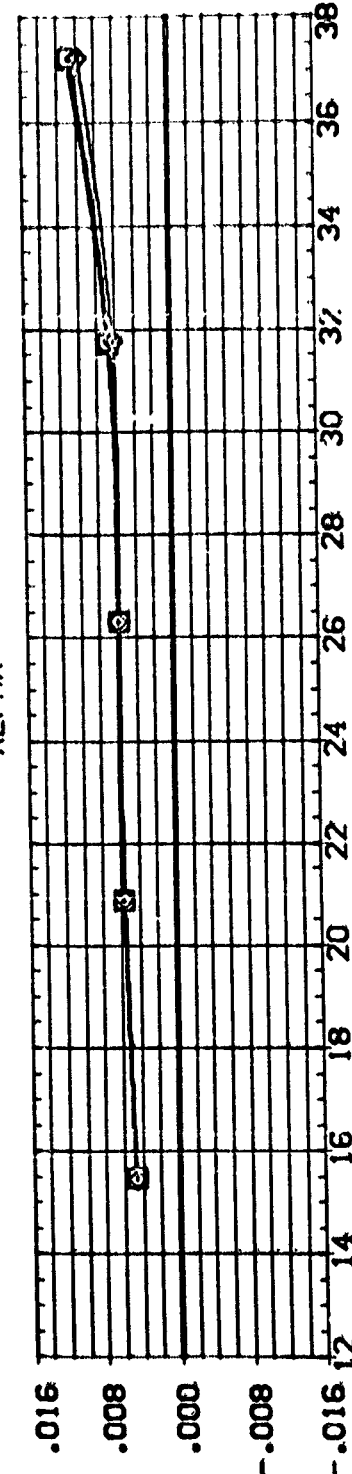
CY

ALPHA



CYN

ALPHA



CBL

ALPHA

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=2.5)



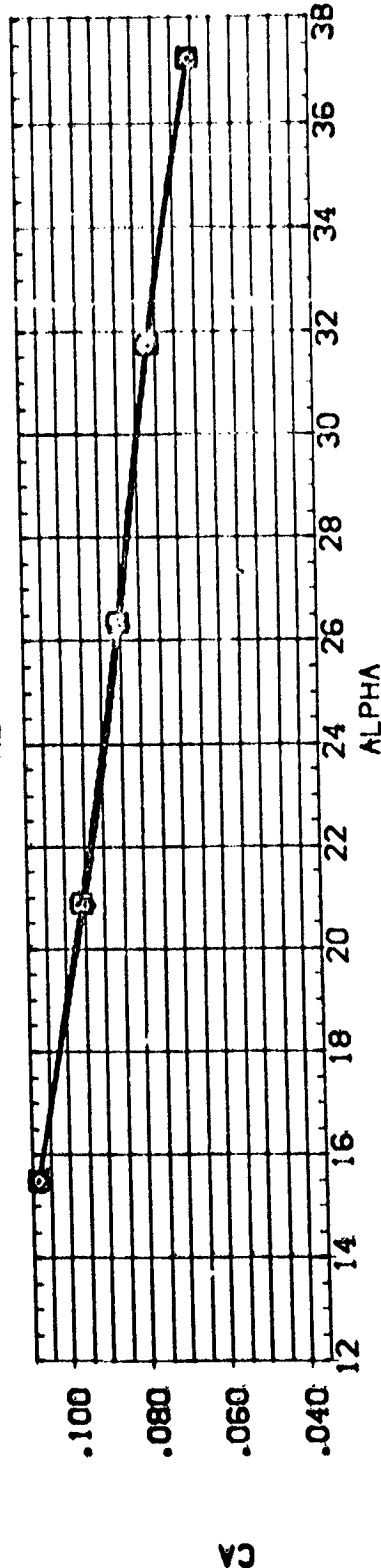
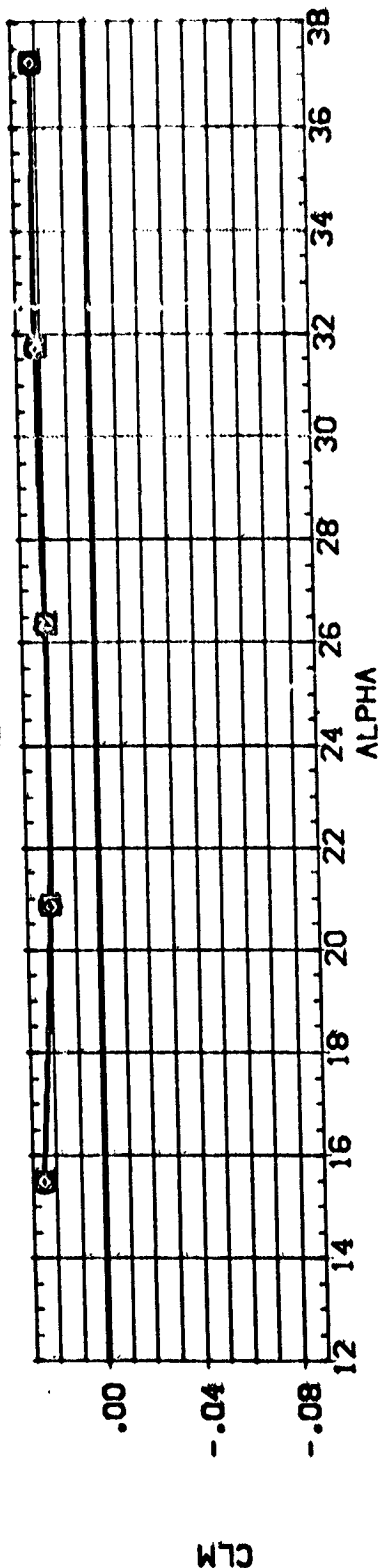
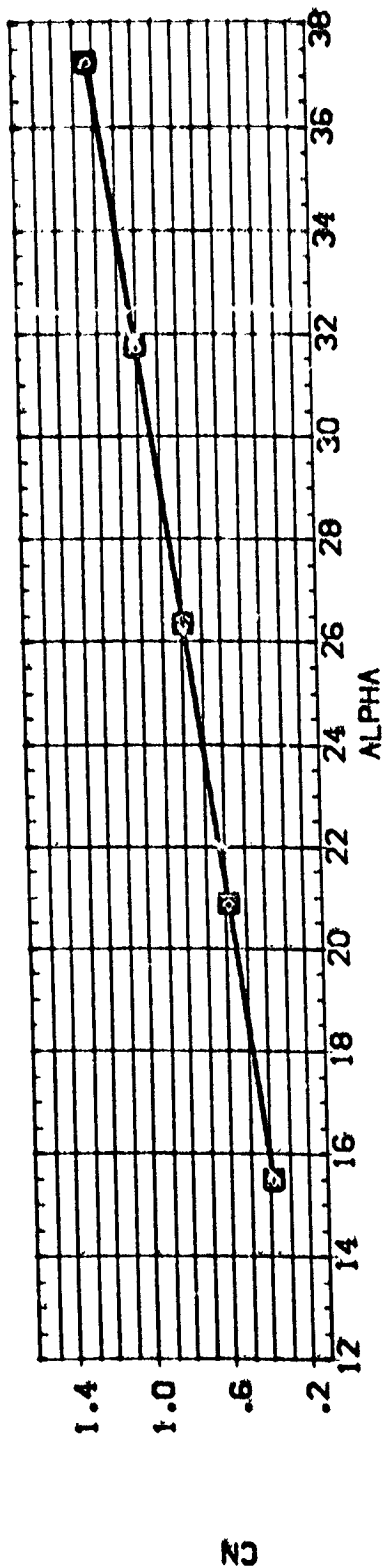
0A-70, UPWT1043, ORB(B19C7F5M6N19)(W107E23)(V7R5)(RPV007)

SYMBOL  
□  
◇

PO-JET  
1.009  
67.695  
225.683

PARAMETRIC VALUES  
BETA -5.000 MACH 2.500  
RWAL 1.720 ELEVTR -40.000  
AILRON .000 BOFLAP -14.250  
RLOFLR 40.000 RUDDER .000

REFERENCE INFORMATION  
SPCF 87.1560  
LREF 7.1222  
BREF 14.2500  
WREF 12.5600  
W40 2.000  
W40 6.0000  
SCALE .0150

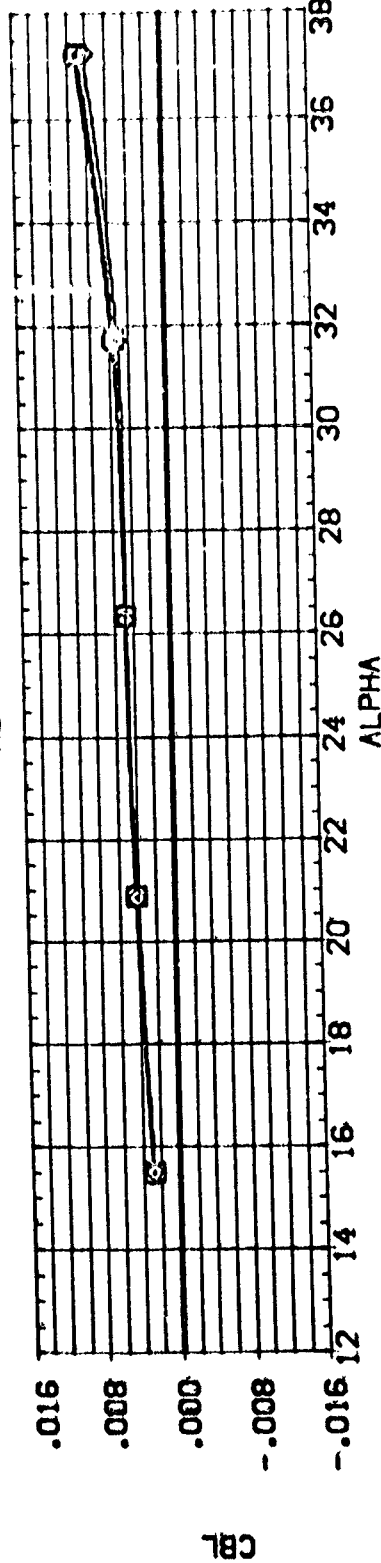
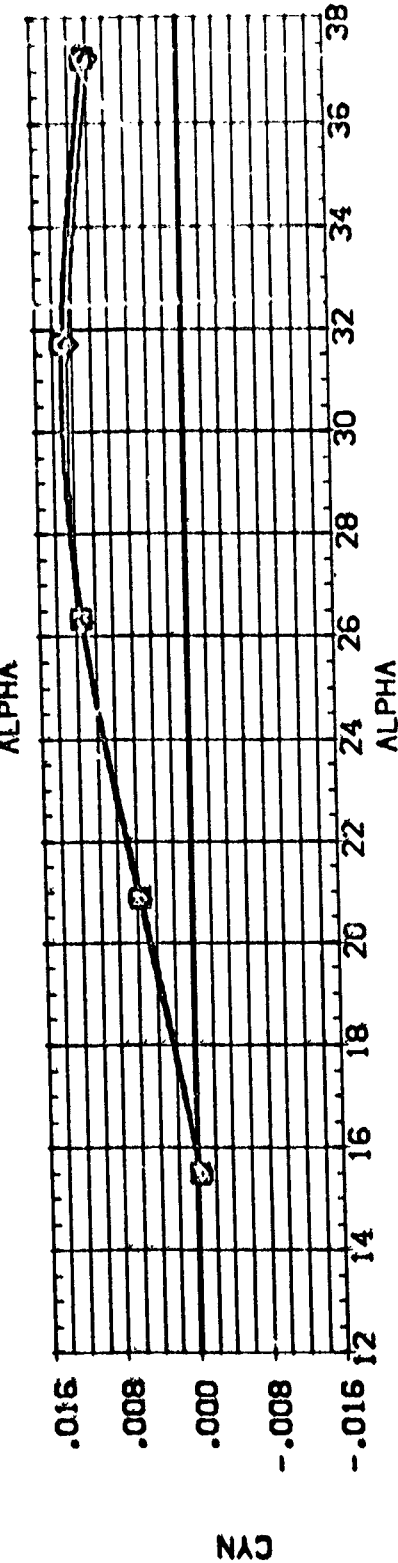
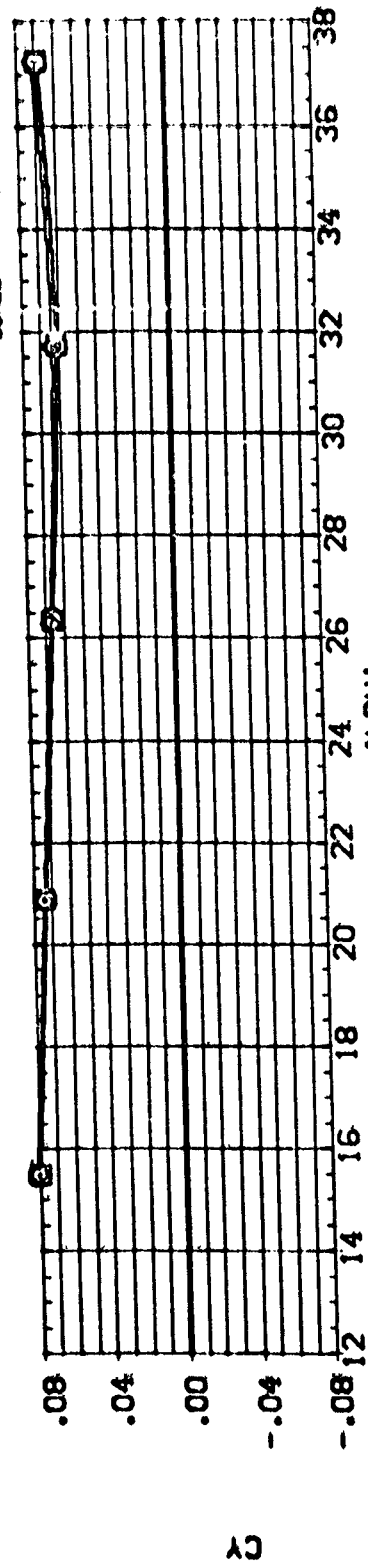


EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=2.5)



GA-70. UPWT1043.ORB(B19C7F5M6N19)(W107E23)(V7R5)(RPV007)

SYMBOL	PG-JET	BETA	PARAMETRIC VALUES				REFERENCE INFORMATION			
			-5.000	MACH	2.500		SRF	87.1560	SG, IN.	SCALES
□	67.695	RV/L	1.720	ELEVTR	-40.000		LR/F	7.1272		SCALES
◇	275.583	AILRON	.000	BOFLAP	-14.250		BR/F	14.2500		SCALES
		RUDFLR	40.000	RUDDER	.000		YPRP	12.5800		SCALES
							ZPRP	6.0000		SCALES
							SCALE	.0150		



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=2.5)



SYNOPSIS

89-1ET  
.315  
70.359  
162.556

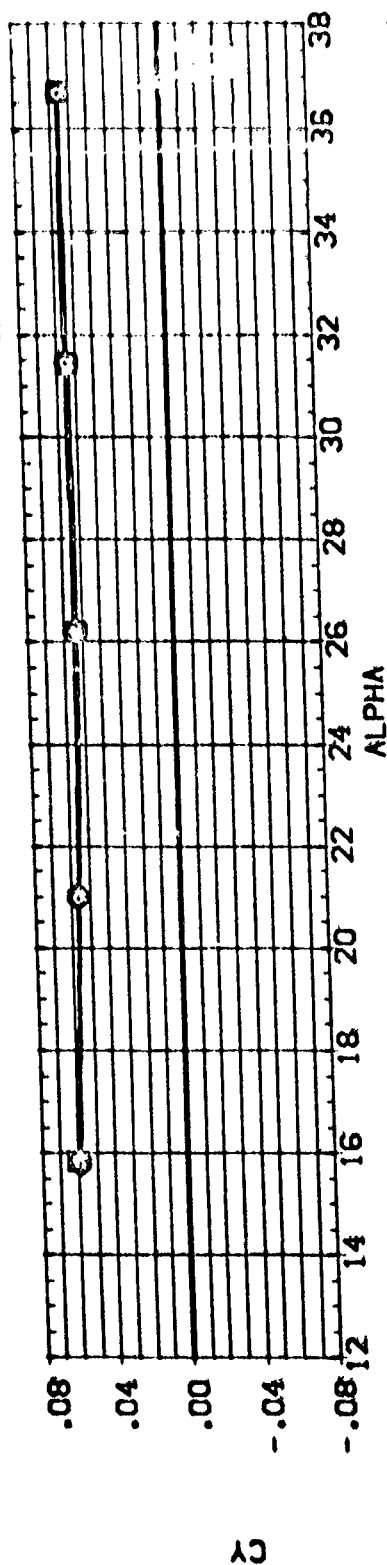
BETA  
REVL  
AILRON  
R.OFLR

PARAMETRIC VALUES
MACH
-5.000
ELEVTR
1.720
BOFLAP
.000
RJ00ER
40.000

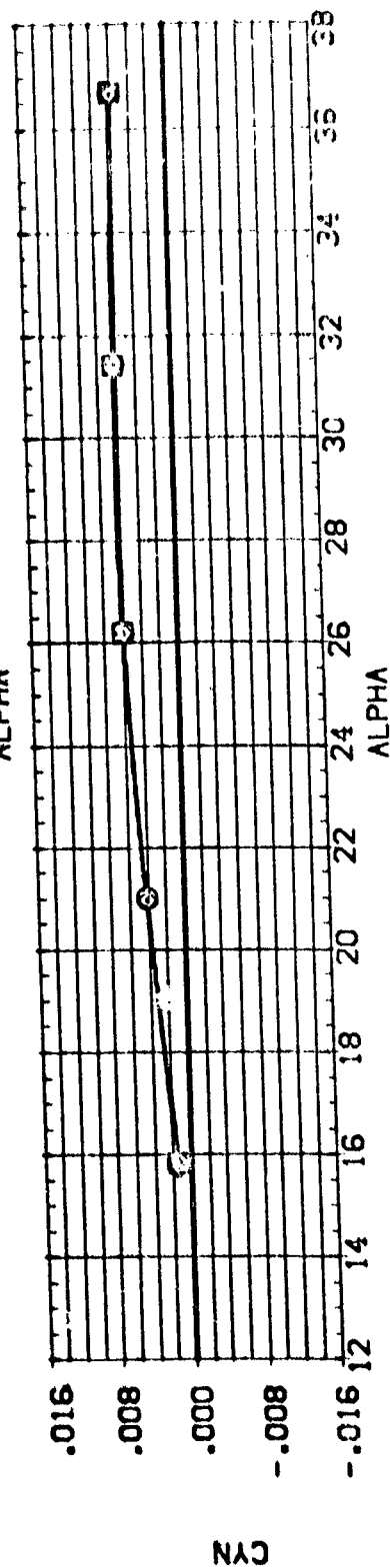
4.500.000.000.

FILE NO. 100-33364

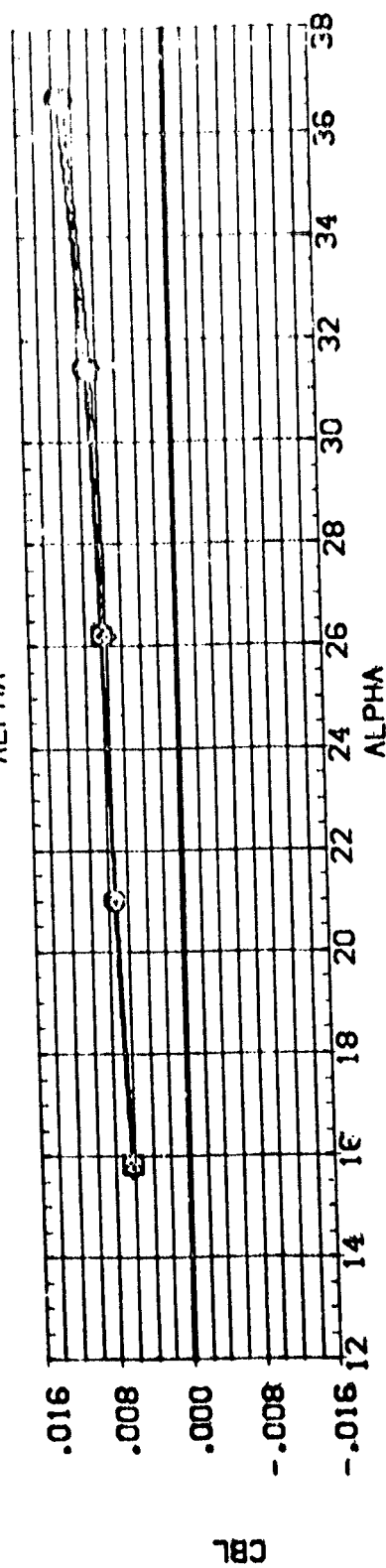
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------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------	------



13



CYN



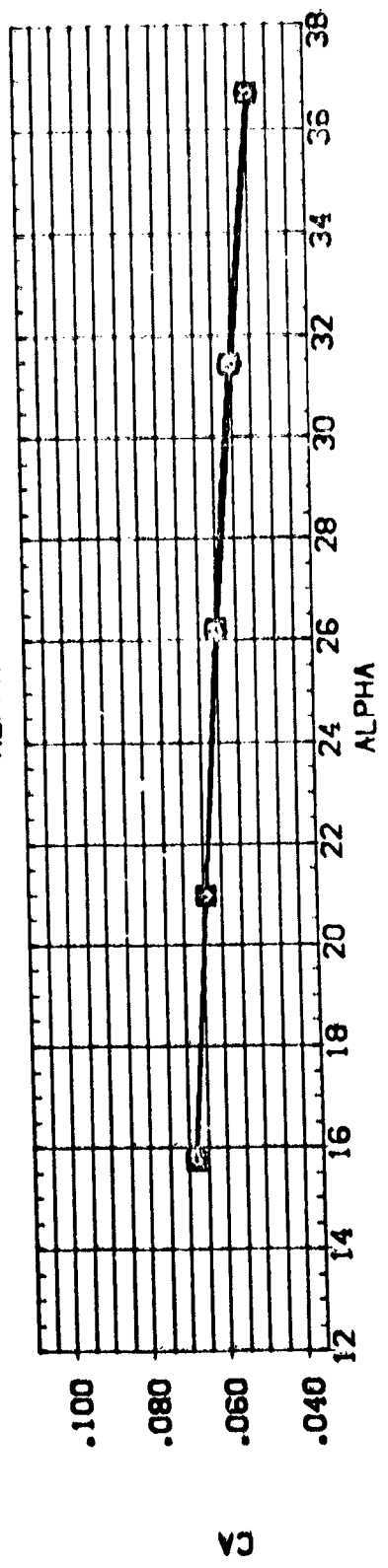
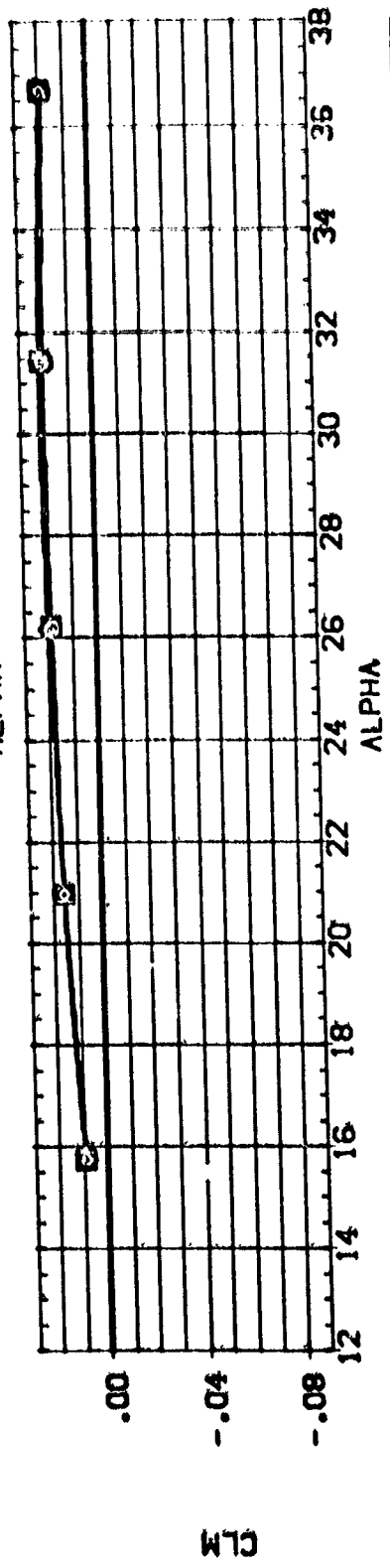
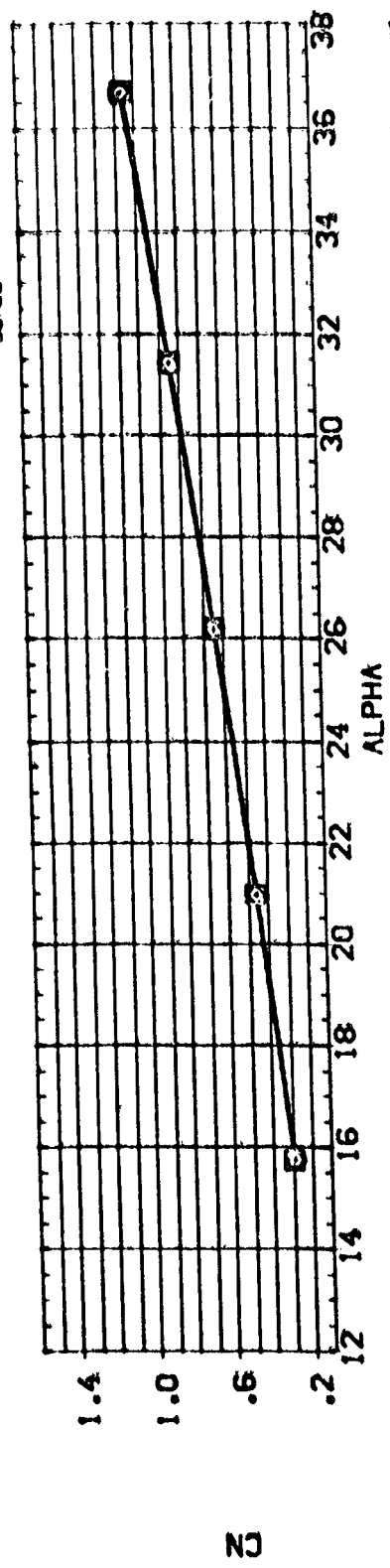
782

EFFECT OF RES ON ORBITER AERØ. CHARACT. (BETA=-5, MACH=4.6)



CA-70. UPWT1043. ORB(B19C7F5M6N19)(W107E23)(V7R5)(RPV013)

SYMBOL	PG-JET	BETA	PARAMETRIC VALUES			REFERENCE INFORMATION		
010	.042	RMAL	-5.000	MACH	4.600	SREF	87.1550	50.11
011	69.414	AILRON	1.720	ELEVTR	-20.000	REF	7.1222	1045
012	163.401	RJFLR	.030	BOFLAP	-14.250	REF	14.7500	1045
			40.000	RJFLR	.008	REF	12.5800	1045
						REF	6.0000	1045
						SCALE	.0150	



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=4.6)

GA-70. UPWT1043.CRB(B19C7F5M6N19)(W107E23)(V7R5)(RPV013)

SYMBOL

○ □ ◇

PG-LET

.042  
69.414  
163.401

BETA

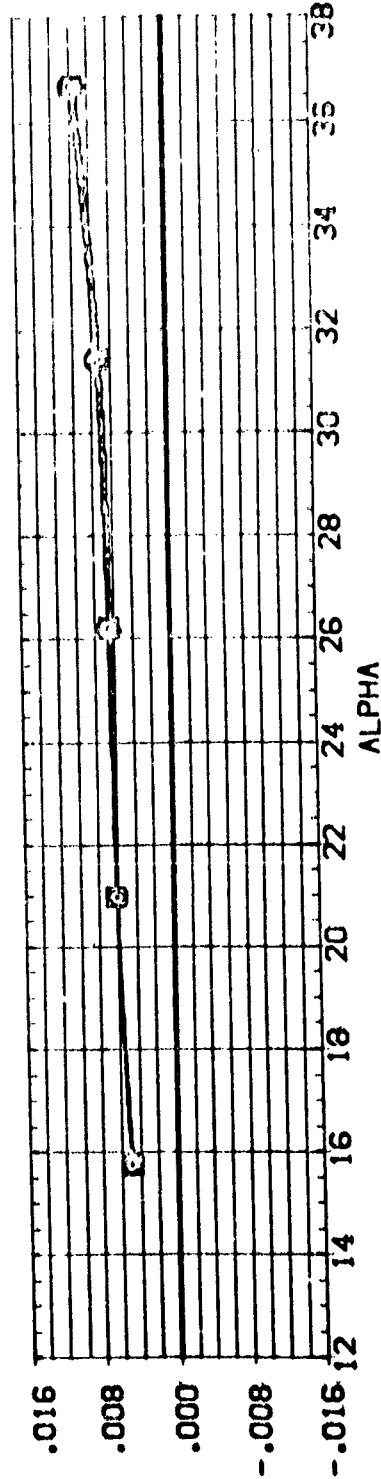
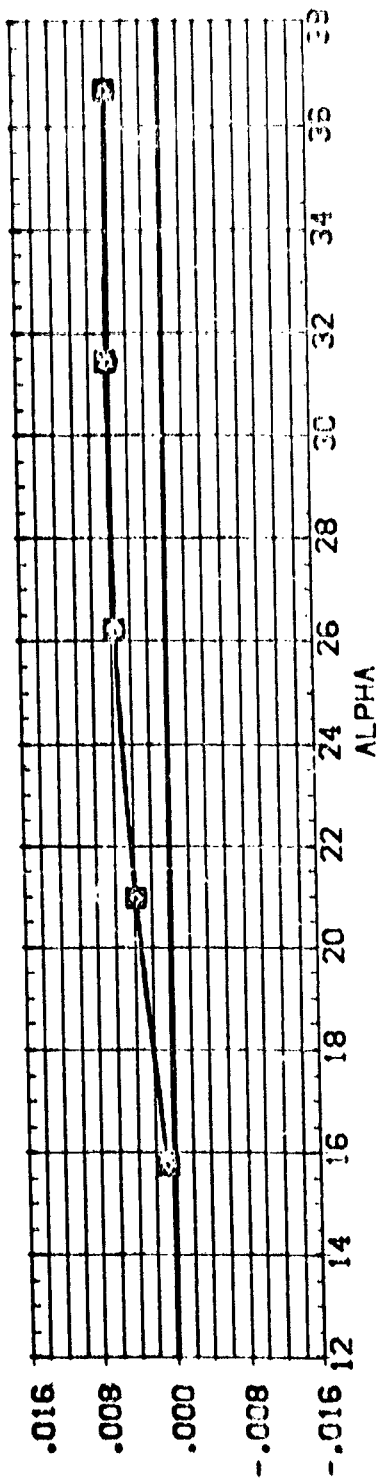
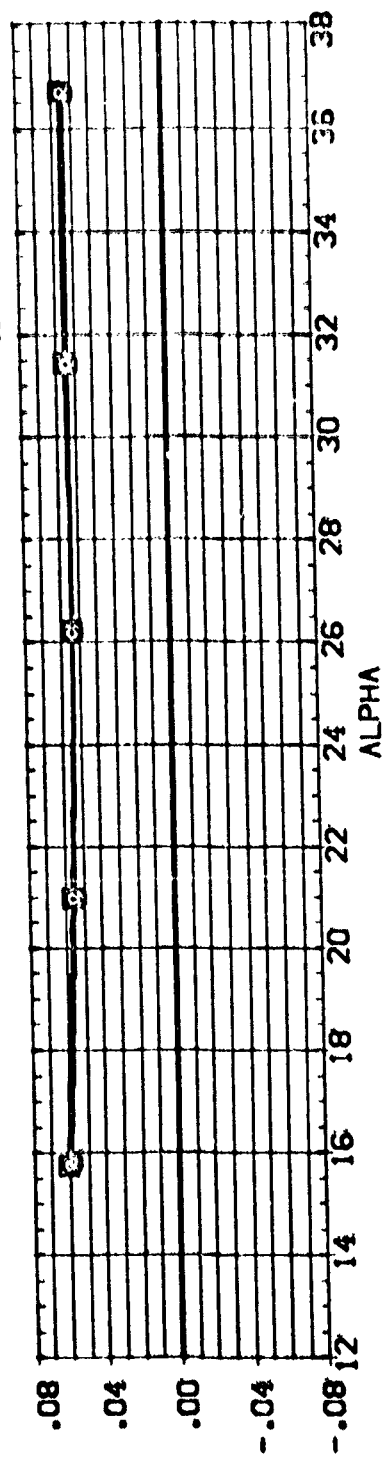
RVL  
A1LRN  
R0FLR

PARAMETRIC VALUES

-5.000 MACH 4.600  
1.720 ELEVTR -20.000  
.000 BOFLAP -14.250  
40.000 PLODER .000

REFERENCE INFORMATION

SREF 87.1582  
LREF 7.1372  
SPREF 14.2500  
XREF 12.5800  
YREF 20.0000  
ZREF 6.0000  
SCALE 0.150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=4.6)



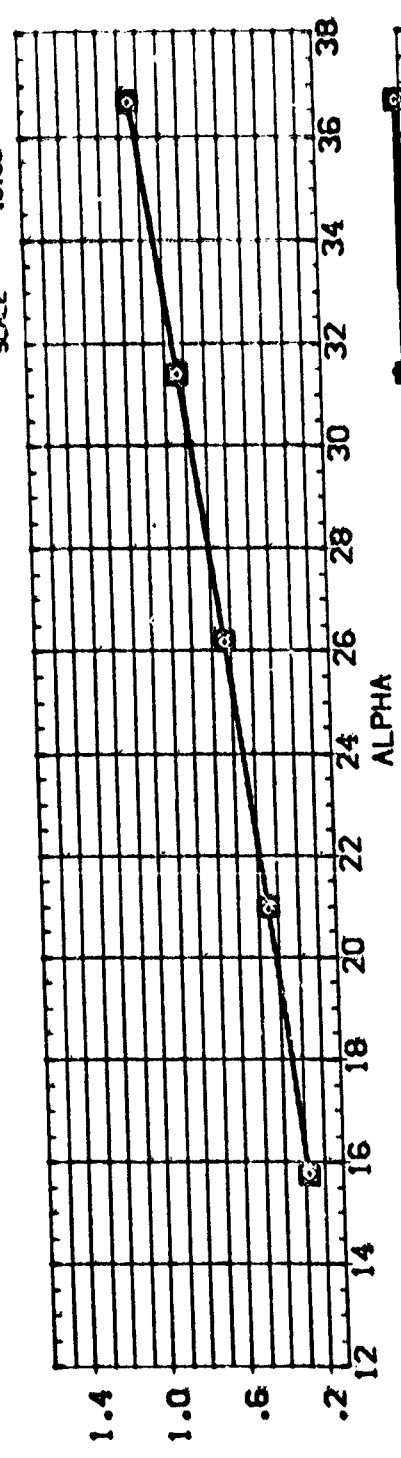
GA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV016)

PO-JET	BETA	PARAMETRIC VALUES	
.112		-5.000	MACH 4.600
71.943	RV/L	1.720	ELEVTR -40.000
165.035	ALLRON	.000	BOFLAP -14.253
	RDOFLR	40.000	RDOER .000

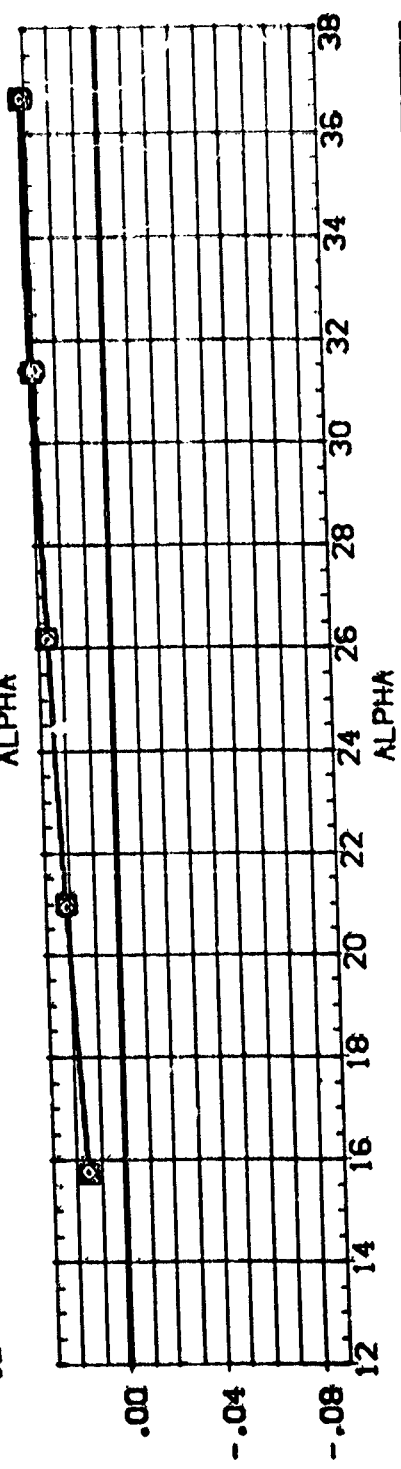
  

REFERENCE INFORMATION	
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ZREF	6.0000
SCALE	.0150

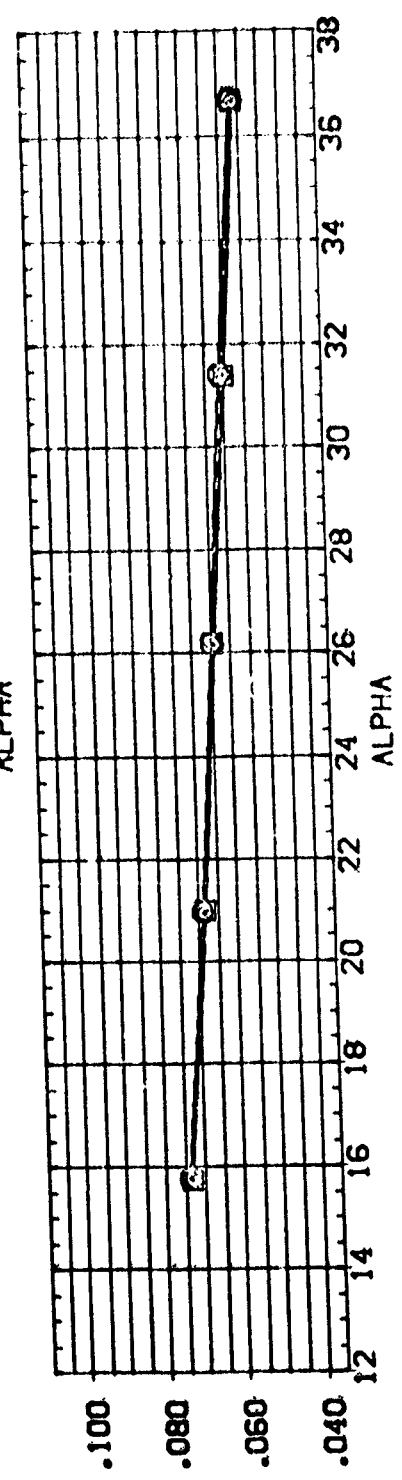
SYMBOL  
 ○ □ ◇



Z



X



Y

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=4.6)

0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV016)

SYMBOL  
□  
◇

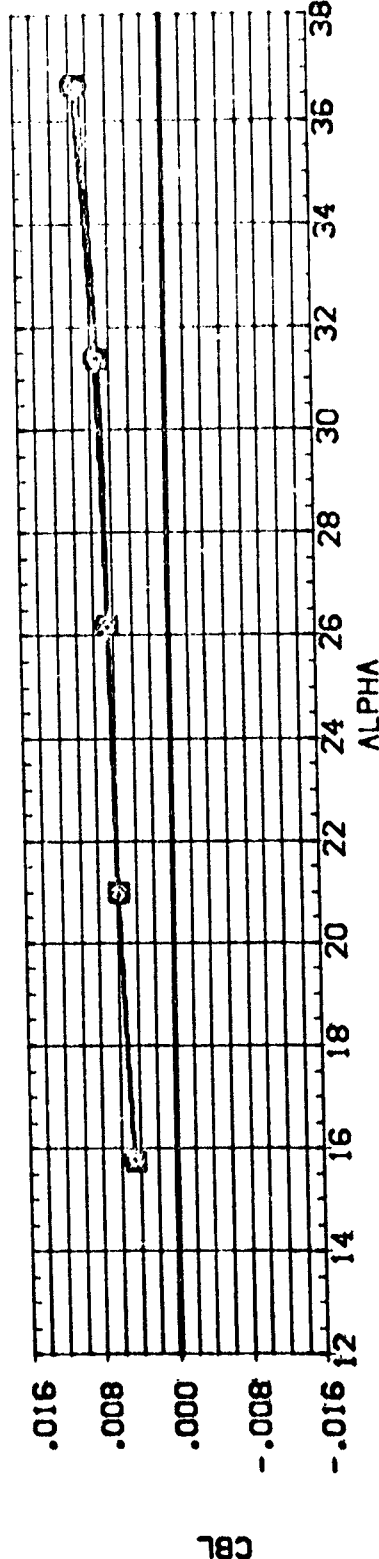
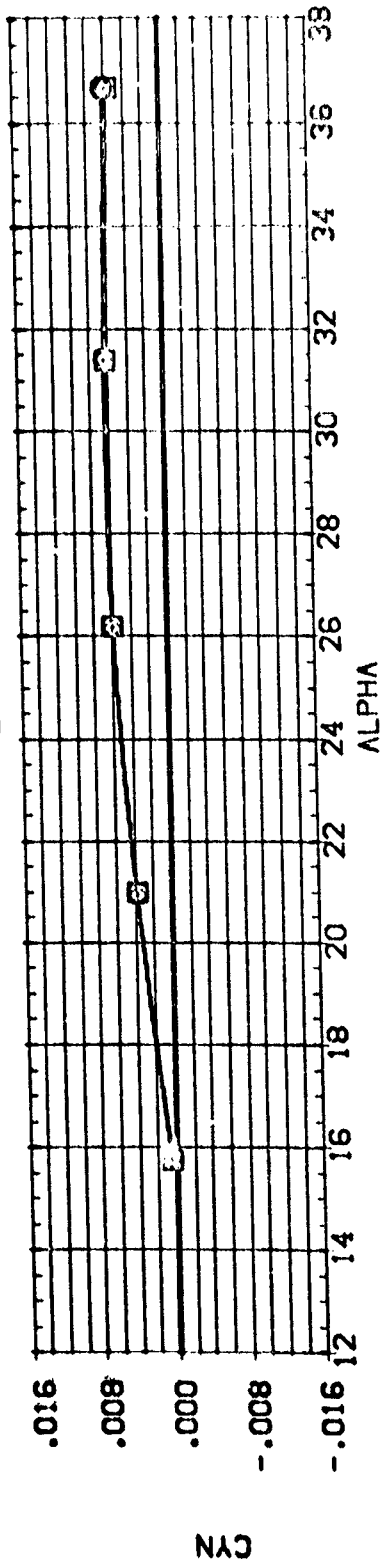
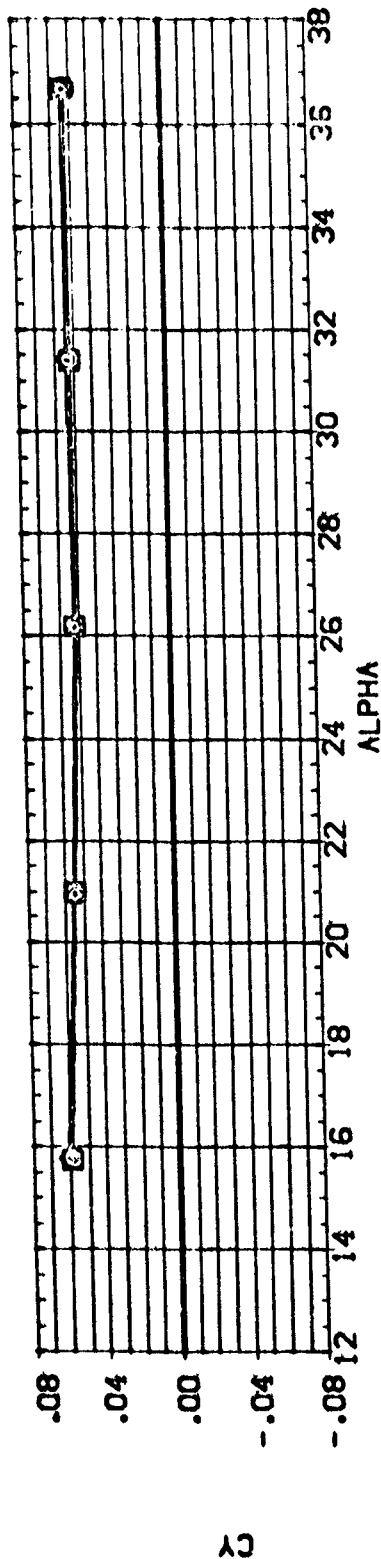
PG-JET  
.112  
71.943  
165.035

BETA  
RVL  
AILRON  
RLOFLR

PARAMETRIC VALUES  
-5.000 MACH  
1.720 ELEVTR  
.000 BOFLAP  
40.000 RUDDER

4.600  
-40.000  
-14.250  
.000

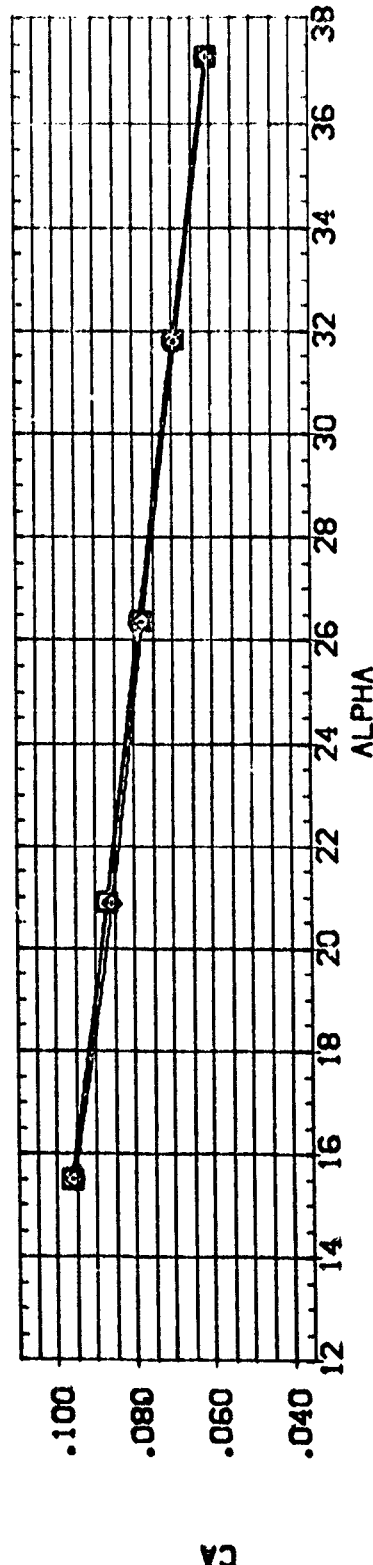
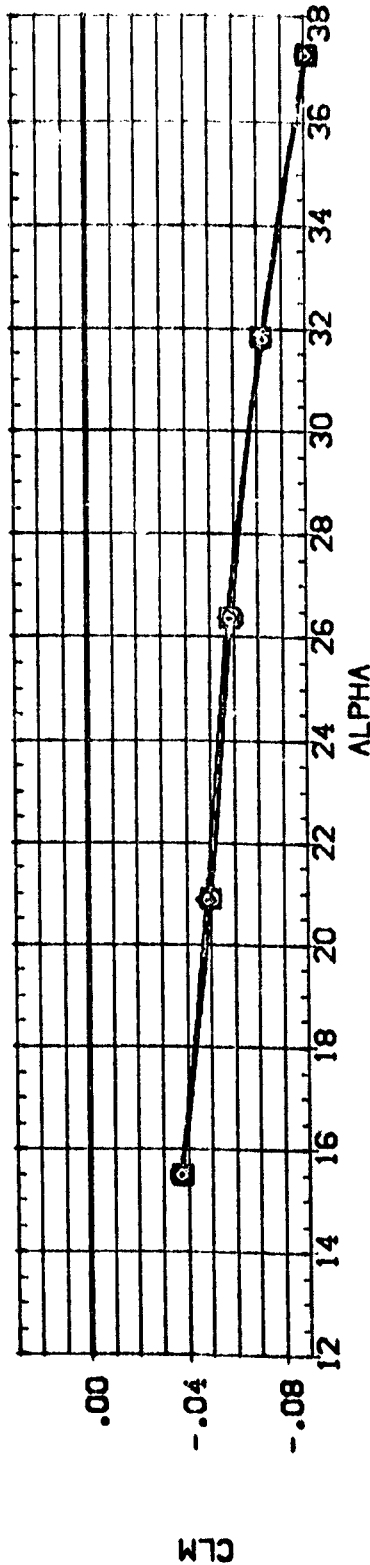
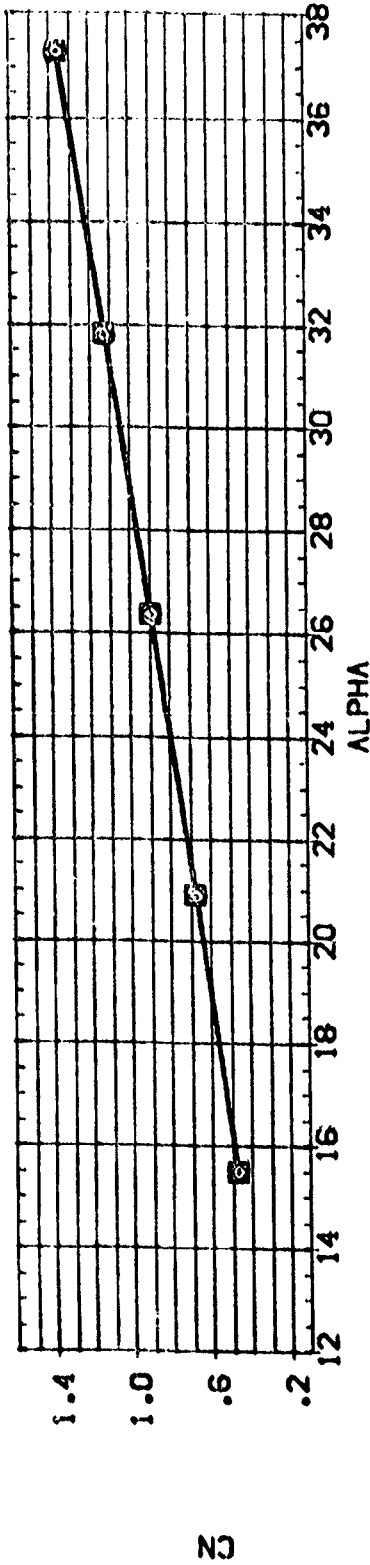
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SREF 87.1560 SC-IN.  
LREF 7.1722 SC-ES  
BREF 14.2500 SC-ES  
XREF 12.5800 SC-ES  
YREF .0000 SC-ES  
ZREF 6.0000 SC-ES  
SCALE .0150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=-5, MACH=4.6)

0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV00C3)

SYMBOL	PB-JET	BETA	PARAMETRIC VALUES			REFERENCE INFORMATION		
			5.000	MACH	2.500	SREF	87.1560	SO.IN.
○	69.240	RM/L	1.720	ELEVTR	.000	LREF	7.1272	INC-ES
□	223.649	AILRON	.000	BOFLAP	.000	BREF	14.0500	INC-ES
◇		RUEFLR	40.000	RLODER	.000	XMRP	12.5800	INC-ES
						YMRP	.0000	INC-ES
						ZMRP	6.0000	INC-ES
						SCALE	.0150	

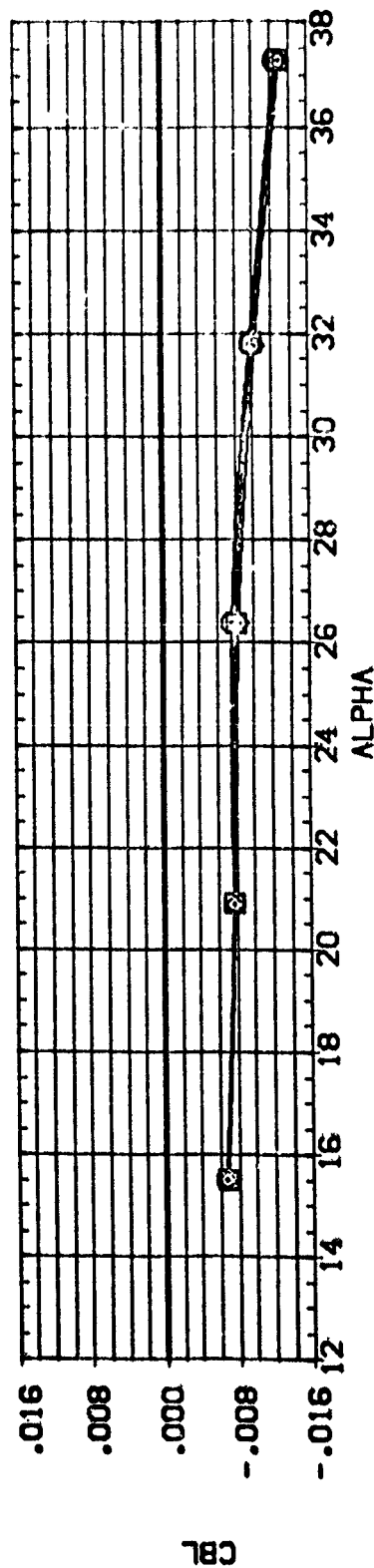
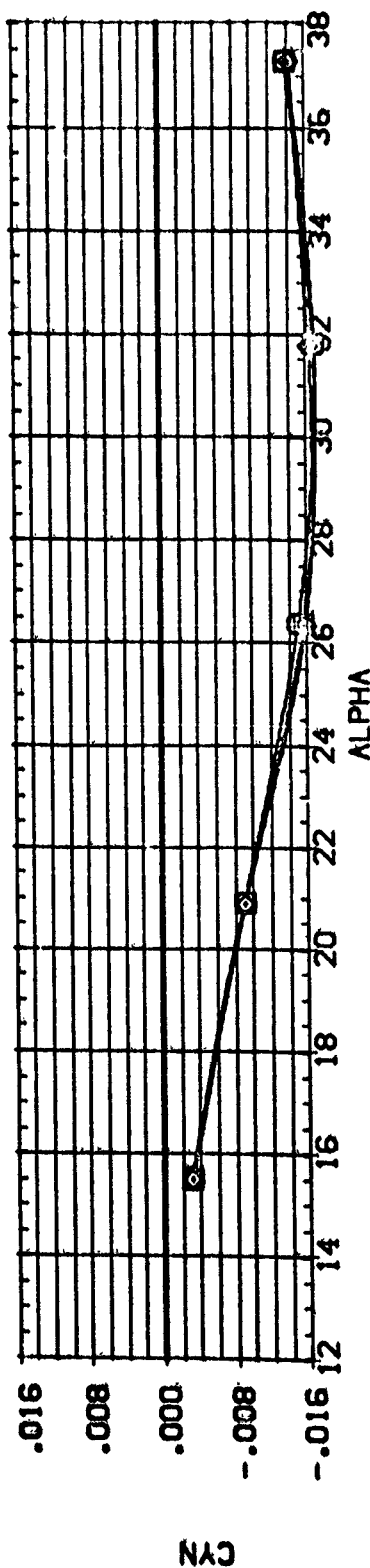
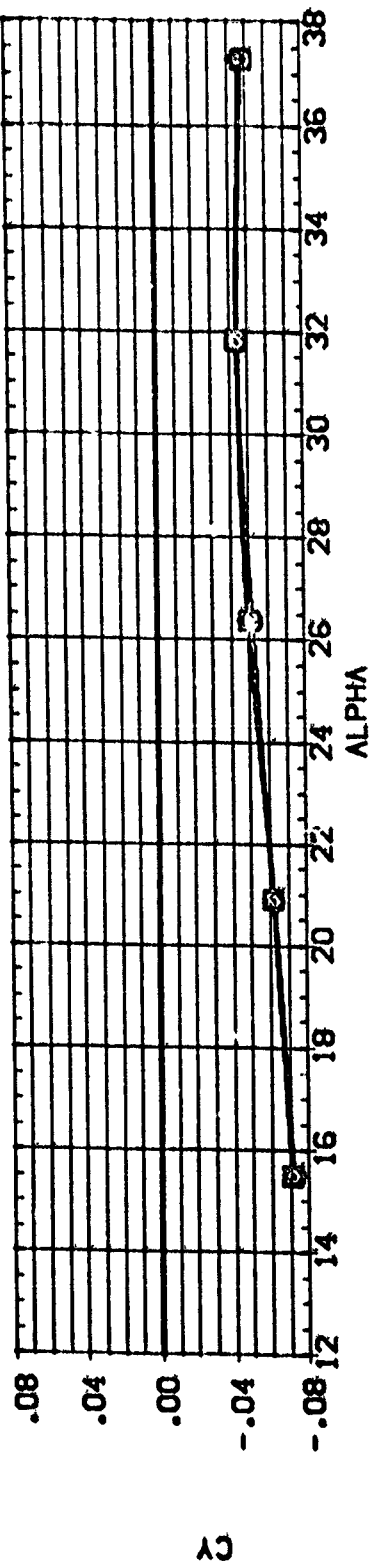


EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=2.5)



QA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV003)

SYMBOL	PG-JET	BETA	PARAMETRIC VALUES			REFERENCE INFORMATION				
			5.000	MACH	2.500	SREF	87.1560	50. IN.		
○	68.240	RV/L	1.720	ELEVTR	.000	LOEF	7.1222	INCHES		
□	223.649	ALLRON	.000	EDFLAP	.000	BREF	14.0500	INCHES		
◇		R-OF-FLR	40.000	RUBBER	.000	XPRP	12.5800	INCHES		
						VPRP	.0000	INCHES		
						ZPRP	6.0000	INCHES		
						SCALE	.0150			



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=2.5)



CA-70, UPWT1043, ORB(819C7F5M6N19)(W107E23)(V7R5)(RPV006)

SYMBOL



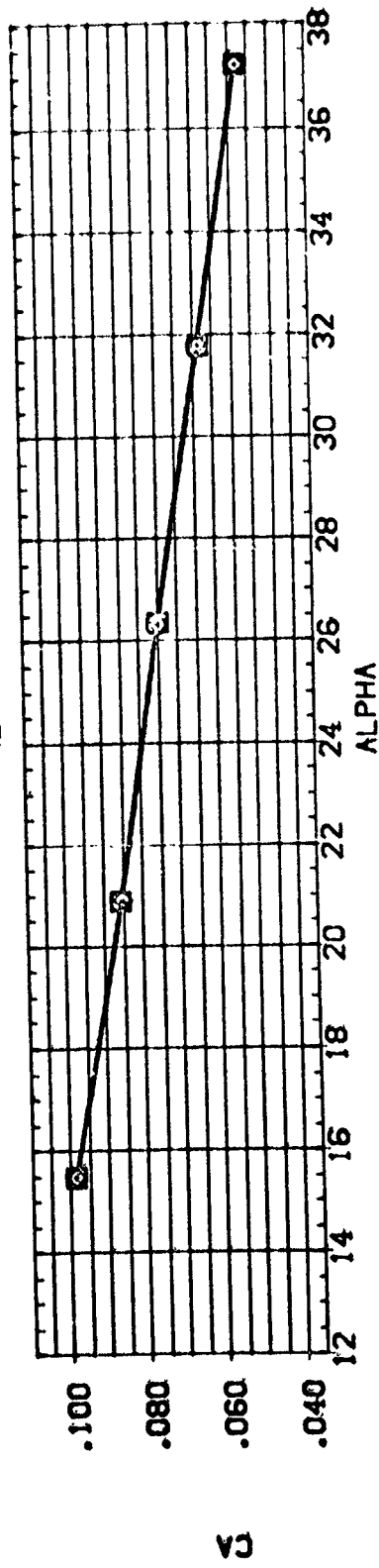
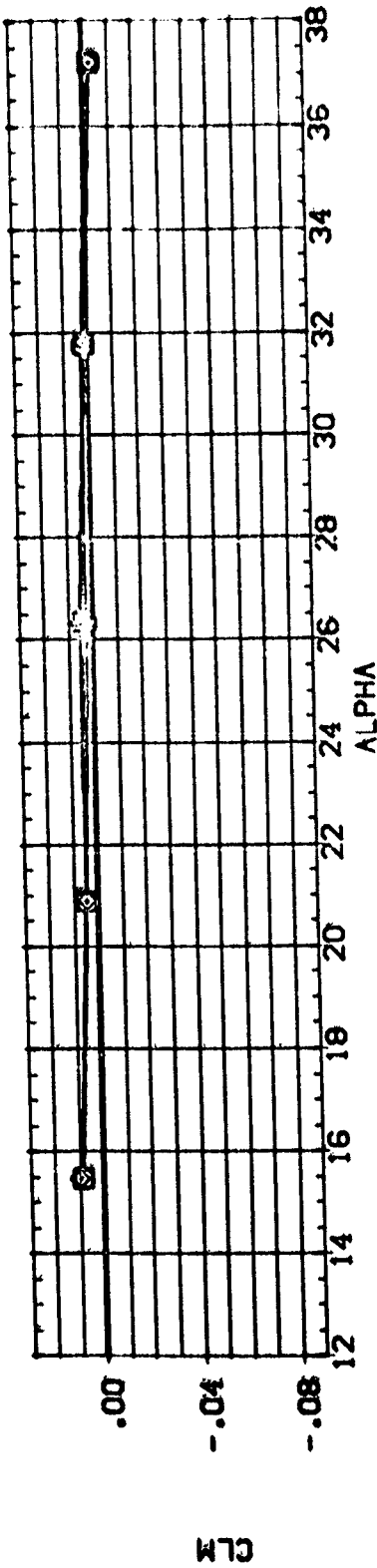
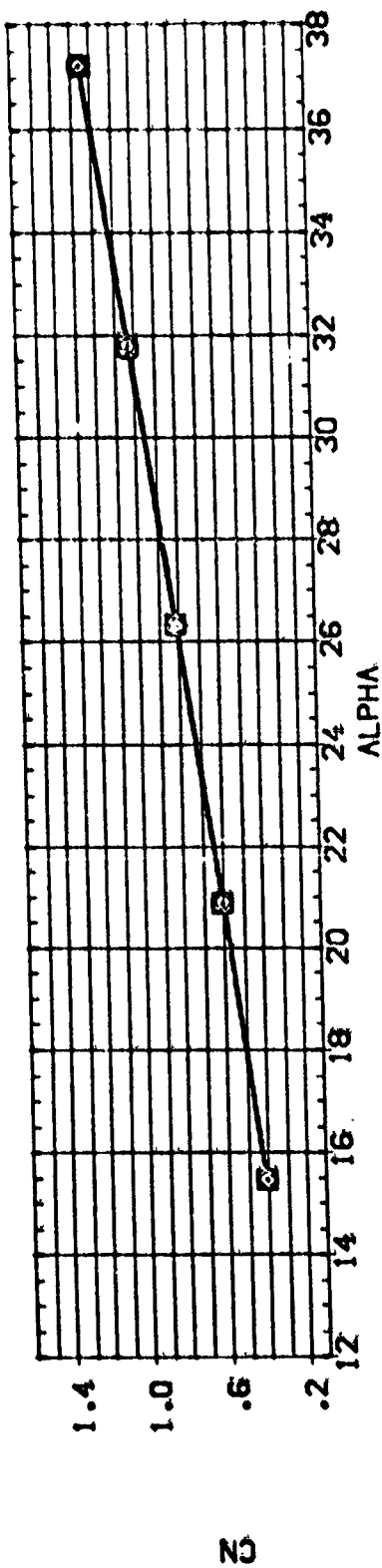
PO-JET  
.276  
63.159  
228.534

BETA  
RW/L  
AILRON  
RUOFLR

PARAMETRIC VALUES  
5.000 MACH  
1.770 ELEVTR  
.000 BOFLAP  
40.000 RUOFLR

2.500  
-20.000  
-14.250  
.000

REFERENCE INFORMATION  
SREF 87.1560  
LREF 7.1222  
BREF 14.0500  
XREF 2.5800  
YREF 0.0000  
ZREF 6.0000  
SCALE 0.0150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=2.5)

0A-70. UPWT1043.0R8(B19C7F5M6N19)(W107E23)(V7R5)(RPV0006)

SYMBOL

□  
□  
◇

PO-JET

.276  
63.159  
228.534

BETA

5.000  
1.720  
.000

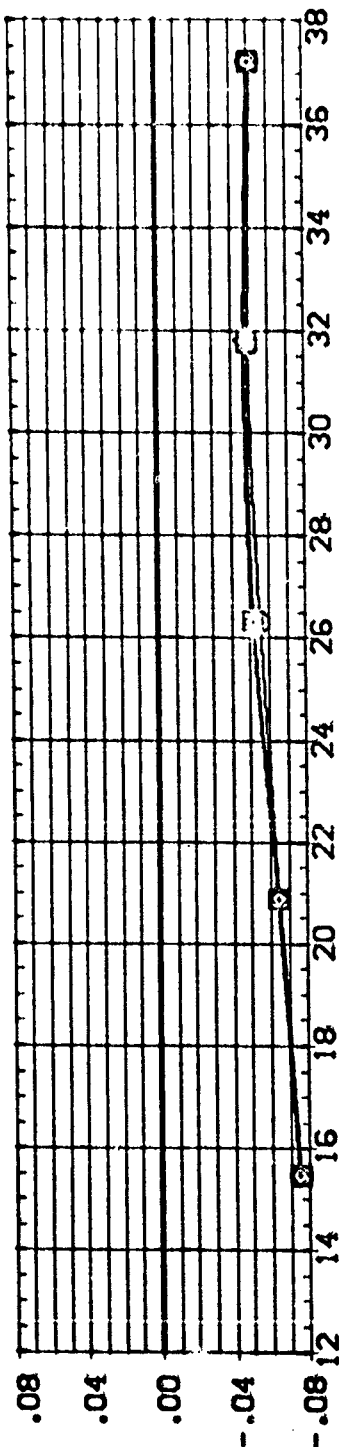
PARAMETRIC VALUES

MACH  
ELEVTR  
BOFLAP  
RUDDER

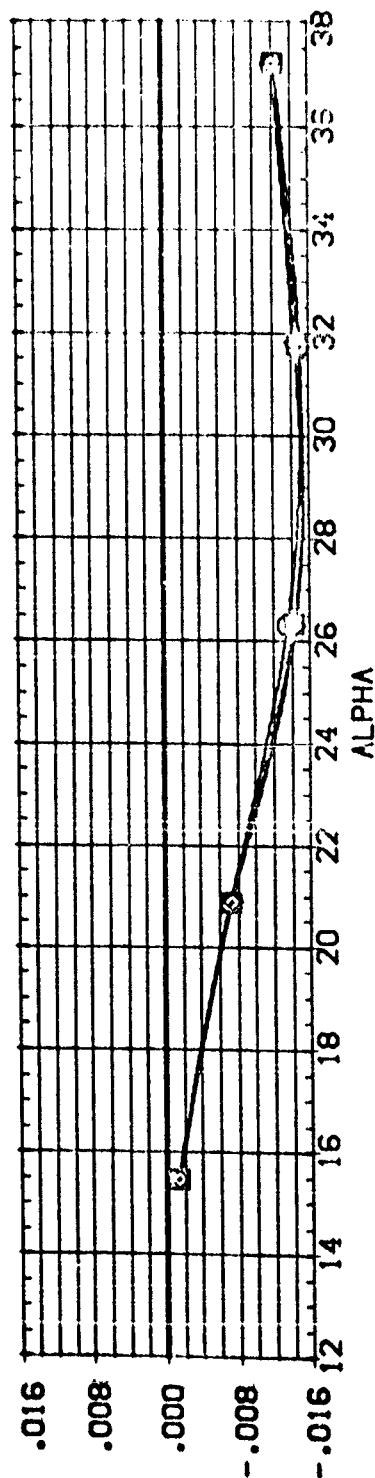
2.500  
-20.000  
-14.250  
.000

REFERENCE INFORMATION

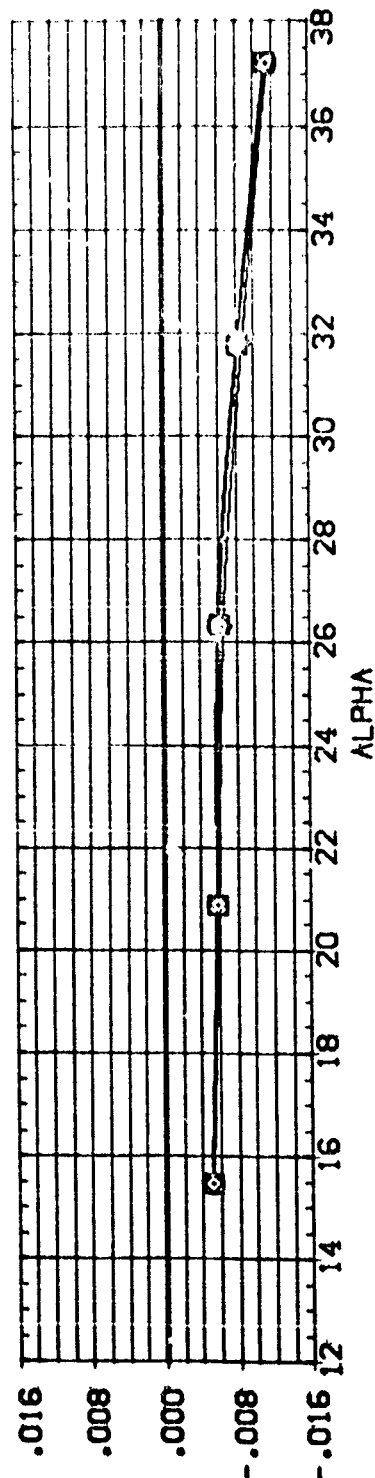
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XREF 12.5800  
YREF .0000  
ZREF 6.0000  
SCALE .0150



CY



CYN



CBL

EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=2.5)



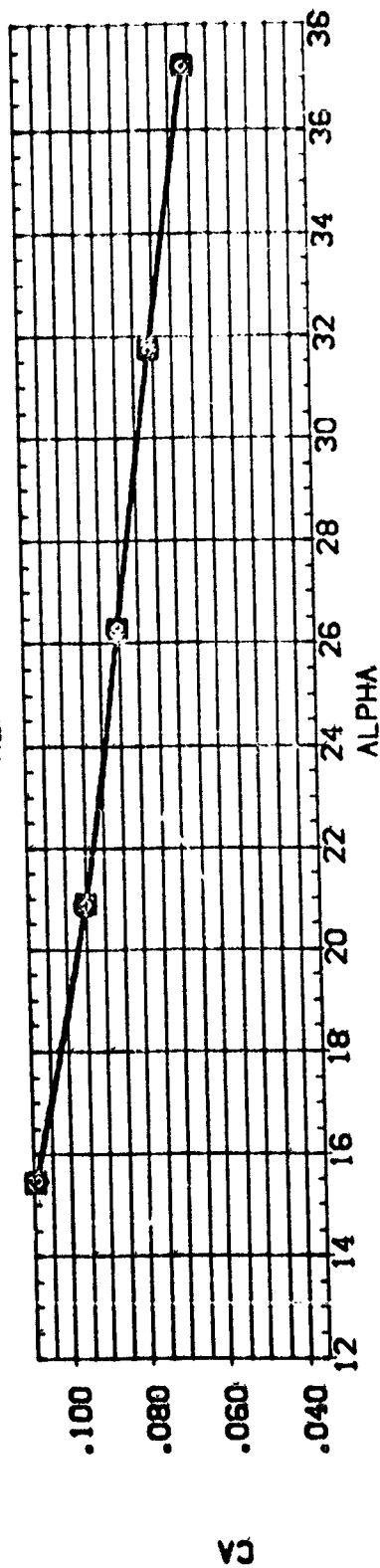
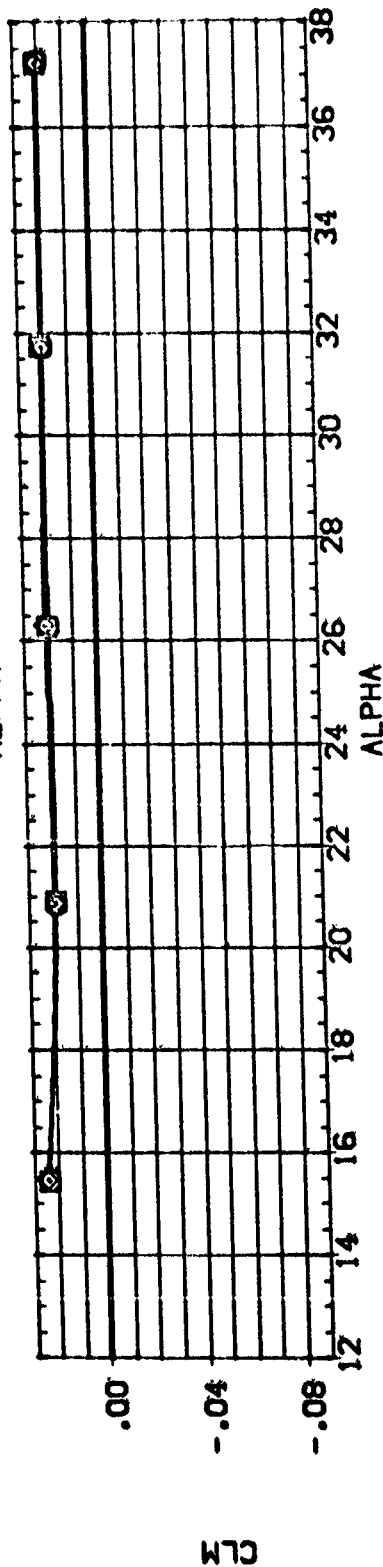
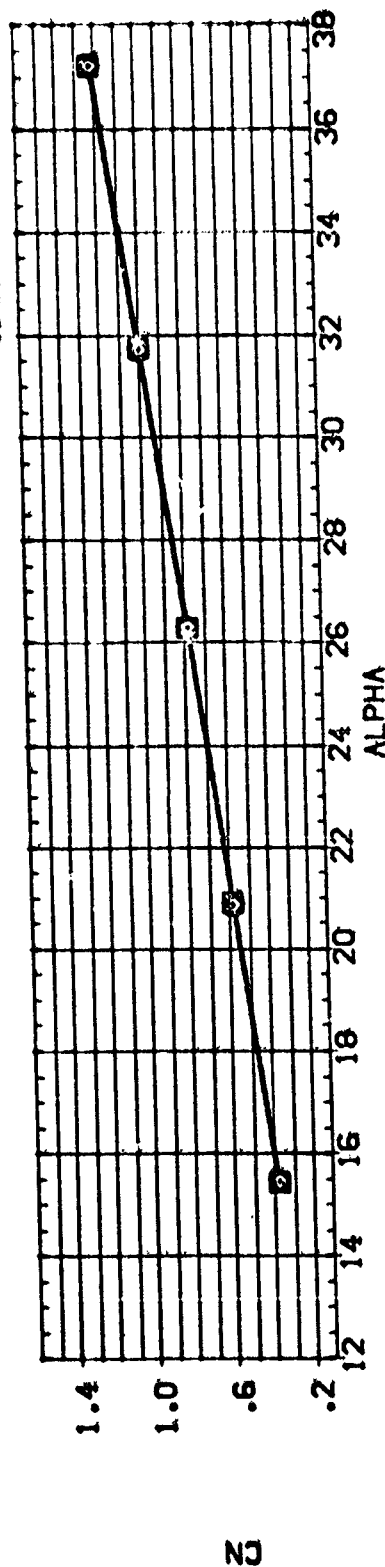
0A-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV009)

SYMBOL  
□  
□  
◇

PO-JET BETA  
.562  
63.892  
277.027

PARAMETRIC VALUES  
- 5.000 MACH 2.500  
1.720 ELEVTR -40.000  
.000 BDFLAP -14.250  
40.000 RJOER .000

REFERENCE INFORMATION  
SPREF 97.1560 SQ. IN.  
LREF 7.1272 INC-ES  
BPREF 14.0500 INC-ES  
XPRP 12.5800 INC-ES  
YPRP .0000 INC-ES  
ZPRP 6.0000 INC-ES  
SCALE .0150



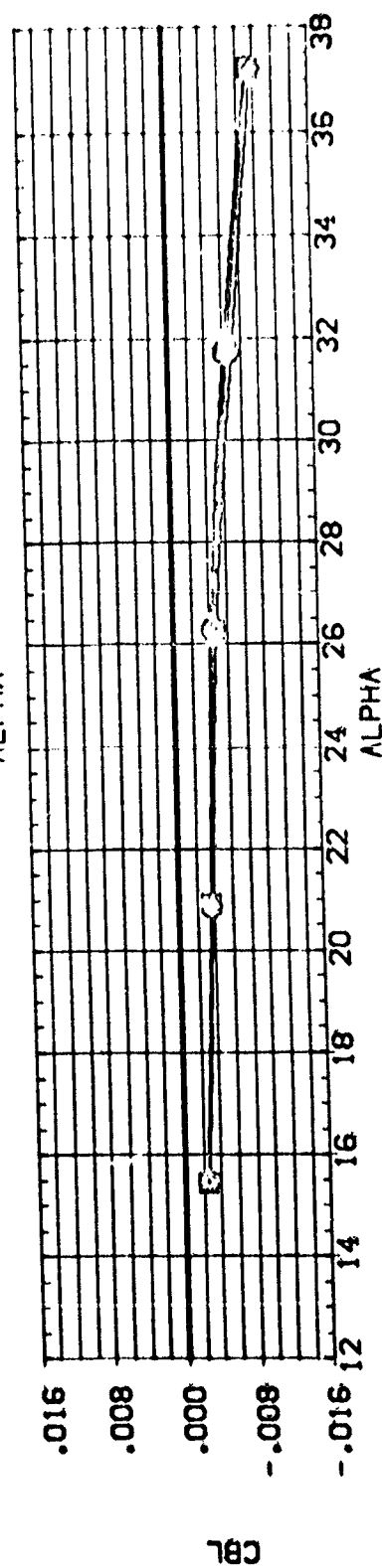
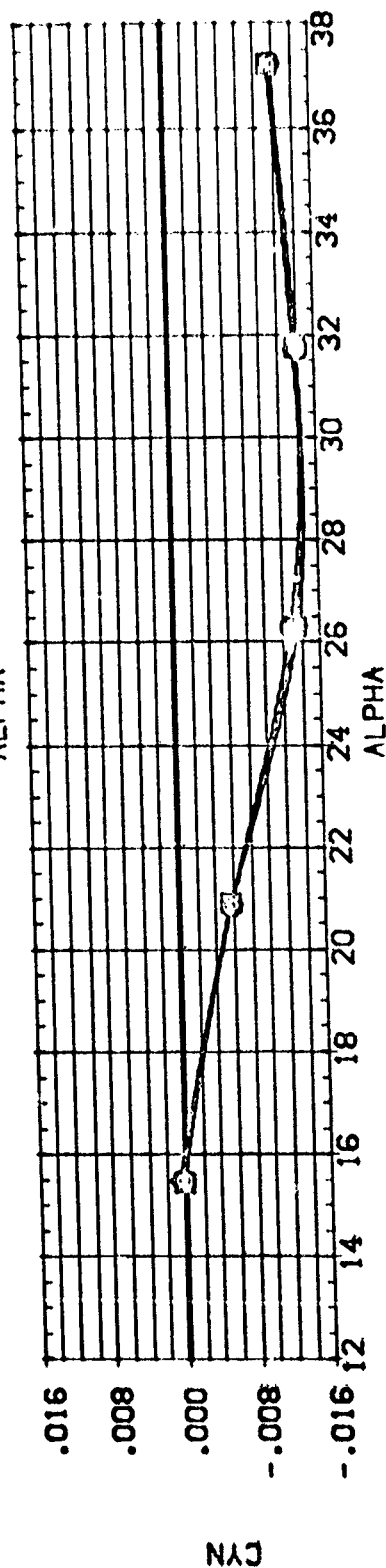
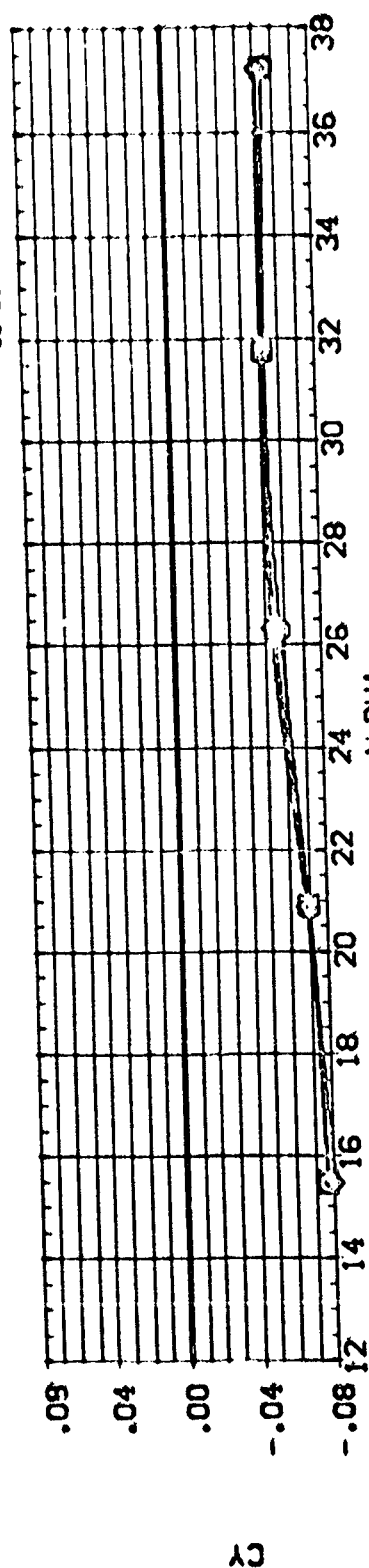
EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=2.5)

CA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV009)

SYMBOL	POW-ET	PARAMETRIC VALUES			
		BETA	MACH	ELEVTR	BOFLAP
011	.562	5.000	2.500	-40.000	
012	63.892	1.720		-14.250	
013	777.027	.000	40.000		.000

REFERENCE INFORMATION

SREF	SO-IN
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7.1222	NGES
14.0500	NGES
12.5800	NGES
.5000	NGES
6.0000	NGES
.0150	NGES

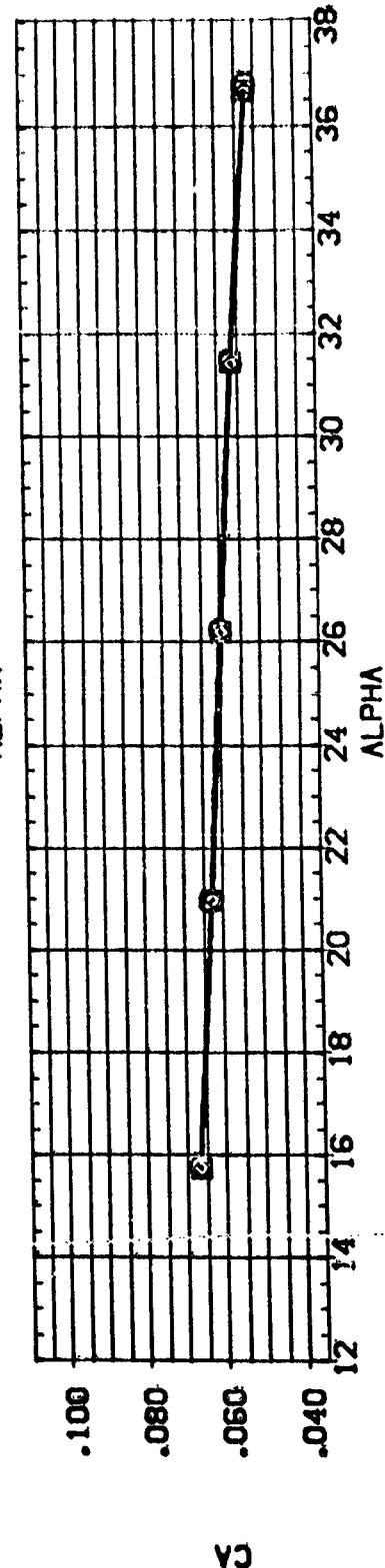
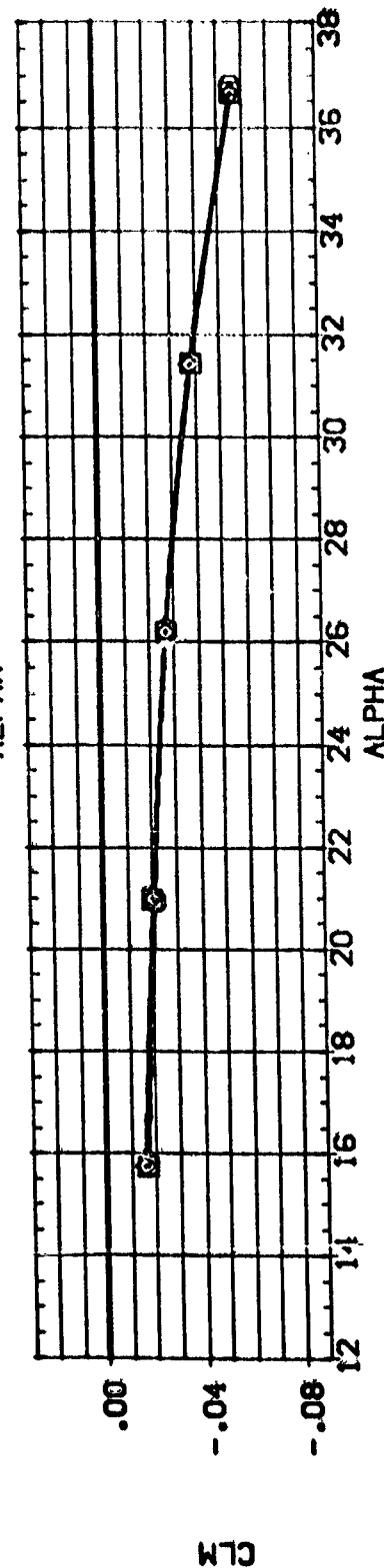
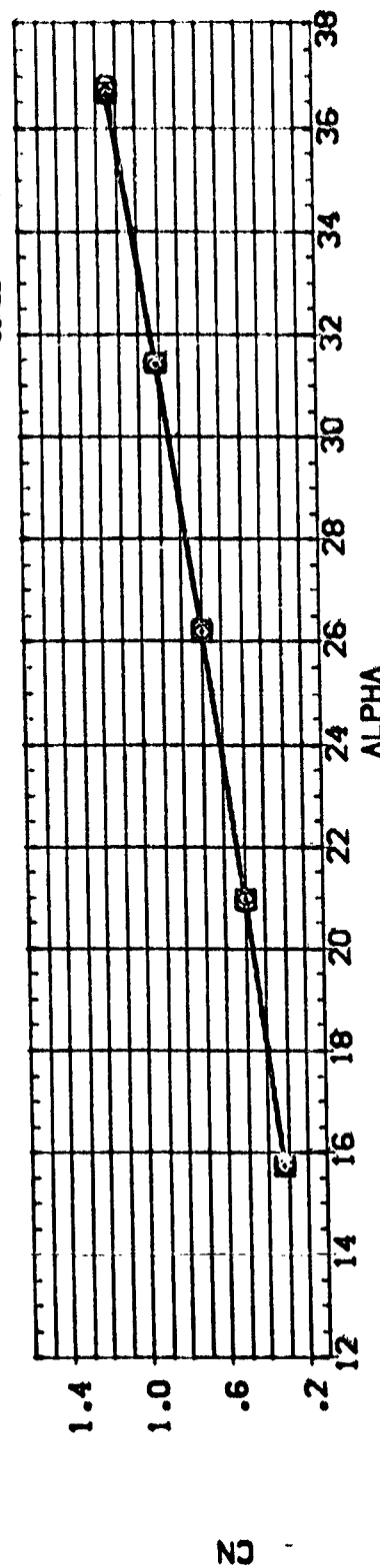


EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=2.5)



GA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV012)

SYMBOL	PG-JET	PARAMETRIC VALUES				REFERENCE INFORMATION			
		BETA	5.000	MACH	4.600	SREF	87.1560	SD.IN	87.1560
□	.317	RV/L	1.720	ELEVTR	.000	LREF	7.1272	NG-ES	NG-ES
□	58.277	AILFON	.000	BOFLAP	.000	BREF	14.0500	NG-ES	NG-ES
◇	163.231	RUFPLR	40.000	RUDDER	.000	YMRP	12.5800	NG-ES	NG-ES
						ZMRP	6.0000	NG-ES	NG-ES
						SCALE	.0150		



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=4.6)

OA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV012)

SYMBOL

PO-JET

.317

BETA

5.000

MACH

4.600

□

58.277

RN/L

1.720

ELEVTR

.000

◇

163.231

ALLRBL

.000

BOFLAP

.000

RJDFLR

40.000

RUDER

.000

REFERENCE INFORMATION

SC. IN.

87.1560

INCHES

7.1272

INCHES

14.0500

INCHES

12.5800

INCHES

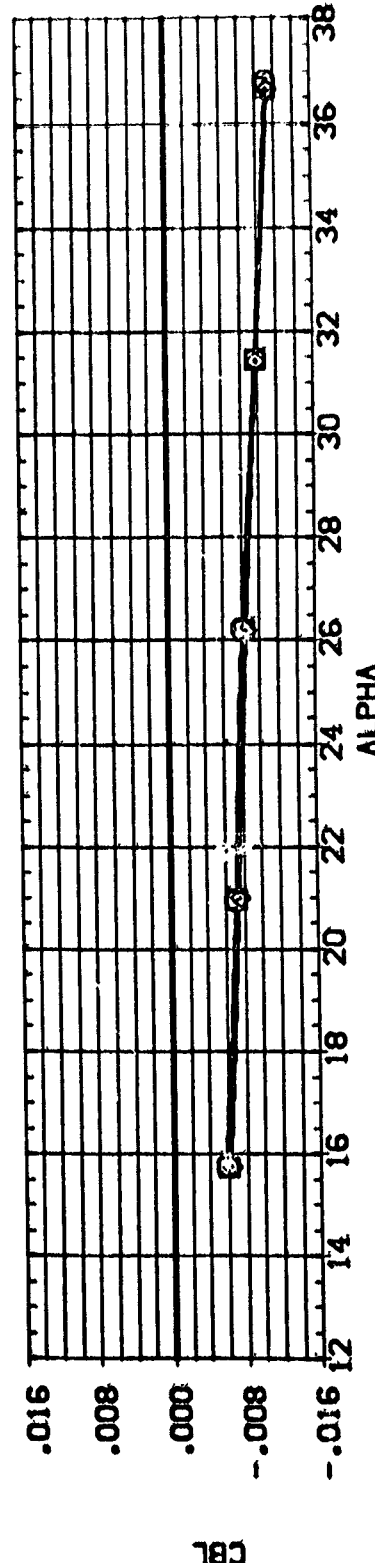
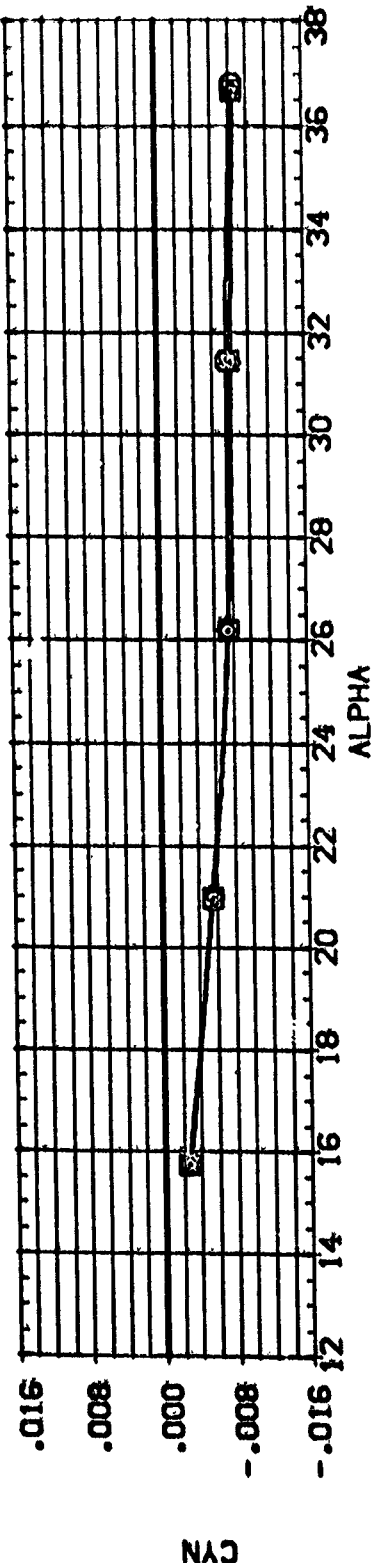
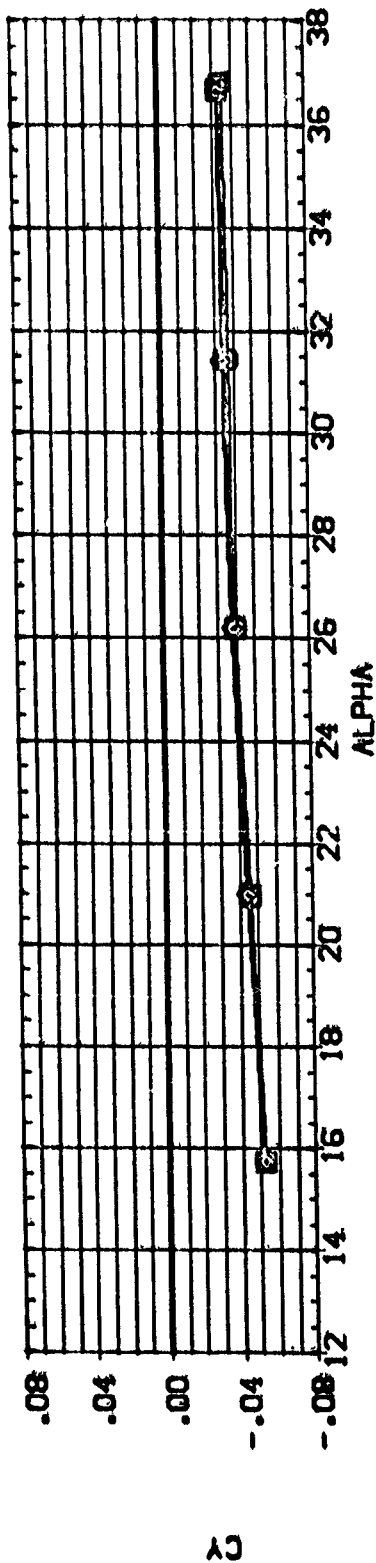
.0000

INCHES

6.0000

INCHES

.0150



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=4.6)



CA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV015)

SYMBOL  
 ○ □ ◇

PO-JET  
 .268  
 70.088  
 161.614

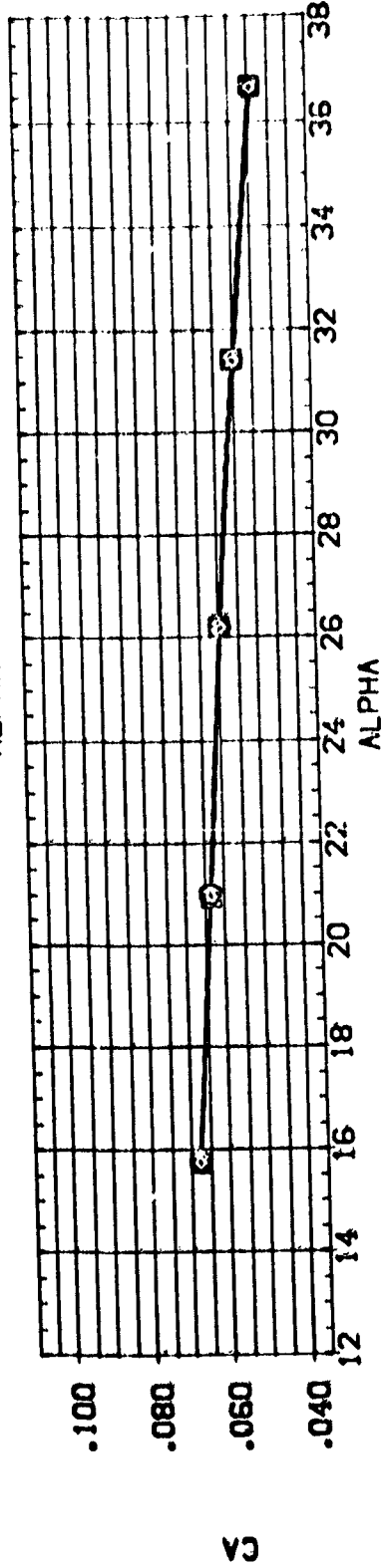
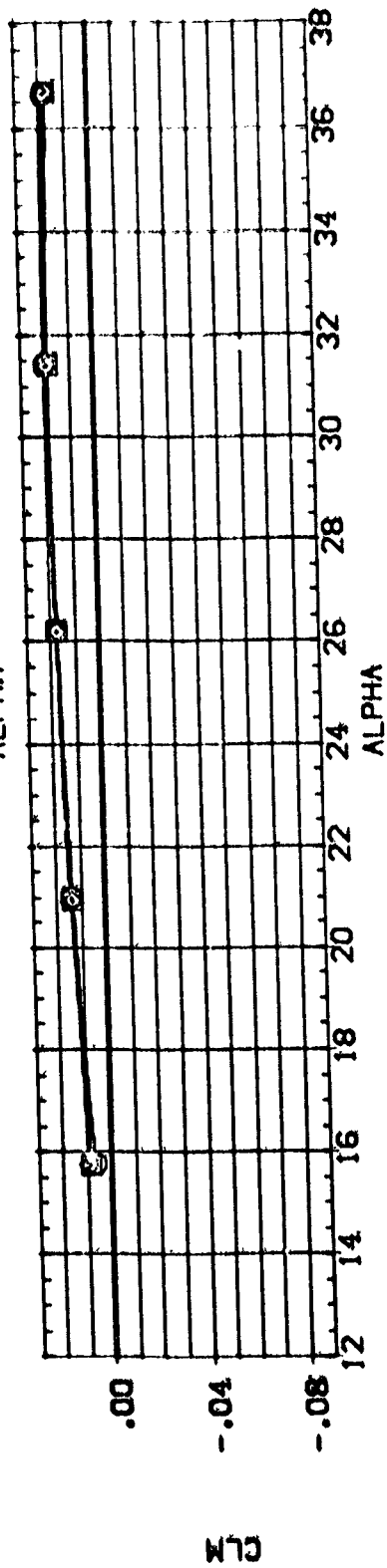
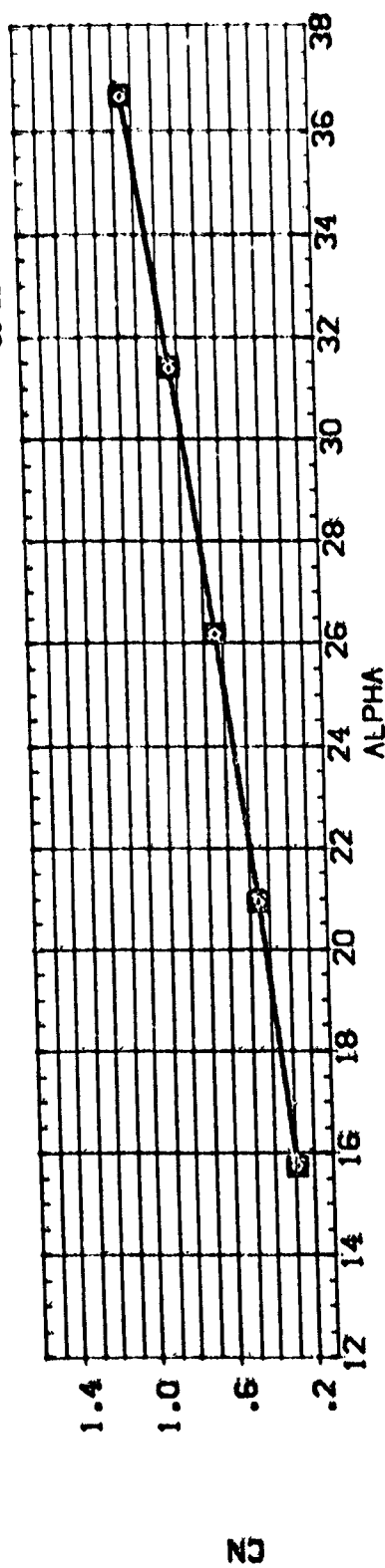
BETA  
 RV/L  
 AILRON  
 RLOFLR

PARAMETRIC VALUES  
 5.000 MACH  
 1.720 ELEVTR  
 .000 BOFLAP  
 40.000 RAJDER

4.600  
 -20.000  
 -14.250  
 .000

REFERENCE INFORMATION  
 SREF 87.1560  
 LREF 7.1222  
 BREF 14.0500  
 XREF 2.5800  
 YREF 1.0000  
 ZREF 6.0000  
 SCALE 0.150

SQ LIN  
 NC-ES  
 NC-ES  
 NC-ES  
 NC-ES  
 NC-ES

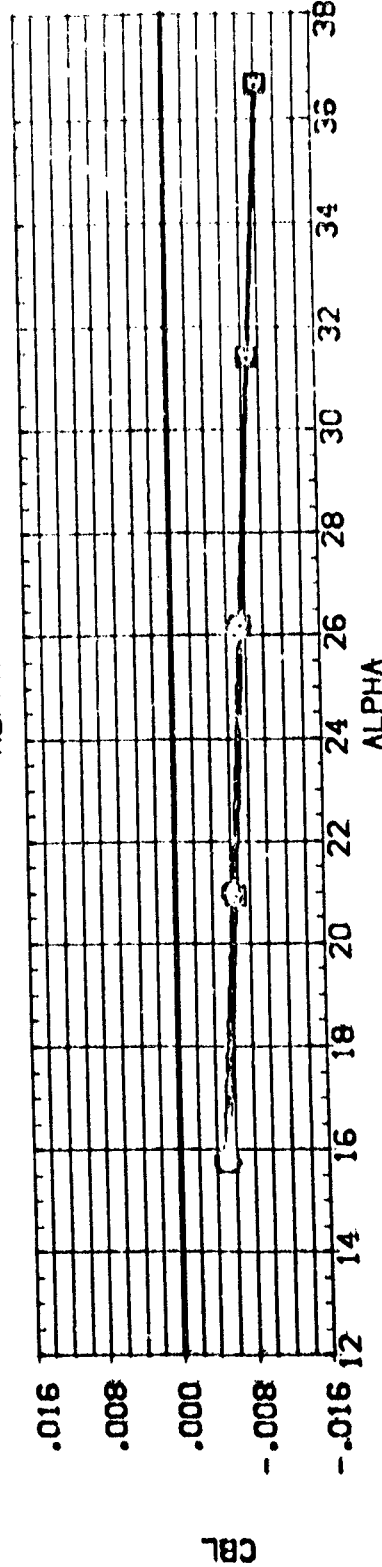
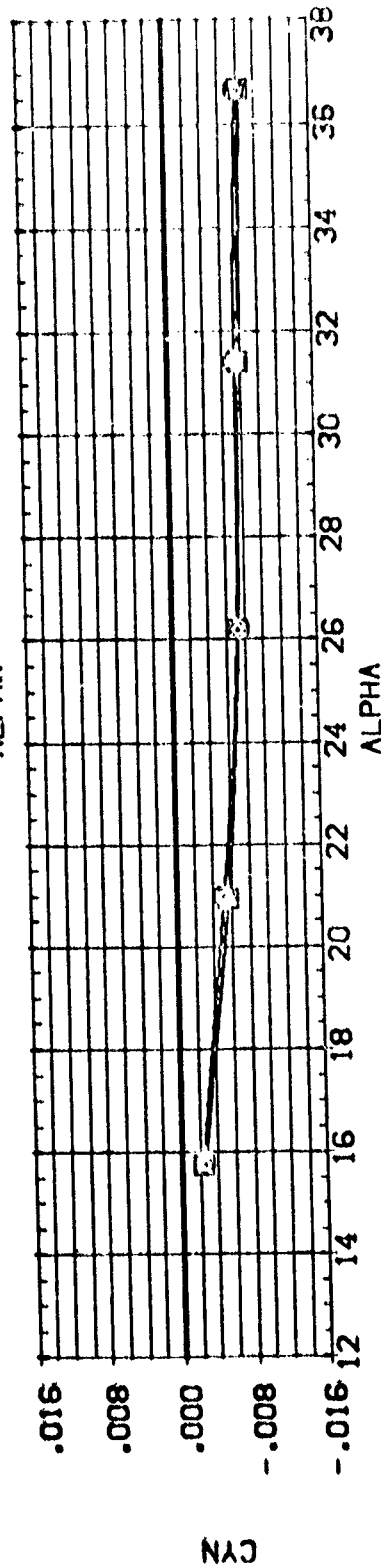
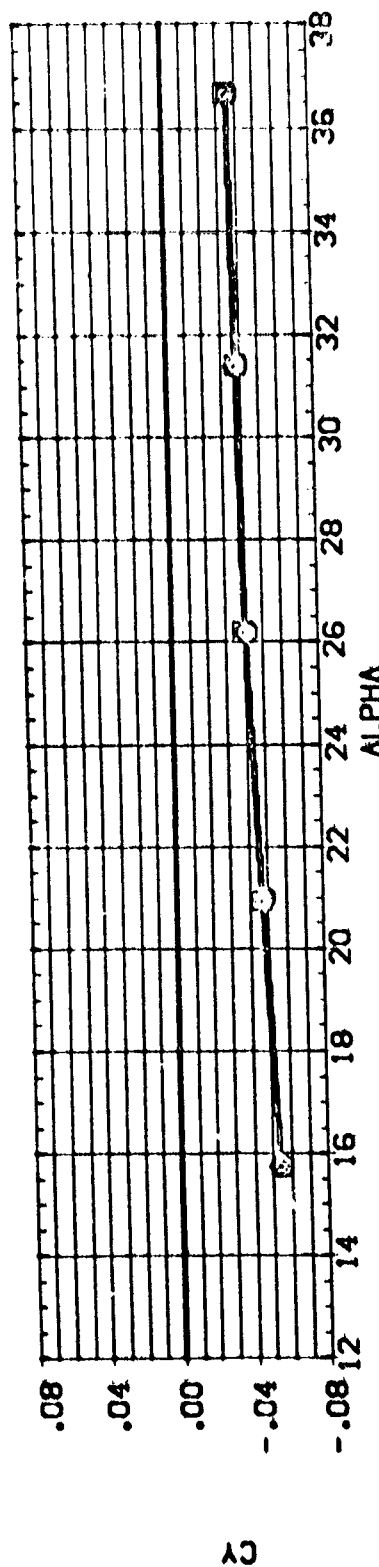


EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=4.6)



GA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV015)

SYMBOL	PG-JET	BETA	PARAMETRIC VALUES	REFERENCE INFORMATION
◇	70.268	5.000	MACH 4.600	SREF 87.1550
	70.388	1.720	ELEVTR -20.000	LREF 7.1222
	161.614	.000	BOFLAP -14.250	BREF 14.0500
		40.000	R-RODR .000	KREF 12.5800
				VREF 6.0000
				SCALE 10.150

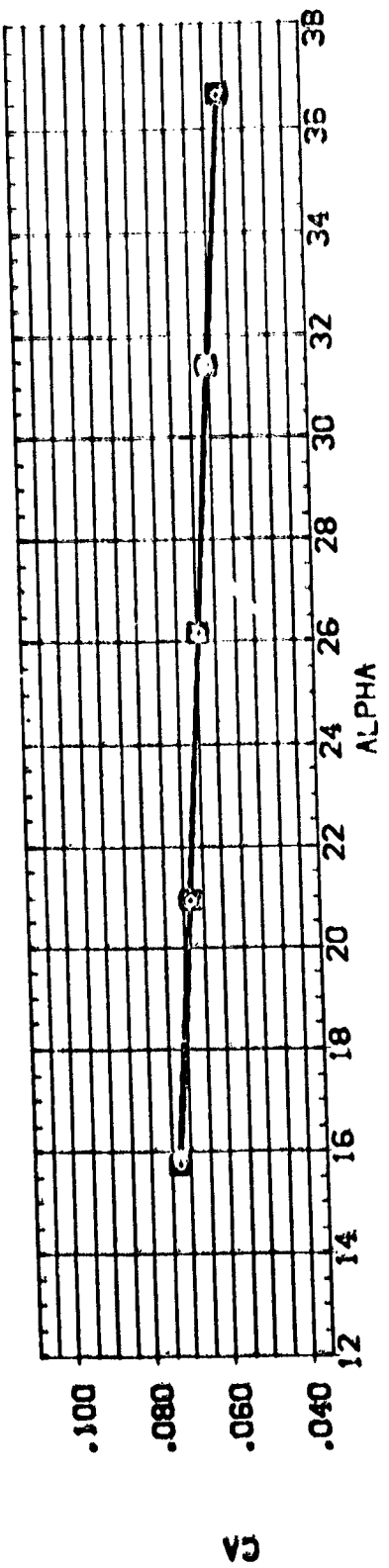
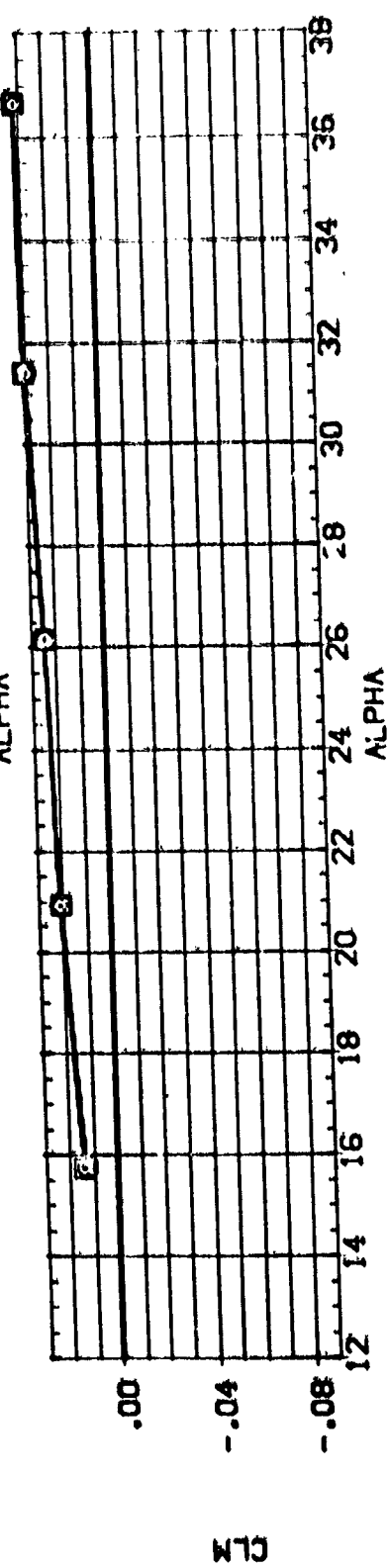
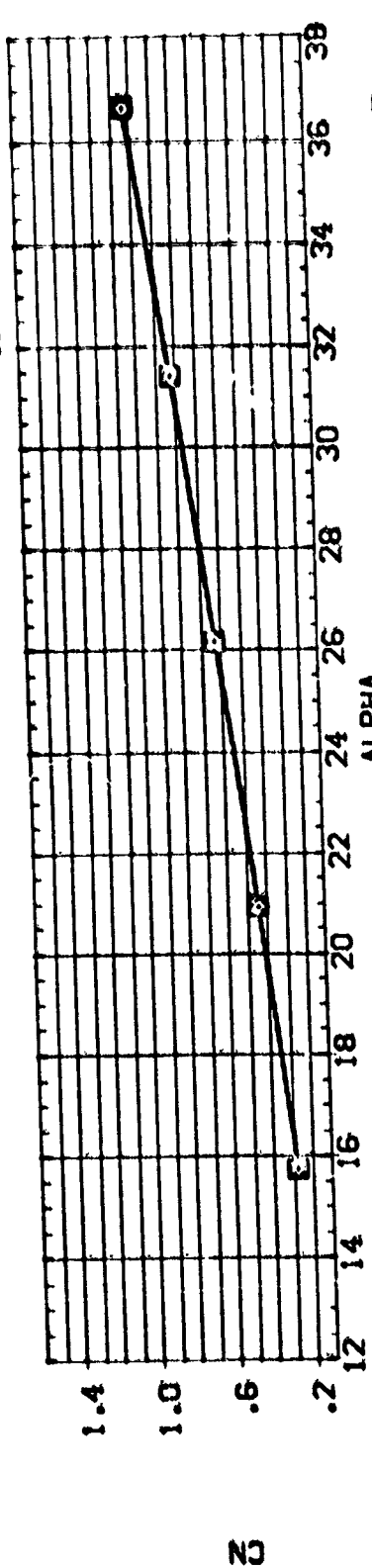


EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=4.6)



CA-70. UPWT1043.0RB(B19C7F5M6N19)(W107E23)(V7R5)(RPV018)

SYMBOL	PG-JET	BETA	PARAMETRIC VALUES	REF	REFERENCE INFORMATION
□	.326	5.000	MACH 4.600	SPREF	87.1560
◇	69.471	1.720	ELEVTR -40.000	LPREF	7.1222
	163.010	.000	BOFLAP -14.250	BPREF	14.0500
		40.000	RUDER .000	APREF	12.5800
				VPREF	.0000
				WPREF	6.0000
				SCALE	.0050



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=4.6)

GA-70. UPWT1043.CRB(B19C7F5M6N19)(W107E23)(V7R5)(RPV018)

SYMBOL

P3-JET

BETA

PARAMETRIC VALUES

5.000

MACH

4.600

ELEVTR

-40.000

BOFLAP

-14.250

RUDER

.000

□

69.471

163.010

ALDRON

P3-FLR

40.000

4.600

4.600

4.600

4.600

4.600

4.600

4.600

◇

69.471

163.010

ALDRON

P3-FLR

40.000

4.600

4.600

4.600

4.600

4.600

4.600

4.600

◇

69.471

163.010

ALDRON

P3-FLR

40.000

4.600

4.600

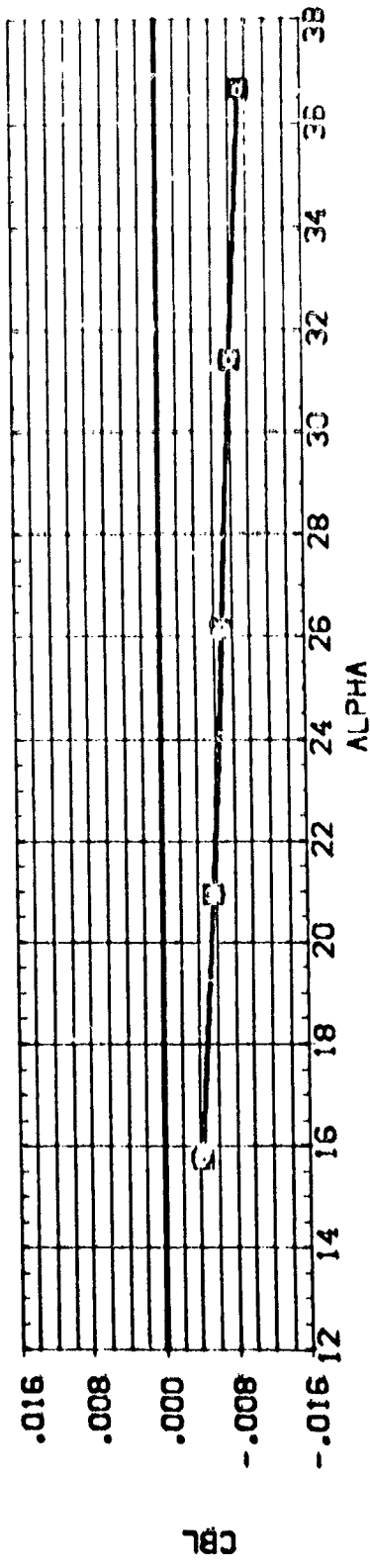
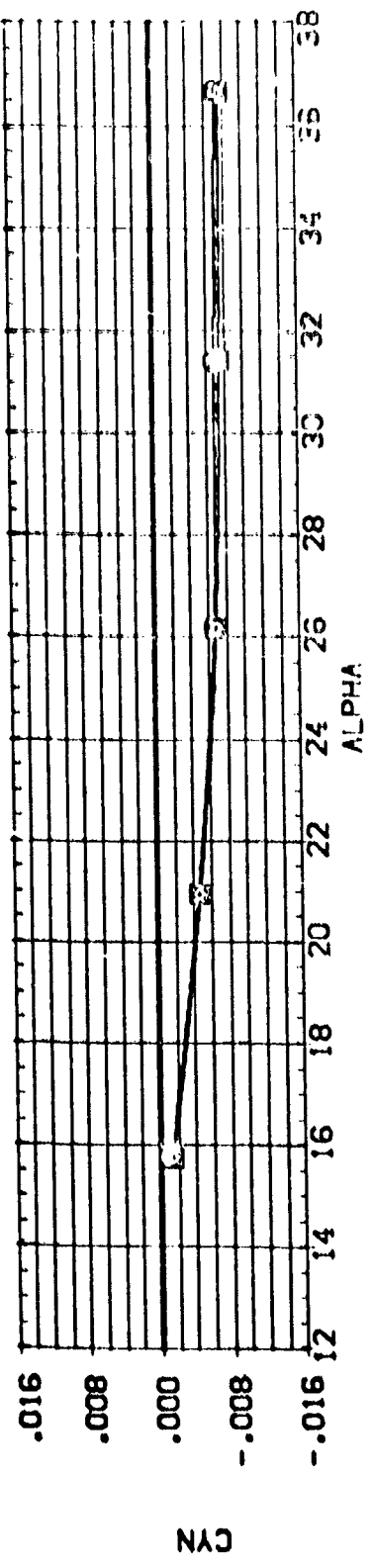
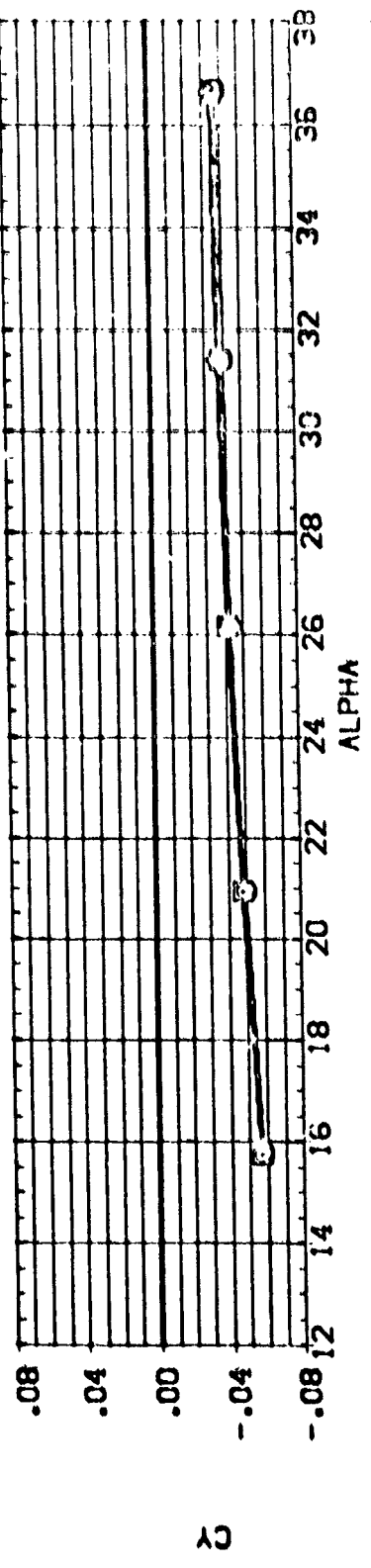
4.600

4.600

4.600

4.600

4.600



EFFECT OF RCS ON ORBITER AERO. CHARACT. (BETA=5, MACH=4.6)

APPENDIX  
TABULATED SOURCE DATA

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Tabulations of the plotted data are available  
from DMS on request.

DATE 09 OCT 75

PAGE 1

## TABULATED SOURCE DATA, LARC UPWT 1943 (0A-7D)

(APV091) ( 94 OCT 75 )

0A-7D, UPWT1043,0RB (819CTF3M6N19) (R107E23) (V7R5)

## REFERENCE DATA

SPEF = 87.1580 SA.IN. WRP = 12.5800 INCHES  
 LREF = 7.1222 INCHES WRP = .0000 INCHES  
 SPEF = 14.0500 INCHES ZRP = 6.0000 INCHES  
 SCALE = .0150

BETA = -5.000 MACH = 2.590  
 RNVL = 1.729 ELEVTR = .000  
 AIRLON = .000 ECLAP = .000  
 RUFLR = 40.000 RUCCER = .000

## PARAMETRIC DATA

RUN NO. 2/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CP17	CP19	CP20	PINF
316	15.506	373.53818	-1.6867	-1.6484	-1.7284	-1.3681	-1.5139	-1.8105	85.38016
316	20.919	373.97348	-1.7388	-1.6727	-1.7800	-1.6493	-1.5636	-1.8111	85.47965
.092	26.415	373.81500	-1.7383	-1.6465	-1.7540	-1.7797	-1.5903	-1.7085	85.39771
.316	31.794	373.48697	-1.6609	-1.6719	-1.7538	-1.7755	-1.4121	-1.8104	85.36843
.092	37.296	373.48697	-1.6609	-1.6719	-1.7285	-1.7591	-1.5138	-1.7338	85.76845
GRADIENT		-0.01085	.00024	-0.00009	.00005	-0.01163	.00028	.00028	-0.00248

RUN NO. 3/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CP17	CP19	CP20	PINF
83.398	15.503	373.81984	-1.7129	-1.6724	-1.7288	-1.3688	-1.6669	-1.8874	85.44454
62.073	20.918	373.81984	-1.7386	-1.6724	-1.7797	-1.6488	-1.6669	-1.7598	85.44454
62.326	26.313	373.81984	-1.7386	-1.6468	-1.7543	-1.7505	-1.5907	-1.7598	85.44454
62.974	31.801	374.17833	-1.6364	-1.6474	-1.7293	-1.7511	-1.5406	-1.8368	85.52647
63.421	37.285	374.02469	-1.6361	-1.6728	-1.7291	-1.7254	-1.6672	-1.7856	85.49136
GRADIENT		.01412	.00047	.00004	.00009	-0.01149	.00023	.00023	.00023

RUN NO. 4/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CP17	CP19	CP20	PINF
227.008	15.495	374.15272	-1.7134	-1.6730	-1.7293	-1.3696	-1.6928	-1.8877	85.52062
225.214	20.910	374.05039	-1.7135	-1.6728	-1.7546	-1.6491	-1.6419	-1.7547	85.49721
225.886	26.361	374.02468	-1.6875	-1.6472	-1.7291	-1.7508	-1.5911	-1.7091	85.49156
226.784	31.779	374.17833	-1.6821	-1.6474	-1.6784	-1.7002	-1.6928	-1.8623	85.52647
224.991	37.321	374.10151	-1.6820	-1.6729	-1.7292	-1.7255	-1.7689	-1.7857	85.50892
GRADIENT		.00047	.00028	.00005	.00014	-0.01140	-0.00037	.00014	.00011

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(APV012) ( 24 OCT 73 )

TABULATED SOURCE DATA, LARC UPLIFT 1043 (0A-73)

0A-73, UPLIFT 1043, ORB (B19C7T51M0N19) (R1DTE23) (VTR5)

REFERENCE DATA

SREF = 87.1505 50.14. YMRP = 12.5000 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 PSF = 14.0590 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0150

BETA = .000 MACH = 2.500  
 RN/L = 1.720 ELEVTR = .000  
 ALLRON = .000 DOFLAP = .000  
 RUFLR = 40.000 RUFLER = .000

PARAMETRIC DATA

RUN NO. 5/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBW	CF17	CF19	CF20	PINF
.092	15.327	373.99909	-.17132	-.16215	-.17293	-.12685	-.14133	-.16111	85.48591
.092	20.895	374.41152	-.17107	-.16938	-.17267	-.16719	-.15626	-.16091	85.12283
.092	26.368	373.51237	-.17639	-.16976	-.16774	-.16737	-.15901	-.16360	85.37430
.092	31.784	374.86968	-.16889	-.16742	-.17030	-.16759	-.15926	-.17359	85.68450
.092	37.286	374.89529	-.16633	-.16742	-.17304	-.17321	-.15927	-.16122	85.69733
GRADIENT		.07834	.00222	.00015	.00003	-.00176	-.00071	.00013	.01791

RUN NO. 6/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBW	CF17	CF19	CF20	PINF
69.464	15.498	374.20393	-.17392	-.16987	-.17294	-.12681	-.16166	-.16623	85.33233
51.114	20.901	374.15272	-.17134	-.16986	-.17293	-.16493	-.15913	-.16113	85.32782
60.269	26.341	374.37636	-.17393	-.16988	-.16723	-.16242	-.16677	-.16370	85.55574
61.406	31.783	374.28075	-.16879	-.16732	-.16786	-.16750	-.16423	-.17095	85.54989
65.684	37.295	374.28075	-.16623	-.16732	-.17349	-.17258	-.16169	-.16114	85.54389
GRADIENT		.00517	.00033	.00014	-.00000	-.00172	-.00009	.00037	.00110

RUN NO. 7/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBW	CF17	CF19	CF20	PINF
223.200	15.492	374.28075	-.17136	-.16988	-.17293	-.12683	-.16676	-.16624	85.54989
219.949	20.914	374.28075	-.16879	-.16732	-.17293	-.16241	-.16169	-.17625	85.54989
214.249	26.360	374.20393	-.16621	-.16475	-.15767	-.15986	-.16675	-.18368	85.53233
222.753	31.791	374.28075	-.16879	-.16732	-.17040	-.17004	-.15915	-.16841	85.54989
226.781	37.396	374.28075	-.16623	-.16732	-.17349	-.17512	-.16423	-.16114	85.54989
GRADIENT		-.00000	.00019	.00009	-.00003	-.00191	.00014	.00033	-.00000

DATE 04 OCT 75

TABULATED SOURCE DATA, LARC UPAT 1243 (04-75)

PAGE 3

04-75, UPAT1243, OFB (LSCT75MON19) (NAD1223) (NTR5)

(APV033) ( 54 OCT 75 )

## REFERENCE DATA

SREF = 07.1500 SQ. IN. YMRP = 12.5870 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.0599 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0125

## PARAMETRIC DATA

BETA = 5.000 MACH = 2.500  
 RV/L = 1.72 ELEVTR = .000  
 AIRPON = .000 EXPLAP = .000  
 RUFLER = 40.000 RUCCOR = .000

RUN NO. 6/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
092	15.488	374.28075	-1.6623	-1.6732	-1.17295	-1.15988	-1.14921	-1.17595	85.54989
092	20.911	374.43439	-1.17395	-1.16990	-1.17297	-1.17260	-1.16679	-1.18371	85.54989
-132	26.369	374.59636	-1.17650	-1.17244	-1.16932	-1.16730	-1.16677	-1.17860	85.55574
-132	31.821	374.75814	-1.17395	-1.16987	-1.17295	-1.17297	-1.16676	-1.17350	85.54403
-132	37.282	374.92079	-1.17650	-1.16732	-1.17295	-1.17298	-1.16423	-1.18624	85.54989
GRADIENT			-1.00038	-1.00000	.00000	-1.00047	-1.00036	-1.00037	-1.00035

RUN NO. 9/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
092	15.492	374.35737	-1.6624	-1.6733	-1.17042	-1.15988	-1.12366	-1.17860	85.56744
092	20.911	374.50775	-1.17395	-1.16988	-1.17040	-1.17004	-1.15408	-1.18624	85.54989
092	26.370	374.66117	-1.17138	-1.16989	-1.16279	-1.16731	-1.16678	-1.17861	85.57330
092	31.846	374.81557	-1.17137	-1.16733	-1.17296	-1.17005	-1.16424	-1.17860	85.56744
092	37.279	374.96878	-1.17651	-1.16734	-1.17042	-1.17260	-1.16425	-1.18680	85.57915
GRADIENT			-1.00033	-1.00005	-1.00005	-1.00047	-1.00019	-1.00019	-1.00035

RUN NO. 10/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
225.649	15.480	374.33196	-1.6623	-1.6477	-1.17043	-1.15988	-1.13380	-1.17808	85.56159
218.725	20.868	374.50636	-1.16623	-1.16477	-1.16278	-1.16750	-1.15409	-1.17860	85.55574
225.886	26.366	374.65737	-1.17134	-1.16733	-1.16533	-1.16731	-1.16931	-1.18371	85.56744
229.467	31.892	374.81196	-1.17394	-1.16733	-1.17296	-1.17259	-1.17185	-1.17351	85.56159
224.544	37.288	374.96196	-1.17650	-1.16989	-1.17296	-1.17259	-1.16931	-1.19134	85.56159
GRADIENT			-1.00052	-1.00023	-1.00028	-1.00036	-1.00016	-1.00047	-1.00011

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TABULATED SOURCE DATA, LARC UFM 1243 (OA-73)

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OA-73, UFW1043, OEB (B19CTF5M6N19) (N107E23) (NTR5)

(APV004) ( 34 OCT 73 )

## REFERENCE DATA

SPDF = 87.1500 50.1N. YMRP = 12.5800 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.0000 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0157

## PARAMETRIC DATA

BETA = -5.000 WACH = 2.500  
 RN/L = 1.720 ELEVTR = -27.000  
 AILRON = .000 BDFLAP = -14.231  
 RUFLR = 40.000 RUDDER = .000

RUN NO. 53/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSI)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
.272	15.491	373.43376	-1.7036	-1.6894	-1.7466	-1.1355	-1.1325	-1.1841	85.35674
.272	20.877	373.56379	-1.7295	-1.7152	-1.8233	-1.16414	-1.16090	-1.17266	85.36601
.272	26.319	373.59333	-1.7806	-1.7405	-1.8230	-1.17684	-1.16086	-1.17263	85.35333
.249	31.774	373.92227	-1.6529	-1.7558	-1.7728	-1.17183	-1.16097	-1.18546	85.46793
.273	37.259	373.69142	-1.6782	-1.7154	-1.16900	-1.16925	-1.15330	-1.17778	85.41527
GRADIENT		.01600	.00023	-.00010	.00028	-.00145	-.00000	.00004	.00066

RUN NO. 54/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSI)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
64.053	15.468	373.74303	-1.7298	-1.7155	-1.7725	-1.1363	-1.17364	-1.19310	85.42698
85.818	20.862	374.05030	-1.7302	-1.7160	-1.7984	-1.16168	-1.16861	-1.17528	85.49721
63.157	26.322	374.71675	-1.7568	-1.7425	-1.7993	-1.17449	-1.16618	-1.17283	85.64938
84.723	31.727	371.95962	-1.6754	-1.7127	-1.7731	-1.17153	-1.17382	-1.19780	85.01728
57.562	37.236	373.64050	-1.6782	-1.6897	-1.7214	-1.16924	-1.16854	-1.17778	85.40357
GRADIENT		-.04217	.00029	.00010	.00024	-.00149	.00015	.00033	-.00364

RUN NO. 55/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSI)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
226.069	15.446	373.61500	-1.7038	-1.7153	-1.7469	-1.13359	-1.17362	-1.19819	85.39771
225.845	20.891	373.53858	-1.6780	-1.7152	-1.7978	-1.16159	-1.16852	-1.17776	85.39016
225.598	26.320	373.56379	-1.7038	-1.7152	-1.7468	-1.17178	-1.16853	-1.18032	85.38801
225.175	31.772	373.76803	-1.6527	-1.7155	-1.7471	-1.17181	-1.17618	-1.18850	85.42283
224.727	37.262	373.61500	-1.6781	-1.7153	-1.7469	-1.16924	-1.18224	-1.18032	85.39771
GRADIENT		.00422	.00014	-.00005	.00009	-.00149	-.00042	.00047	.00097



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TABULATED SOURCE DATA, LARC UPWT 1243 (04-73)

(APV003) ( 04 OCT 73 )

04-73, UPWT1043,CFB(B:9C7F3MON19): (M07E23) (VTR5)

REFERENCE DATA

SREF = 87.1581 58. IN. XREF = 12.1600 INCHES  
 LREF = 7.1222 INCHES YREF = .0700 INCHES  
 BREF = 14.0500 INCHES ZREF = 6.0000 INCHES  
 SCALE = .0150

BETA = .0000 MUEN = 2.370  
 RUL = 1.720 ELEVTR = -20.000  
 AIRLON = .000 DISTLAP = -14.550  
 RUDTLR = 40.000 RUCCCP = .000

PARAMETRIC DATA

RUN NO. 567 0 RUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CP20	PINF
.277	15.469	373.87106	-.16785	-.16645	-.17473	-.13111	-.14318	-.16760	85.45624
.277	20.894	373.81384	-.17042	-.16920	-.17472	-.16164	-.15841	-.17325	85.44434
.053	26.364	373.94787	-.17301	-.17150	-.17219	-.16160	-.12797	-.18347	85.47380
.053	31.784	374.12711	-.16790	-.16905	-.17222	-.16169	-.16101	-.17274	85.51477
.053	37.235	373.89666	-.16272	-.16845	-.17218	-.16420	-.16096	-.17016	85.46209
GRADIENT		.00000	.00023	.00000	.00014	-.00122	-.00070	-.00005	.00151

RUN NO. 577 0 RUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CP20	PINF
62.711	15.482	373.15409	-.17031	-.16889	-.17207	-.13346	-.16337	-.17260	85.29238
62.712	20.983	374.05030	-.17045	-.16904	-.17475	-.16168	-.16353	-.17018	85.45721
62.488	26.393	374.10151	-.17046	-.16904	-.17221	-.15915	-.16861	-.18548	85.50892
63.383	31.743	373.51257	-.16780	-.16895	-.17212	-.16158	-.16344	-.17010	85.37430
63.159	37.243	373.69182	-.16525	-.16542	-.17211	-.16416	-.16347	-.17778	85.41527
GRADIENT		.01009	.00024	.00009	.00014	-.00113	-.00001	-.00019	.00251

RUN NO. 586 0 RUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CP20	PINF
226.743	15.458	373.89666	-.16786	-.16901	-.17218	-.13367	-.16804	-.17784	85.46209
226.742	20.905	374.30636	-.17049	-.16908	-.17224	-.16172	-.16357	-.17277	85.55574
226.294	26.336	374.10151	-.17046	-.16904	-.16967	-.15915	-.17623	-.18803	85.50892
226.071	31.770	373.46156	-.16779	-.16894	-.17467	-.16412	-.16597	-.17009	85.36260
223.622	37.221	373.41015	-.16521	-.16893	-.17211	-.16411	-.16596	-.18030	85.55089
GRADIENT		-.03342	.00015	.00001	-.00021	-.00116	-.00004	-.00004	-.00764

OA-73, UPAT1243,ORB(B19CTFSW0419) (R10TE23) (VTR3)

(APV0506) ( 24 OCT 73 )

## REFERENCE DATA

SREF = 07.1500 50. IN. XMRP = 12.5000 INCHES  
 LRFP = 7.1222 INCHES YMRP = .0000 INCHES  
 BRFP = 14.0500 INCHES ZMRP = 0.0000 INCHES  
 SCALE = .0100

## PARAMETRIC DATA

BETA = 5.000 MACH = 2.500  
 RV/L = 1.720 ELEVIR = -20.970  
 AILRON = .000 BOFLAP = -14.250  
 PUCLFL = 40.000 RUCCOR = .000

RUN NO. 59/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBN	CF17	CF19	CF20	PINF
.276	15.449	373.07106	-.17299	-.17415	-.17727	-.15656	-.15580	-.18291	85.45624
.276	23.868	373.07106	-.17542	-.17157	-.17727	-.16928	-.16350	-.18546	85.45624
.092	26.317	373.89566	-.17814	-.17413	-.17473	-.16928	-.17368	-.18851	85.46209
.092	31.772	373.89566	-.17957	-.17157	-.17982	-.16928	-.17112	-.17271	85.46209
.092	37.228	373.94787	-.18072	-.17414	-.18238	-.17436	-.17621	-.18922	85.47380
GRADIENT		.00329	-.00038	-.00000	-.00023	-.00265	-.00107	-.00055	.00079

RUN NO. 60/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBN	CF17	CF19	CF20	PINF
63.159	15.432	374.02469	-.17502	-.17415	-.17729	-.15659	-.15553	-.18293	85.49136
67.197	25.878	373.97348	-.17501	-.17158	-.17474	-.16675	-.15336	-.19312	85.47965
61.368	26.349	373.67106	-.17299	-.17413	-.16965	-.16674	-.17112	-.18291	85.45624
60.697	31.778	373.97348	-.17558	-.17158	-.17729	-.16675	-.16859	-.17017	85.47965
62.711	37.222	373.97348	-.18072	-.17414	-.17983	-.17184	-.17113	-.18547	85.47965
GRADIENT		-.00189	-.00033	-.00000	-.00034	-.00036	-.00177	-.00033	-.00043

RUN NO. 61/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBN	CF17	CF19	CF20	PINF
228.534	15.454	374.05030	-.16788	-.17160	-.17221	-.15405	-.15561	-.18903	85.49721
227.862	25.895	374.07593	-.16789	-.16648	-.16457	-.16168	-.15846	-.17274	85.50506
226.296	26.309	374.10151	-.17560	-.17160	-.17221	-.16677	-.17623	-.18803	85.50892
224.282	31.803	374.12711	-.17817	-.17417	-.18494	-.17186	-.17369	-.17529	85.51477
226.072	37.214	374.25514	-.17819	-.17418	-.17987	-.17442	-.17371	-.18550	85.54403
GRADIENT		.07847	-.00057	-.00024	-.00056	-.00094	-.00168	-.00055	.00194

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TABULATED SOURCE DATA, LABC UPWT 1943 (CA-70)

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(APR002) (24 OCT 73)

CA-70, UPL11043, ORB (B19C773MON19) (R407223) (VTR5)

## REFERENCE DATA

SRF = 87.1968 SQ. IN. YMRP = 12.0000 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 DRFP = 14.0000 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = -1.0000 MACH = 2.5000  
 RNUL = 1.7200 ELEVTR = -471.0000  
 AIRLON = .0000 BDTAP = -31.2500  
 RUOFLR = 41.0000 FUTTER = .0000

RUN NO. 22/ C RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
1.009	15.432	373.94787	-.16125	-.16234	-.16796	-.12198	-.07293	-.17871	85.47380
.784	20.872	373.51297	-.16889	-.16995	-.17806	-.11987	-.09031	-.18389	85.37430
.561	26.362	373.53816	-.17147	-.16995	-.18084	-.11773	-.10831	-.17886	85.39016
.783	31.733	373.66621	-.17663	-.17766	-.18320	-.11772	-.12106	-.18633	85.42942
.561	37.234	373.79424	-.16894	-.17255	-.18267	-.11519	-.10583	-.18534	85.43868
GRADIENT		-.03281	-.00042	-.00152	-.00056	-.01229	-.00177	-.00009	-.00064

RUN NO. 23/ C RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
67.695	15.477	373.99909	-.16389	-.16491	-.16797	-.12161	-.12623	-.18127	85.48331
63.891	20.871	374.12711	-.17156	-.17204	-.17817	-.11999	-.15672	-.17109	85.51477
63.891	26.335	374.15272	-.17156	-.17305	-.18072	-.17525	-.12373	-.18129	85.52062
63.867	31.738	374.22954	-.17414	-.17773	-.18073	-.17526	-.16942	-.18640	85.53616
63.891	37.264	374.25314	-.16911	-.17362	-.17816	-.17526	-.15674	-.16856	85.54403
GRADIENT		.01127	-.00024	-.00142	-.00042	-.01224	-.00135	-.00019	.00250

RUN NO. 24/ C RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
225.683	15.481	374.58802	-.16659	-.16756	-.16806	-.12198	-.14413	-.18990	85.62012
225.683	20.857	374.17833	-.16903	-.17005	-.17363	-.16000	-.17195	-.17875	85.52647
227.230	26.296	372.74440	-.16877	-.17240	-.17543	-.17504	-.16664	-.17344	85.19872
226.832	31.732	373.43376	-.17680	-.18119	-.18062	-.17514	-.17184	-.18996	85.35874
224.116	37.252	373.69182	-.16892	-.17310	-.18066	-.17518	-.17696	-.17359	85.41527
GRADIENT		-.04639	-.00023	-.00146	-.00055	-.00223	-.00120	-.00038	-.01060

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TABULATED SOURCE DATA, LAPC UPUT 1943 (0A-73)

(APV000) ( 94 OCT 73 )

0A-73, UP-T1943, OFB (B19C7F5JMN19) (NDJTEZ3) (V7H5)

## REFERENCE DATA

SRP = 0".160 56-IN. YMRP = 12.3693 INCHES  
 LPEP = 7.1222 INCHES YMRP = .0000 INCHES  
 BRP = 14.2500 INCHES YMRP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = .000 MACH = 2.500  
 RNVL = 1.720 ELEVTR = -40.000  
 AIRLON = .000 E5FLAP = -14.250  
 RUFTLR = 40.000 RUCCCP = .000

RUN NO. 28/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
337	15.484	374.02469	-16127	-16491	-16797	-12945	-10500	-16893	85.49136
336	20.821	374.02469	-17154	-17023	-17306	-16505	-14401	-17363	85.49136
336	26.297	373.97348	-17668	-17514	-17505	-16505	-14053	-18127	85.47965
336	31.768	374.02469	-17668	-17515	-18325	-17268	-15416	-17872	85.49136
336	37.212	373.99909	-16897	-17258	-18070	-17014	-15669	-17872	85.49136
GRADIENT		-10094	-10038	-10038	-10066	-10163	-10288	-10047	-10021

RUN NO. 29/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
64.337	15.441	373.76863	-16379	-16487	-16793	-12938	-13887	-16593	85.43283
62.771	20.892	374.02469	-17154	-17003	-17306	-16251	-16177	-17618	85.49136
63.442	26.277	373.76863	-17408	-17511	-17048	-16247	-17189	-18379	85.43283
63.690	31.787	373.56379	-17682	-17508	-18374	-17516	-14390	-16845	85.38601
64.114	37.198	373.71742	-16893	-17254	-17811	-17309	-15664	-18124	85.42112
GRADIENT		-10106	-10028	-10037	-10061	-10175	-10032	-10042	-10027

RUN NO. 30/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
228.593	15.489	373.81984	-16380	-16468	-16794	-12938	-15920	-16849	85.44454
227.874	20.841	373.81984	-16637	-16744	-16794	-15993	-16428	-17615	85.44454
225.012	26.368	373.97348	-17668	-17514	-17505	-16505	-17446	-18382	85.47965
225.460	31.749	373.84545	-17668	-17768	-18322	-17775	-16428	-16085	85.47965
227.697	37.177	373.87106	-16895	-17050	-17813	-17012	-16175	-18126	85.45624
GRADIENT		-10028	-10038	-10038	-10066	-10183	-10010	-10019	-10054

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TABULATED SOURCE DATA, LARC UPWT 1943 (0A-73)

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0A-73, UPWT1943,08B (B15C7F5M0N19) (M07E23) (VTR:)

(APW029) 1 04 OCT 73

REFERENCE DATA

SREF = 87.1703 SQ. IN. YREF = 12.1000 INCHES  
LREF = 7.1222 INCHES YREF = .0700 INCHES  
BREF = 14.0500 INCHES ZREF = 6.0000 INCHES  
SCALE = .0100

BETA = 5.0000 RADN = 2.9900  
RUL = 1.720 ELEVIR = -40.0000  
AIRLON = .0000 BOFLAP = -14.830  
RUCCLR = 40.0000 RUCCER = .0000

PARAMETRIC DATA

RUN NO. 23/ 0 RUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(FSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
.562	15.461	373.71742	-.17150	-.16742	-.17556	-.15737	-.12613	-.17100	85.42112
.336	25.849	373.79424	-.17408	-.16999	-.17558	-.17310	-.14395	-.17668	85.43868
.336	26.283	373.76863	-.17665	-.17767	-.17812	-.16756	-.15919	-.17359	85.43263
.115	31.732	373.61984	-.17922	-.17768	-.16832	-.16502	-.15412	-.16633	85.44454
.114	37.268	373.76863	-.17408	-.17311	-.16831	-.16501	-.16681	-.17359	85.43263
GRADIENT	.00233		-.00019	-.00042	-.00070	-.00019	-.00169	-.00023	.00053

RUN NO. 26/ 0 RUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(FSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
63.892	15.423	373.61984	-.17151	-.16744	-.17303	-.15738	-.11856	-.16849	85.44454
63.893	25.890	373.69666	-.17152	-.17001	-.17350	-.16903	-.13636	-.16636	85.46259
63.867	26.292	373.71742	-.17407	-.17310	-.17811	-.16755	-.16426	-.17668	85.42112
63.221	31.763	373.79424	-.17665	-.17768	-.16831	-.16247	-.17189	-.16380	85.43868
63.892	37.268	373.76863	-.17665	-.17311	-.16831	-.16501	-.16681	-.17359	85.43263
GRADIENT	-.00374		-.00028	-.00042	-.00089	-.00023	-.00242	-.00014	-.00086

RUN NO. 27/ 0 RUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(FSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
227.027	15.458	373.97348	-.16897	-.16490	-.17305	-.15741	-.12622	-.15342	85.47963
224.769	25.853	373.92227	-.17153	-.17001	-.16541	-.16249	-.15922	-.14361	85.46795
226.579	26.513	373.92227	-.17667	-.17313	-.16069	-.16758	-.17191	-.16381	85.46795
226.579	31.730	373.99909	-.18182	-.18026	-.16833	-.16505	-.17954	-.17617	85.48551
223.237	37.215	374.05030	-.17669	-.17771	-.16974	-.16252	-.17193	-.17616	85.49721
GRADIENT	.00424		-.00047	-.00066	-.00098	-.00023	-.00215	-.00033	.00037

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## TABULATED SOURCE DATA, LARC UPMT 1043 (0A-73)

(APPROX) ( 24 OCT 73 )

0A-73, UPMT 1043, OBS (DISCRETEMENTS) (OUTFEEDS) (VTRIS)

## REFERENCE DATA

SPOT = 87.1500 SQ. IN. YMRP = 12.5000 INCHES  
 LRP = 7.1222 INCHES YMRP = .0000 INCHES  
 BRP = 14.3500 INCHES YMRP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = -5.000  
 RVUL = 1.720  
 AILSON = .000  
 RUSFLR = 40.000  
 PIVM = 4.800  
 CLEVER = .000  
 EOTLAP = .000  
 RUCCER = .000

RUN NO. 11/ 0 RVUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PIVF
75.359	15.839	202.48810	-.02870	-.02097	-.04984	-.03498	-.01472	-.06932	13.67524
69.463	20.985	202.47906	-.03344	-.02096	-.04983	-.03497	-.01470	-.06511	13.66993
73.358	26.212	202.46550	-.03344	-.02097	-.04984	-.03497	-.01940	-.06932	13.66992
75.582	31.486	202.44741	-.02869	-.02096	-.04984	-.03497	-.01940	-.06298	13.66780
	36.708	202.49282	-.03345	-.02097	-.04984	-.03498	-.02410	-.06398	13.67385
GRADIENT		.00070	-.00009	-.00018	-.00027	-.00001	-.00045	-.00099	.00005

RUN NO. 12/ 0 RVUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PIVF
75.359	15.827	202.51975	-.03345	-.02571	-.04984	-.03498	-.02979	-.06511	13.67268
69.463	26.188	202.47906	-.03345	-.02570	-.04984	-.03498	-.03347	-.06511	13.66993
73.358	31.412	202.47454	-.03345	-.02570	-.04984	-.03498	-.03347	-.06511	13.66993
75.582	36.729	202.46550	-.03344	-.02570	-.04984	-.03497	-.03347	-.06511	13.66992
GRADIENT		-.00259	.00000	.00000	-.00026	.00000	-.00223	.00000	-.00017

RUN NO. 13/ 0 RVUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PIVF
162.356	15.738	202.50619	-.03345	-.02570	-.04984	-.03498	-.03348	-.06511	13.67177
163.899	20.951	202.48810	-.03345	-.02570	-.04984	-.03498	-.03347	-.06511	13.67584
163.676	26.206	202.51975	-.03345	-.02571	-.04984	-.03498	-.03347	-.06511	13.67268
163.229	31.456	202.51071	-.03345	-.02571	-.04984	-.03498	-.03347	-.06511	13.67267
162.781	36.694	202.51975	-.03345	-.02571	-.04984	-.03498	-.03347	-.06511	13.67268
GRADIENT		.00095	.00000	-.00000	.00018	-.00000	-.00027	-.00000	.00006

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TABULATED SOURCE DATA, LARC UPAD 1243 (04-73)

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(APR 23) (24 OCT 73)

04-73, UPAD1543,CFB (B13C77)M619; (N2J7E23) (V7R3)

# REFERENCE DATA

SPOT = 87.1565 32.1N. 106P = 12.5600 INCHES  
 LPT = 7.1222 INCHES 106P = .0000 INCHES  
 SPOT = 14.0000 INCHES 106P = 6.0000 INCHES  
 SCALE = .0150

# PARAMETRIC DATA

BETA = .000 RADN = 4.670  
 RN/L = 1.72 ELEV = .000  
 AIR/OM = .000 ELEV = .000  
 PLE/LR = 43.000 ELEV = .000

RUN NO. 14/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
.317	15.745	202.54236	-.03820	-.02571	-.04984	-.03968	-.03248	-.05169	13.67421
.317	20.961	202.52880	-.03820	-.02571	-.04984	-.03968	-.03348	-.06311	13.67529
.094	26.215	202.46810	-.03345	-.02571	-.04984	-.03968	-.03347	-.06311	12.67524
.094	31.461	202.51592	-.03820	-.02572	-.04984	-.03969	-.03348	-.06311	13.67512
.317	36.743	202.49715	-.03345	-.02570	-.04984	-.03968	-.03347	-.06311	13.67515
GRADIENT		-.00121	.00000	.00000	.00000	.00000	.00000	.00000	-.00000

RUN NO. 15/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
70.137	15.737	202.51071	-.03345	-.02571	-.04984	-.03968	-.03817	-.06311	13.67527
71.480	20.964	202.46810	-.03345	-.02570	-.04984	-.03968	-.03816	-.06311	13.67524
72.598	26.186	202.48358	-.03345	-.02570	-.04984	-.03968	-.03816	-.06311	13.67524
64.766	31.447	202.48810	-.03820	-.02570	-.04984	-.03968	-.03816	-.06311	13.67524
65.214	36.707	202.54236	-.03820	-.02571	-.04984	-.03968	-.03817	-.06311	13.67421
GRADIENT		.00121	-.00000	-.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 16/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
181.887	15.736	202.54658	-.03346	-.02571	-.04984	-.03968	-.03817	-.06311	13.67451
183.453	20.961	202.40220	-.03343	-.02568	-.04983	-.03966	-.04284	-.06311	13.66474
184.572	26.190	202.56245	-.03346	-.02572	-.04984	-.03969	-.04286	-.06311	13.67543
184.572	31.425	202.51071	-.03820	-.02571	-.04984	-.03968	-.04285	-.06311	13.67527
184.796	36.744	202.56497	-.03346	-.02569	-.04984	-.03969	-.03817	-.06311	13.67523
GRADIENT		.00127	-.00000	.00000	.00000	.00000	.00000	.00000	.00000

RUN NO. 17/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
397.752	15.765	202.53332	-.03346	-.02571	-.04984	-.03968	-.03817	-.06311	13.67360
413.416	20.961	202.46550	-.03344	-.02570	-.04984	-.03967	-.03816	-.06311	13.66972
420.577	26.199	202.53332	-.03346	-.02571	-.04984	-.03968	-.04286	-.06311	13.67360
440.046	31.398	202.54236	-.03346	-.02571	-.04984	-.03968	-.03817	-.06311	13.67421
446.983	36.750	202.52427	-.03820	-.02571	-.04984	-.03968	-.04286	-.06311	13.67360
GRADIENT		.00112	-.00000	-.00000	.00000	.00000	.00000	.00000	.00000

REFERENCE DATA

SPDF = 87.1583 58. IN.

LRPF = 7.1222 INCHES

DRPF = 14.0370 INCHES

SCALE = .0150

YRPF = 12.5800 INCHES

ZRPF = .0000 INCHES

ZRPF = 6.0000 INCHES

PARAMETRIC DATA

BETA = 9.000

FMOM = 4.670

FMVL = 1.720

ELVTR = .000

ATURON = .000

SLAP = .000

FUDTLP = 40.000

FLCTIO = .000

RUN NO. 18/ 0

FMVL = 1.72

GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSP)	CFB	CFSC	CFEM	CF17	CF19	CF20	FMF
.317	15.774	202.3280	-.03345	-.02371	-.04984	-.03968	-.03817	-.06511	13.67329
.317	20.948	202.47404	-.03819	-.02370	-.04984	-.03967	-.03816	-.06511	13.66323
.317	26.229	202.51323	-.03820	-.02371	-.04984	-.03968	-.04285	-.06511	13.67029
.094	31.455	202.52619	-.03820	-.02370	-.05454	-.03968	-.03816	-.06511	13.67172
.094	36.805	202.51323	-.03345	-.02371	-.04984	-.03968	-.03817	-.06511	13.67229
GRADIENT	.0000	.0000	.0000	-.00001	-.00009	-.00000	.00000	.00000	.00000

RUN NO. 19/ 0

FMVL = 1.72

GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSP)	CFB	CFSC	CFEM	CF17	CF19	CF20	FMF
58.277	15.724	202.54206	-.03820	-.02371	-.04984	-.03968	-.03817	-.06511	13.67421
65.515	20.953	202.51323	-.03820	-.02371	-.05454	-.03968	-.03817	-.06511	13.67229
63.424	26.183	202.37035	-.03818	-.02368	-.05453	-.03966	-.04284	-.06511	13.66361
66.333	31.442	202.56949	-.03821	-.02372	-.05453	-.03969	-.03817	-.06511	13.67674
67.900	36.888	202.56045	-.03820	-.02372	-.04984	-.03969	-.03817	-.06511	13.67343
GRADIENT	.00174	.00000	-.00000	-.00000	.00000	-.00000	.00000	.00000	.00012

RUN NO. 20/ 0

FMVL = 1.72

GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSP)	CFB	CFSC	CFEM	CF17	CF19	CF20	FMF
163.231	15.734	202.50167	-.03820	-.02373	-.05454	-.03968	-.04285	-.06511	13.67146
162.113	20.980	202.53332	-.03820	-.02374	-.04984	-.03968	-.04286	-.06511	13.67350
164.390	26.183	202.49715	-.03820	-.02370	-.04984	-.03969	-.04754	-.06511	13.67116
166.588	31.428	202.47906	-.03819	-.02370	-.05924	-.03967	-.04285	-.06511	13.66323
167.483	36.864	202.49715	-.03820	-.02370	-.05454	-.03968	-.04285	-.06511	13.67116
GRADIENT	-.00121	.00000	.00000	.00027	-.00018	.00000	.00000	-.00000	-.00008



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TABULATED SOURCE DATA, LARC UPRR 1943 (04-75)

(APV013) ( 24 OCT 75 )

04-75, UPRR1045, ORB (B19C771M0N19) (N157E23: N77E5)

## REFERENCE DATA

SPD = 67.1500 50 IN. YMEP = 12.5000 INCHES  
 LPEY = 7.1222 INCHES YMEP = .0000 INCHES  
 BPEY = 14.0000 INCHES ZMEP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

DATA = -5.0000 NALON = 4.0000  
 ENUL = 1.7200 ELEV8 = -20.0000  
 ALLCON = .0000 POFPLAP = -14.0000  
 POFPLR = 40.0000 FUELER = .0000

RUN NO. 44/ 0 ENUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
69.414	15.792	202.49262	-.04083	-.04263	-.03744	-.03799	-.03176	-.07268	13.67085
71.652	20.964	202.43193	-.04083	-.04263	-.03744	-.03799	-.03176	-.07268	13.66810
72.995	26.212	202.47002	-.04083	-.04263	-.03744	-.03799	-.03176	-.07268	13.66810
69.862	31.469	202.46550	-.04083	-.04263	-.03744	-.03799	-.03176	-.07268	13.67146
69.969	36.702	202.47002	-.04083	-.04263	-.03744	-.03799	-.03176	-.07268	13.66780
GRADIENT		.00000	-.00000	.00000	-.00000	-.00000	-.00000	.00000	-.00000

RUN NO. 45/ 0 ENUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
69.414	15.743	202.46550	-.04083	-.04263	-.03744	-.03799	-.04114	-.07268	13.66302
71.652	20.957	202.46097	-.04557	-.04263	-.06214	-.03799	-.04114	-.07268	13.66871
72.995	26.212	202.47002	-.04083	-.04736	-.03744	-.04260	-.04114	-.07268	13.66302
69.862	31.469	202.46550	-.04083	-.04736	-.03744	-.04260	-.04114	-.07268	13.66932
69.969	36.702	202.47002	-.04083	-.04736	-.03744	-.04260	-.04114	-.07268	13.66932
GRADIENT		.00000	-.00000	-.00000	.00000	-.00000	-.00000	.00000	-.00000

RUN NO. 46/ 0 ENUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
163.401	15.730	202.49262	-.04559	-.04263	-.06214	-.04260	-.04114	-.07268	13.67085
164.968	20.957	202.48358	-.04083	-.04736	-.06214	-.04260	-.04114	-.07268	13.67024
164.744	26.178	202.50167	-.04083	-.04736	-.03744	-.04260	-.04114	-.07268	13.67146
164.072	31.455	202.51071	-.04083	-.04736	-.03274	-.04260	-.04114	-.07268	13.67024
162.954	36.702	202.50167	-.04083	-.04263	-.03274	-.04260	-.04114	-.07268	13.67146
GRADIENT		.00000	-.00018	.00000	.00000	-.00000	-.00000	.00000	-.00000

04-73, UPM11043, ORB (B19C77516H191 (AC07E23) (V7R3)

(APV014) ( 24 OCT 73 )

## REFERENCE DATA

CREF = 67.1501 50.14. XMRP = 12.5803 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.0970 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0159

## PARAMETRIC DATA

BETA = .0000 WACH = 4.690  
 RNVL = 1.720 ELEVTR = -20.990  
 ATTURON = .0000 BOFLAP = -14.250  
 RUFLR = 40.000 RUDEER = .0000

RUN NO. 47/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
.042	15.760	202.49715	-.04558	-.04736	-.06214	-.04260	-.04114	-.07268	13.67115
.042	20.944	202.50619	-.04558	-.04736	-.05744	-.04260	-.04114	-.07268	13.67177
.042	26.188	202.50167	-.04558	-.04736	-.05744	-.04260	-.04114	-.07268	13.67146
.042	31.423	202.51995	-.04084	-.04264	-.05274	-.04261	-.04115	-.07268	13.67268
.042	36.679	202.51073	-.04083	-.04263	-.05744	-.04260	-.04583	-.07268	13.67207
GRADIENT		.00078	.00027	.00027	.00027	.00000	-.00016	.00000	.00000

RUN NO. 48/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
66.281	15.759	202.51523	-.04083	-.04263	-.05744	-.04260	-.04583	-.07268	13.67230
69.661	20.954	202.53784	-.04084	-.04264	-.05744	-.04261	-.04584	-.07267	13.67390
66.504	26.199	202.53784	-.04558	-.04264	-.05744	-.04261	-.04584	-.07267	13.67390
68.969	31.445	202.50167	-.04558	-.04263	-.05744	-.04260	-.04114	-.07268	13.67146
70.532	36.690	202.50167	-.04083	-.04263	-.05744	-.04260	-.04583	-.07268	13.67146
GRADIENT		-.00121	-.00009	.00000	.00000	.00000	.00009	-.00000	-.00000

RUN NO. 49/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
163.631	15.759	202.52427	-.04558	-.04264	-.05744	-.04261	-.04583	-.07268	13.67299
164.746	20.984	202.52427	-.04084	-.04736	-.05744	-.04261	-.04583	-.07268	13.67299
164.299	26.166	202.50167	-.04558	-.04736	-.05744	-.04260	-.04583	-.07268	13.67146
164.075	31.420	202.54688	-.04084	-.04264	-.05744	-.04261	-.04584	-.07267	13.67451
163.528	36.690	202.49262	-.04083	-.04263	-.05744	-.04260	-.04583	-.07268	13.67395
GRADIENT		-.00078	.00018	.00009	-.00000	.00000	-.00000	.00000	-.00005

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TABULATED SOURCE DATA, LARC UPAT 1943 (04-73)

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04-73, UPAT:043,CFB(819C7F5MON33) (N07FE23) (VTR3)

(APV033) (24 OCT 73)

## REFERENCE DATA

SPET = 07.1500 50. IN. YMRP = 12.5020 INCHES  
 LPEP = 7.1222 INCHES YMRP = .0000 INCHES  
 SPEP = 14.0590 INCHES ZMRP = 6.0020 INCHES  
 SCALE = .0120

## PARAMETRIC DATA

BETA = 3.000 WACH = 4.000  
 RNUL = 1.720 ELEVTR = -20.000  
 AILPON = .000 BDFLAP = -14.250  
 RUDEPLR = 40.000 RUDECR = .000

RUN NO. 51/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
.268	15.769	202.51523	-.04303	-.04268	-.05744	-.04268	-.04115	-.07268	13.67236
.044	20.983	202.50619	-.04558	-.04268	-.05744	-.04268	-.04114	-.07268	13.67177
.044	26.182	202.53332	-.04558	-.04268	-.06214	-.04261	-.04115	-.07267	13.67363
.044	31.459	202.51523	-.04558	-.04268	-.05744	-.04268	-.04115	-.07268	13.67236
.044	36.688	202.51975	-.04558	-.04268	-.05274	-.04261	-.04115	-.07268	13.67268
GRADIENT		.00339	-.00318	-.00000	.00018	-.00000	-.00000	-.00000	.00072

RUN NO. 51/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
73.288	15.744	202.53784	-.04558	-.04264	-.06214	-.04261	-.04584	-.07267	13.67393
72.192	20.948	202.52427	-.04558	-.04264	-.06214	-.04261	-.04583	-.07268	13.67299
66.732	26.204	202.52427	-.05033	-.04264	-.06214	-.04261	-.04588	-.07268	13.67299
66.969	31.497	202.56497	-.04559	-.04264	-.06214	-.04261	-.04584	-.07267	13.67373
70.983	36.681	202.56949	-.04559	-.04264	-.06214	-.04261	-.04584	-.07267	13.67604
GRADIENT		.00199	-.00000	-.00000	-.00000	-.00000	-.00000	-.00000	.00015

RUN NO. 52/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q(PSE)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
181.814	15.736	202.52427	-.04558	-.04736	-.05744	-.04261	-.04583	-.07268	13.67299
184.076	20.948	202.53784	-.04558	-.04264	-.06214	-.04261	-.05032	-.07267	13.67393
182.286	26.182	202.53784	-.04558	-.04264	-.05744	-.04261	-.05032	-.07267	13.67393
189.496	31.491	202.55140	-.04558	-.04264	-.05744	-.04261	-.04584	-.07267	13.67482
182.286	36.669	202.58757	-.04559	-.04264	-.06214	-.04261	-.04584	-.07267	13.67728
GRADIENT		.00268	-.00000	.00018	-.00009	-.00000	-.00009	.00000	.00019

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TABULATED SOURCE DATA, LARC UPWT 1043 (0A-73)

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0A-73, UPWT1043, OFB(819K7F346819) (N437E23) (V783)

(AP-018) (24 OCT 73)

## REFERENCE DATA

SPF = 87.1501 SQ. IN. XMRP = 12.5800 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 DREF = 14.0970 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0125

## PARAMETRIC DATA

BETA = -5.000 MACH = 4.690  
 RN/L = 1.720 ELEVTR = -40.000  
 AIRLSEN = .000 EXFLAP = -14.250  
 RUFLER = 40.000 FUDDER = .000

RUN NO. 31/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
.112	15.766	202.45193	-.03610	-.03901	-.05432	-.03474	-.00976	-.06020	13.66810
.112	20.961	202.46350	-.03611	-.03901	-.05432	-.03944	-.02394	-.06492	13.66972
.112	26.207	202.42228	-.03610	-.03901	-.04962	-.03944	-.02383	-.06491	13.66597
.112	34.430	202.42480	-.03335	-.03974	-.05432	-.03944	-.02383	-.06491	13.66627
.112	36.675	202.47002	-.03336	-.03974	-.05433	-.03945	-.02383	-.06668	13.66932
GRADIENT		-.00009	.00027	-.00027	-.00000	-.00016	-.00072	.00090	-.00001

RUN NO. 32/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
71.943	15.724	202.37051	-.03809	-.03973	-.05432	-.03943	-.02851	-.06491	13.66261
70.376	20.992	202.39767	-.03810	-.03973	-.05432	-.03944	-.03789	-.06491	13.66444
72.614	26.182	202.37959	-.04284	-.03973	-.04961	-.03943	-.03789	-.06491	13.66322
72.166	34.392	202.37959	-.03809	-.03973	-.04961	-.03943	-.03789	-.04607	13.66322
71.495	36.674	202.42228	-.03810	-.03974	-.04962	-.03944	-.03321	-.04607	13.66597
GRADIENT		.00156	-.00000	-.00000	.00027	-.00000	-.00016	.00128	.00011

RUN NO. 33/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	PINF
163.035	15.756	202.39767	-.04284	-.03973	-.05432	-.03944	-.03789	-.06020	13.66444
164.987	20.931	202.37055	-.04284	-.03973	-.05432	-.03943	-.03789	-.06491	13.66261
164.363	26.172	202.40220	-.04285	-.03973	-.05432	-.03944	-.03789	-.06491	13.66474
163.915	31.426	202.37507	-.03809	-.03973	-.05432	-.03943	-.03789	-.05078	13.66291
163.466	36.674	202.39315	-.03810	-.03973	-.04962	-.03943	-.03789	-.05685	13.66415
GRADIENT		-.00009	.00027	.00000	.00018	.00000	-.00000	.00117	-.00001

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TABULATED SOURCE DATA, LARC UPWT 1043 (04-73)

(JPMV17) ( 04 OCT 73 )

04-73, UPWT1043,0681819C7F5M0W191 (0307223) (VTR3)

PARAMETRIC DATA

BETA = .000 MAGN = 4.091  
 RVAL = 1.720 ELEVAT = -40.000  
 ALTUDN = .000 ESTAP = -14.230  
 RUFLR = 40.000 RUFLER = .000

REFERENCE DATA

SPT = 87.1580 SB-IN. XMRP = 12.1600 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.0000 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0100

RUN NO. 34/ 0 RVAL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PST)	CPB	CFSC	CFBM	CF17	CF19	CP20	PINF
-117	19.751	232.38411	-.04284	-.03973	-.05432	-.03943	-.03323	-.06491	13.66352
.107	20.959	232.44289	-.03810	-.03974	-.05432	-.03944	-.03790	-.06491	13.66369
.106	26.196	232.37059	-.03809	-.03973	-.05432	-.03943	-.03789	-.06491	13.66281
.106	31.441	232.38883	-.03810	-.03973	-.05432	-.03943	-.03789	-.05078	13.66383
-.116	36.677	232.42932	-.03810	-.03974	-.05432	-.03944	-.03790	-.05078	13.66356
GRADIENT		.00000	.00000	-.00000	.00000	-.00000	-.00000	.00000	.00000

RUN NO. 35/ 0 RVAL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PST)	CPB	CFSC	CFBM	CF17	CF19	CP20	PINF
71.938	19.723	232.46097	-.03811	-.03974	-.05432	-.03944	-.03790	-.06491	13.66871
79.371	20.959	232.41576	-.03810	-.03973	-.05432	-.03944	-.03790	-.06491	13.66566
69.926	26.181	232.41124	-.03810	-.03973	-.05432	-.03944	-.03790	-.05049	13.66535
69.924	31.393	232.42028	-.03810	-.03974	-.05432	-.03944	-.03790	-.05049	13.66597
69.924	36.659	232.45645	-.03811	-.03974	-.05432	-.03944	-.03790	-.06020	13.66841
GRADIENT		-.00000	-.00000	-.00000	.00000	-.00000	.00000	.00000	-.00001

RUN NO. 36/ 0 RVAL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PST)	CPB	CFSC	CFBM	CF17	CF19	CP20	PINF
163.029	15.751	232.39315	-.04284	-.03973	-.05432	-.03943	-.04728	-.06020	13.66413
164.356	20.930	232.42932	-.04289	-.03974	-.05432	-.03944	-.04728	-.06491	13.66558
163.696	26.163	232.40220	-.04285	-.03973	-.05432	-.03944	-.04728	-.06491	13.66474
163.462	31.428	232.44289	-.04285	-.03974	-.05432	-.03944	-.04259	-.06007	13.66749
163.015	36.674	232.43837	-.03810	-.03971	-.05432	-.03944	-.04259	-.06020	13.66719
GRADIENT		.00199	.00018	.00018	.00000	-.00000	.00027	.00016	.00013

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(APV018) ( 24 OCT 75 )

TABULATED SOURCE DATA, LARC UPUT 1243 (04-75)  
04-75, UPUT1243,088 (B19CTF5MON19) (N40TEZ3) (VTR5)

## REFERENCE DATA

SREF = 87.1260 SQ. IN. XREF = 12.5800 INCHES  
 LREF = 7.1222 INCHES YREF = .0000 INCHES  
 BREF = 14.0000 INCHES ZREF = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = 5.0000 WACH = 4.0000  
 RNVL = 1.720 ELEV0 = -47.0000  
 ALRON = .0000 EDPLAP = -14.250  
 RUOFLR = 40.0000 RUOCCR = .0000

RUN NO. 37/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
.326	15.749	202.42480	-.03810	-.03974	-.05432	-.03944	-.04728	-.06491	13.66827
.102	20.946	202.40220	-.04285	-.03973	-.05432	-.03944	-.04728	-.06963	13.66474
.326	26.168	202.43837	-.04285	-.03974	-.05903	-.03944	-.04728	-.06491	13.66719
.101	31.433	202.37958	-.04284	-.03973	-.05432	-.03943	-.04727	-.05549	13.66322
.103	36.686	202.39767	-.03810	-.03973	-.05432	-.03944	-.04728	-.05549	13.66444
GRADIENT		-.00147	.00000	.00000	.00000	.00000	.00000	.00063	-.00010

RUN NO. 39/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
69.471	15.722	202.43837	-.03810	-.03974	-.05432	-.03944	-.04728	-.06963	13.66719
70.142	20.946	202.44741	-.03810	-.03974	-.05903	-.03944	-.04259	-.06963	13.66780
69.919	26.177	202.45193	-.03810	-.03973	-.05903	-.03944	-.04728	-.06962	13.66810
69.699	31.419	202.41576	-.03810	-.03973	-.05903	-.03944	-.04728	-.06963	13.66566
69.699	36.686	202.42480	-.03810	-.03974	-.05963	-.03944	-.04259	-.05078	13.66627
GRADIENT		-.00112	-.00000	.00000	-.00018	-.00008	.00009	.00072	-.00054

RUN NO. 40/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	Q (PSF)	CFB	CFSC	CFBM	CF17	CF19	CF20	FINF
163.010	15.741	202.44743	-.04285	-.03974	-.05432	-.03944	-.04728	-.06491	13.66780
163.438	20.943	202.40220	-.03810	-.03973	-.05432	-.04414	-.04728	-.06020	13.66474
163.906	26.168	202.41124	-.03810	-.03973	-.05432	-.03944	-.04728	-.05549	13.66535
164.352	31.429	202.40220	-.03810	-.03973	-.05432	-.03944	-.04728	-.05078	13.66474
164.576	36.650	202.42028	-.03810	-.03974	-.05432	-.03944	-.04728	-.06491	13.66597
GRADIENT		-.00104	.00018	.00009	.00000	.00009	-.00000	.00018	-.00007



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TABULATED SOURCE DATA, LARC UPMT 1943 (0A-75)

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0A-75, UPMT1943,0R2(B19C7F5M0N19) (M107E23) (VTR5)

RPV0531 ( 04 OCT 75 )

## REFERENCE DATA

SREP = 07.1500 SQ. IN. YMRP = 12.5000 INCHES  
LREP = 7.1222 INCHES YMRP = .0000 INCHES  
BREP = 14.0500 INCHES ZMRP = 6.0000 INCHES  
SCALE = .0150

## PARAMETRIC DATA

BETA = -5.000 WACH = 2.500  
RNAL = 1.720 ELEVTR = .000  
ATLORN = .000 EOLAP = .000  
PUETLR = 40.000 FUECOR = .000

RUN NO. 2/ 0 RNAL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CY	CL	CE	L/C
.316	15.506	-5.17892	.46984	.09676	-.03565	.00699	.00196	.07426	.42616	.21884	1.95562
.316	20.919	-5.18797	.67522	.08823	-.04775	.00766	.00741	.06826	.59921	.32351	1.65222
.092	26.419	-5.19736	.89841	.07893	-.05967	.00802	.01269	.06199	.76950	.47297	1.65393
.316	31.794	-5.19285	1.12195	.07504	-.07196	.00893	.01446	.05666	.91659	.65266	1.45387
.092	37.296	-5.19566	1.36583	.06120	-.09009	.01228	.01141	.06318	1.04945	.87629	1.19761
GRADIENT	-.00078	.04111	-.00164	-.00244	.00023	.00048	.00048	-.00063	.02859	.03016	-.05370

RUN NO. 3/ 0 RNAL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CY	CL	CE	L/C
63.198	15.503	-5.17982	.47278	.09685	-.03562	.00656	.00183	.07386	.42959	.21969	1.95587
62.078	20.918	-5.19104	.67556	.08808	-.04835	.00770	.00728	.07004	.59919	.32346	1.65357
62.526	26.313	-5.19947	.89495	.07881	-.05851	.00803	.01227	.06381	.76724	.46744	1.64136
62.974	31.801	-5.19880	1.12321	.06969	-.07184	.00893	.01418	.05953	.91718	.65213	1.45367
63.421	37.285	-5.20422	1.36245	.06116	-.08864	.01211	.01084	.06834	1.04697	.87400	1.19791
GRADIENT	-.00304	.04591	-.00165	-.00238	.00023	.00046	.00046	-.00047	.02831	.03007	-.05601

RUN NO. 4/ 0 RNAL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CY	CL	CE	L/C
227.005	15.495	-5.18127	.47401	.09694	-.03645	.00648	.00178	.07662	.43038	.22005	1.95579
225.214	20.910	-5.18946	.67674	.08752	-.04856	.00771	.00734	.06910	.60093	.32328	1.65284
225.886	26.361	-5.19991	.89780	.07815	-.05999	.00811	.01242	.06567	.76916	.46597	1.64278
226.761	31.779	-5.19895	1.11521	.06965	-.06942	.00860	.01364	.06052	.91134	.64552	1.45360
226.991	37.321	-5.20667	1.36375	.06117	-.08901	.01164	.01050	.07018	1.04744	.87546	1.19844
GRADIENT	-.00311	.04068	-.00164	-.00231	.00021	.00043	.00043	-.00039	.02831	.02999	-.05559

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TABULATED SOURCE DATA, LARC UPWT 1243 (04-73)

(04-73, UPWT1243,08B (B19C7F5M6M19) (N12JTE23) (V7R53)

## PARAMETRIC DATA

## REFERENCE DATA

SPEY = 87.1500 SBL IN. XMRP = 12.5800 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.5500 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0130

BETA = .0000  
 RNUL = 1.72  
 AILPON = .0000  
 PUFLUR = 40.0000

RUN NO. 5/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CC	L/C
.092	15.127	-.00164	.47137	.09790	-.03290	-.00016	-.00039	.00213	.42796	.00000	1.94782
.092	20.695	-.00297	.67063	.08935	-.04267	.00013	-.00138	.00433	.59464	.00000	1.84598
.092	26.360	-.00427	.89304	.08003	-.05937	.00028	-.00184	.00624	.76459	.00000	1.63297
.092	31.784	-.01024	1.12206	.06794	-.07040	.00019	-.00234	.00998	.91671	.00000	1.41504
.092	37.286	-.01660	1.36623	.05936	-.08877	-.00074	-.00131	.01031	1.0164	.00000	1.25712
GRADIENT		-.00768	.04120	-.00185	-.00259	-.00002	-.00004	.00040	.02887	.00000	-.00000

RUN NO. 6/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CC	L/C
69.464	15.498	.00041	.47112	.09778	-.03227	-.00013	-.00066	.00131	.42786	.00000	1.94397
51.114	20.931	-.00336	.67070	.08942	-.04246	.00007	-.00124	.00441	.59466	.00000	1.84310
60.289	26.341	-.00419	.89442	.07923	-.05975	.00037	-.00182	.00616	.76640	.00000	1.63279
61.408	31.783	-.01022	1.12282	.06787	-.07054	.00001	-.00237	.01003	.91915	.00000	1.41593
63.884	37.293	-.01666	1.36685	.05791	-.08869	-.00097	-.00134	.01042	1.05228	.00000	1.25751
GRADIENT		-.00073	.04121	-.00186	-.00259	-.00003	-.00005	.00044	.02888	.00000	-.00000

RUN NO. 7/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CC	L/C
228.270	15.492	.00184	.46778	.09781	-.03171	-.00016	-.00060	.00050	.42466	.00000	1.93723
218.949	20.914	-.00225	.67208	.08862	-.04271	.00008	-.00131	.00445	.59615	.00000	1.84748
214.249	26.390	-.00660	.89145	.07753	-.05478	-.00013	-.00264	.00899	.76417	.00000	1.64145
222.733	31.791	-.01275	1.12064	.06751	-.06897	-.00048	-.00270	.01193	.91695	.00000	1.41506
228.781	37.308	-.01783	1.36680	.05768	-.08880	-.00168	-.00068	.00959	1.05221	.00000	1.25754
GRADIENT		-.00091	.04122	-.00186	-.00258	-.00007	-.00002	.00047	.02883	.00000	-.00000





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TABULATED SOURCE DATA, LARC UPMT 1043 (OM-73)

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OM-73, UPMT1043, JRB(B19CTFS-6419) (M07E23) (V7R3)

(RPV033) ( 24 OCT 73 )

REFERENCE DATA

SPOT = 87.1500 SBL IN. XSEP = 12.5800 INCHES  
L'OF = 7.1222 INCHES YSEP = .0000 INCHES  
ZSEP = 14.5500 INCHES ZSEP = 6.0000 INCHES  
SCALE = .0150

BETA =  
FNUL =  
ALURM =  
FUDFLR =

PARAMETRIC DATA

5.000 WACH = 2.500  
1.720 ELEVTR = .000  
.000 ECHLAP = .000  
40.000 FUDCEP = .000

RUN NO. 01 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CC	L/D
.092	15.480	5.18734	.47203	.09617	-.03723	-.00672	-.00302	-.07149	.43924	.21887	1.96586
	25.911	5.19282	.67784	.06685	-.04992	-.00704	-.00953	-.06147	.60215	.32506	1.86402
	26.349	5.19643	.90060	.07863	-.05923	-.00736	-.01495	-.05169	.77181	.47073	1.63971
	31.621	5.19506	1.12737	.06976	-.07167	-.00970	-.01686	-.04391	.92585	.65354	1.40908
	37.282	5.18379	1.36849	.06078	-.09098	-.01236	-.03188	-.04783	1.09204	.87733	1.19317
GRADIENT		-.00010	.04113	-.00161	-.00237	-.00725	-.00054	-.00120	.02870	.05025	-.00007

RUN NO. 94 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CC	L/D
69.243	15.492	5.18593	.47396	.09609	-.03751	-.00679	-.00309	-.07063	.43069	.21928	1.96586
52.059	25.911	5.19283	.67812	.08689	-.04995	-.00767	-.00954	-.06147	.60244	.32519	1.86402
67.430	26.373	5.19693	.89369	.07815	-.05950	-.00799	-.01115	-.04973	.76777	.46786	1.64134
63.196	31.846	5.18959	1.12832	.06911	-.07195	-.00991	-.01674	-.04391	.92201	.65405	1.40969
63.408	37.279	5.18374	1.36643	.06077	-.09073	-.01279	-.03191	-.04753	1.09245	.87600	1.19314
GRADIENT		-.00014	.04102	-.00162	-.00236	-.00726	-.00054	-.00117	.02860	.05016	-.00047

RUN NO. 10 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CC	L/D
223.649	15.480	5.18731	.47197	.09580	-.03720	-.00666	-.00301	-.07148	.42928	.21830	1.96549
216.729	25.888	5.19431	.67486	.08539	-.04826	-.00778	-.00999	-.06228	.59995	.32059	1.87157
223.606	26.366	5.19593	.89258	.07784	-.05746	-.00830	-.01580	-.04876	.76316	.45614	1.64147
229.467	31.802	5.18942	1.12365	.06946	-.07233	-.01026	-.01633	-.04473	.91836	.65117	1.41032
224.944	37.288	5.18420	1.36514	.06076	-.09053	-.01296	-.01421	-.04571	1.04925	.87537	1.19359
GRADIENT		-.00031	.04099	-.00158	-.00240	-.00728	-.00054	-.00127	.02856	.05017	-.00053

DATE 29 OCT 73

TABULATED SOURCE DATA, LARC UPNR 1243 (OM-73)

OM-73, UPNR1043, ORB1819CTF5M0N131, R4107E231 (VTR-3)

(RPV024) ( 24 OCT 73 )

## REFERENCE DATA

SREF = 87.1380 SA. IN. XMRP = 12.1600 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 DREF = 14.0570 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = -5.000 WAOA = 2.500  
 RNVL = 1.720 ELEVTR = -20.000  
 AIRPOL = .000 DDTLAP = -14.200  
 RUFLR = 40.000 RUFLER = .000

RUN NO. 53/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-LET	ALPHA	BETA	CN	CA	CLM	CEL	CYN	CY	CL	CD	L/D
.272	15.431	-5.17364	.41006	.09823	.01043	.02468	.00126	.07423	.36893	.22418	1.07683
.272	20.877	-5.16771	.62483	.08617	.02617	.02360	.00685	.08940	.53447	.29608	1.80516
.272	26.319	-5.19732	.81864	.07374	.02444	.02584	.01209	.06337	.77019	.40593	1.68313
.049	34.774	-5.19783	1.03380	.06392	.02216	.02679	.01363	.06014	.84521	.53870	1.41174
.273	37.239	-5.25409	1.26176	.05228	-.00155	.01084	.01050	.06908	.97239	.80350	1.20784
GRADIENT		-.00130	.03918	-.00210	-.00031	.00023	.00546	-.00036	.02788	.02787	-.02928

RUN NO. 54/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-LET	ALPHA	BETA	CN	CA	CLM	CEL	CYN	CY	CL	CD	L/D
9.051	15.468	-5.17876	.45820	.09824	.01008	.02468	.00112	.07393	.36722	.20339	1.80409
65.818	20.865	-5.18625	.62248	.08995	.02645	.02560	.00591	.06848	.53216	.29429	1.81228
63.137	26.325	-5.19969	.81242	.07329	.02496	.02594	.01211	.06402	.69478	.42776	1.52421
64.723	31.727	-5.19826	1.02481	.06399	.02423	.02661	.01314	.06179	.81502	.53324	1.41214
37.382	37.236	-5.23489	1.25552	.05223	-.00108	.01054	.01058	.06928	.96798	.80129	1.20403
GRADIENT		-.00125	.03892	-.00210	-.00045	.00023	.00546	-.00037	.02773	.02747	-.02913

RUN NO. 55/ 0 RNVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-LET	ALPHA	BETA	CN	CA	CLM	CEL	CYN	CY	CL	CD	L/D
228.009	15.446	-5.17815	.41162	.09869	.01009	.02467	.00106	.07675	.37047	.20479	1.80389
229.841	20.861	-5.18769	.62653	.08976	.02592	.02589	.00586	.06917	.53613	.29540	1.81383
229.399	26.320	-5.19809	.81814	.07481	.02448	.02602	.01224	.06327	.77016	.42980	1.62972
229.175	31.773	-5.19979	1.03264	.06413	.02423	.02624	.01314	.06209	.84411	.53829	1.41139
224.727	37.262	-5.25359	1.26196	.05227	-.00005	.02668	.01020	.06945	.97272	.80167	1.20795
GRADIENT		-.00115	.03901	-.00210	-.00040	.00019	.00545	-.00040	.02774	.02749	-.02929

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TABULATED SOURCE DATA, LARC UPAR 1043 (04-73)

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(PPWD33) ( 04 OCT 73 )

04-73, UPLT1043,088 (019CTF51043:19) (M01TE23) (V0157)

REFERENCE DATA

SREF = 07.1500 50.1N. XMRP = 12.5000 INCHES  
LREF = 7.1222 INCHES YMRP = .0000 INCHES  
SREF = 14.0000 INCHES ZMRP = 6.0000 INCHES  
SCALE = .0150

PARAMETRIC DATA

BETA = .000 WACH = 2.500  
RM/L = 1.72 ELEV8 = -27.000  
AIRLON = .000 EXPLAP = -14.250  
RUCPLR = 40.000 FUSDER = .000

RUN NO. 56/ 0 RM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
.277	15.469	.01912	.41016	.09801	.01323	-.00015	-.00008	.00036	.36703	.25434	1.05592
.277	20.894	.01375	.60350	.08703	.01014	.00007	-.00060	.00359	.33277	.29806	1.79893
.053	26.364	.01517	.81478	.07333	.00688	.00028	-.00156	.00557	.63639	.42332	1.62225
.053	31.784	.01133	1.03089	.06137	.00406	.00042	-.00183	.00786	.84398	.59515	1.41579
.053	37.233	.00795	1.25832	.04856	-.00167	-.00077	-.00034	.00582	.97220	.79397	1.21266
GRADIENT	-.00049		.03993	-.00231	-.00066	-.00002	-.00003	.00028	.02790	.02739	-.02365

RUN NO. 57/ 0 RM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
02.711	15.481	.02107	.40767	.09832	.01322	-.00037	-.00015	-.00040	.35664	.27337	1.80101
02.712	20.983	.01840	.60621	.08664	.00960	.00017	-.00099	.00273	.53498	.29798	1.79339
02.488	26.398	.01573	.81438	.07466	.00743	.00019	-.00170	.00561	.69629	.42395	1.62222
03.303	31.743	.01120	1.02783	.06146	.00484	-.00012	-.00187	.00832	.84178	.59372	1.41345
03.159	37.243	.00741	1.25898	.04854	-.00117	-.00013	-.00025	.00589	.97287	.80057	1.21223
GRADIENT	-.00064		.03913	-.00230	-.00062	-.00004	-.00002	.00033	.02799	.02743	-.02343

RUN NO. 58/ 0 RM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
226.743	15.458	.02263	.41009	.09855	.01267	-.00007	-.00007	-.00132	.35899	.27429	1.80524
226.742	20.903	.01725	.60267	.08672	.01014	.00011	-.00073	.00272	.53276	.29806	1.79713
226.294	26.336	.01581	.80969	.07490	.00688	-.00047	-.00227	.00674	.63242	.42633	1.62414
226.071	31.770	.01018	1.02472	.06156	.00523	-.00087	-.00217	.00913	.83977	.59156	1.41717
225.622	37.221	.00311	1.25520	.04866	-.00160	-.00169	-.00018	.00780	.97229	.79397	1.21554
GRADIENT		-.00085	.03884	-.00230	-.00062	-.00009	-.00003	.00045	.02774	.02727	-.02370

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TABULATED SOURCE DATA, LARC UPMT 1043 (0A-73)

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0A-73, UPMT1043,085(819CTF3W0M19) (M1JTE23) (VTR3)

PP40901 ( 04 OCT 73 )

## REFERENCE DATA

SPEF = 07.1990 50.1M. XMRP = 12.5893 INCHES  
 LPEF = 7.1222 INCHES YMRP = .0000 INCHES  
 BPEF = 14.0770 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0153

BETA = 5.0990 P40M = 2.570  
 RV/L = 1.725 ELEV8R = -20.000  
 ATLONR = .0000 E7SLAP = -14.250  
 FUDFLR = 49.000 FUDWR = .000

## PARAMETRIC DATA

RUN NO. 59/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CE	L/C
.276	15.449	5.18712	.40843	.08861	.00802	-.00513	-.00128	-.07508	.36837	.25411	1.80474
.276	29.888	5.19740	.60615	.08594	.00496	-.00374	-.00722	-.06429	.53591	.25815	1.87935
.052	26.337	5.19911	.81788	.07529	.00262	-.00391	-.01351	-.05512	.69974	.40708	1.82694
.052	31.772	5.19211	1.03304	.06386	.00239	-.00282	-.01495	-.04900	.84461	.25803	1.41196
.052	37.228	5.18246	1.25815	.05301	-.00298	-.01138	-.01212	-.01374	.56372	.80018	1.80705
GRADIENT		-.00014	.03991	-.00208	-.00046	-.00028	-.00054	.00117	.02775	.02775	-.00028

RUN NO. 60/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CE	L/C
63.159	15.452	5.18775	.41087	.08872	.00858	-.00513	-.00142	-.07555	.36972	.25482	1.80487
67.187	29.878	5.19740	.60397	.08578	.00495	-.00374	-.00721	-.06418	.53581	.25811	1.87943
61.368	26.349	5.19465	.81638	.07484	.00393	-.00218	-.01423	-.05151	.69935	.40941	1.82629
60.697	31.778	5.19153	1.03121	.06347	.00193	-.00321	-.01476	-.04900	.84519	.25772	1.41272
62.711	37.222	5.18237	1.25485	.05288	-.00245	-.01132	-.01211	-.01371	.56724	.80118	1.80777
GRADIENT		-.00018	.03882	-.00209	-.00047	-.00027	-.00053	.00117	.02785	.02744	-.00031

RUN NO. 61/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CE	L/C
229.534	15.454	5.18568	.41087	.08854	.00966	-.00508	-.00135	-.07421	.36875	.25448	1.80241
227.862	29.895	5.19107	.60901	.08532	.00496	-.00386	-.00735	-.06418	.53553	.25891	1.87974
226.296	26.379	5.19514	.81114	.07498	.00324	-.00637	-.01441	-.05132	.69148	.40872	1.82679
224.282	31.803	5.18645	1.03241	.06386	.00113	-.00374	-.01429	-.04782	.84373	.25439	1.41112
226.072	37.234	5.18033	1.25557	.05302	-.00275	-.01153	-.01208	-.01490	.98785	.81118	1.80742
GRADIENT		-.00028	.03882	-.00207	-.00053	-.00029	-.00052	.00121	.02753	.02748	-.00030



DATE 09 OCT 73

TABULATED SOURCE DATA, LARC UPLINK 1043 (3A-73)

PAGE 23

0A-73, UPLINK 1043, OFB (B19C775MON19) (AC07E23) (V7B5)

09A0073 ( 24 OCT 73 )

## REFERENCE DATA

SRP = 87.1505 SQ. IN. YMRP = 12.5000 INCHES  
LREF = 7.1222 INCHES YMRP = .0000 INCHES  
DREF = 14.0000 INCHES ZMRP = 6.0000 INCHES  
SCALE = .0150

## PARAMETRIC DATA

BETA = -5.0000 YMRP = 2.5000  
RVL = 1.7200 ELEV = -40.0000  
ALUPON = .0000 ELEV = -16.0000  
FUTL = 40.0000 ELEV = .0000

RUN NO. 22/ 0 RVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CEL	CYN	CY	CL	C	LC
1.009	15.452	-5.17901	.38366	.10226	.02344	.00281	-.00052	.00074	.34114	.20662	1.65110
.784	20.672	-5.19152	.58726	.09611	.02036	.00415	.00567	.00320	.51448	.29973	1.72249
.561	26.362	-5.20131	.79553	.08583	.02027	.00480	.01145	.00644	.67471	.43214	1.72880
.785	31.733	-5.19872	1.01018	.07714	.02184	.00571	.01311	.00666	.81276	.59754	1.73157
.568	37.234	-5.20358	1.23337	.06586	.02254	.00613	.01544	.00695	.94268	.80075	1.73928
GRADIENT		-.00101	.03307	-.00191	-.00010	.00026	.00054	-.00065	.02774	.02709	-.02074

UN NO. 23/ 0 RVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CEL	CYN	CY	CL	C	LC
87.693	15.477	-5.17869	.38697	.10812	.02490	.00219	-.00031	.00076	.34409	.20746	1.65958
63.691	20.671	-5.19249	.58632	.09559	.01995	.00407	.00564	.00405	.51395	.29809	1.72239
63.691	26.333	-5.20325	.79754	.08519	.01969	.00476	.01152	.00579	.67638	.42715	1.72830
63.667	31.738	-5.19812	1.01040	.07657	.02112	.00554	.01294	.00660	.81982	.59432	1.73179
63.691	37.264	-5.20322	1.23237	.06551	.02168	.00688	.01025	.00632	.94112	.79332	1.73829
GRADIENT		-.00106	.03883	-.00191	-.00009	.00025	.00052	-.00062	.02732	.02819	-.02409

RUN NO. 24/ 0 RVL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CEL	CYN	CY	CL	C	LC
225.683	15.483	-5.17990	.39317	.10814	.02339	.00280	-.00052	.00060	.34340	.20649	1.65350
225.683	20.657	-5.19203	.58465	.09510	.01956	.00407	.00553	.00411	.51249	.29702	1.72839
227.295	26.298	-5.20248	.79254	.08482	.02058	.00480	.01127	.00579	.67692	.42717	1.72831
226.802	31.732	-5.19382	1.01457	.07669	.02226	.00512	.01224	.00582	.81407	.59359	1.73145
224.118	37.232	-5.20494	1.22783	.06550	.02279	.00625	.00961	.00610	.94768	.79356	1.73934
GRADIENT		-.00107	.03876	-.00191	-.00007	.00022	.00050	-.00055	.02749	.02711	-.02070

DATE 29 OCT 72

## TABULATED SOURCE DATA, LARC UPWT 1043 (0A-72)

0A-72, UPWT1043, OFB (81907546N191 N487223) (W703)

REMARKS: 1 04 OCT 72 1

## REFERENCE DATA

SPDF = 87.1500 SQ. IN. WHP = 12.5000 INCHES  
 LSPF = 7.1222 INCHES WHP = .0000 INCHES  
 SPDF = 14.0570 INCHES WHP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = .000  
 FM/L = 1.72  
 AT/LON = .000  
 FID/LP = 40.000

RUN NO. 28/ 0 FM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CT	CL	CC	LC
84.339	15.494	.01411	.38260	.10738	.02794	.00019	.00005	.00249	.24001	.20569	1.65073
62.771	21.682	.01849	.57740	.09618	.02598	.00242	-.00022	.07231	.27244	.28933	1.71245
63.442	26.297	.01891	.78375	.09574	.02407	.00032	-.00118	.06142	.66644	.42437	1.55270
63.690	31.789	.01407	1.00207	.07430	.02286	.00028	-.00128	.00563	.81263	.03102	1.37487
64.114	37.212	.00770	1.22578	.06201	.02252	-.00031	-.00048	.00622	.91471	.00700	1.18219
GRADIENT		-.00022	.03891	-.00207	-.00029	-.00062	-.00002	.00020	.00765	.00005	-.00005

RUN NO. 29/ 0 FM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CT	CL	CC	LC
84.339	15.441	.01574	.38613	.10763	.02792	.00116	.00012	.00157	.24004	.20555	1.65070
62.771	21.692	.02170	.58239	.09592	.02432	.00114	-.00057	.00049	.27341	.29719	1.71170
63.442	26.277	.01337	.79533	.09514	.02406	.00179	-.00179	.00691	.66713	.42442	1.55215
63.690	31.787	.01049	1.00175	.07440	.02421	.00113	-.00142	.00742	.81231	.00591	1.37482
64.114	37.198	.00556	1.22515	.06229	.02271	.00111	-.00045	.00718	.91403	.00711	1.18215
GRADIENT		-.00018	.03855	-.00205	-.00024	-.00111	-.00003	.00013	.00742	.00005	-.00005

RUN NO. 30/ 0 FM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CT	CL	CC	LC
229.593	15.489	.01724	.38599	.10761	.02794	.00000	.00019	.00071	.24021	.20579	1.65080
227.474	21.641	.01848	.57769	.09593	.02401	.00145	-.00031	.00021	.27250	.28978	1.71170
228.012	26.368	.01456	.78733	.09494	.02494	.00025	.00009	.00591	.66769	.42479	1.55070
229.490	31.749	.00795	.99791	.07484	.02514	-.00000	.00019	.00319	.81247	.00714	1.18247
227.697	37.177	.00144	1.22475	.06214	.02117	-.00127	.00009	.00072	.91411	.00710	1.18230
GRADIENT		-.00076	.03864	-.00206	-.00021	-.00002	.00000	.00044	.00711	.00005	-.00005



DATE 09 OCT 73

TABULATED SOURCE DATA, LARC UPAGE 1043 (0A-73)

(RPV739) ( 24 OCT 73 )

0A-73, UPAGE 1043, ORB (B:9C7F5MON9) (M:07E23) (V7B3)

## REFERENCE DATA

SREF = 67.1500 SQ. IN. XREF = 12.5000 INCHES  
LREF = 7.1222 INCHES YREF = .0000 INCHES  
BREF = 14.0300 INCHES ZREF = 6.0000 INCHES  
SCALE = .0150

## PARAMETRIC DATA

BETA = 5.000 RADN = 2.590  
RM/L = 1.720 ELEVTR = -40.000  
AIRLON = .000 BOFLAP = -14.250  
PIEFLLR = 40.000 FIDDER = .000

RUN NO. 25/ 0 RM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CC	L/D
63.892	15.461	5.18336	.38041	.10932	.02417	-.00250	.00025	-.07650	.33750	.25687	1.63221
.338	20.849	5.19438	.58106	.09545	.01891	-.00367	-.00569	-.06927	.59979	.25620	1.72583
.338	26.283	5.20091	.78903	.08564	.01934	-.00423	-.01277	-.07353	.66954	.42618	1.57102
.115	31.732	5.19362	1.00375	.07626	.02020	-.00616	-.01399	-.03171	.81529	.59383	1.37894
.114	37.268	5.18778	1.22621	.06824	.02018	-.00916	-.01162	-.05435	.93566	.79358	1.17755
GRADIENT	.00014	.03882	-.00194	-.00029	-.00032	-.00029	-.00039	.00113	.02756	.02707	-.02313

RUN NO. 26/ 0 RM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CC	L/D
63.892	15.423	5.18338	.38035	.10928	.02416	-.00261	.00025	-.07649	.33760	.25647	1.63509
63.893	20.891	5.19437	.58164	.09559	.01891	-.00364	-.00569	-.06925	.57650	.25624	1.71997
63.867	26.292	5.19744	.78760	.08328	.02029	-.00454	-.01329	-.05482	.66835	.42532	1.57147
63.221	31.763	5.19307	1.00415	.07599	.02040	-.00632	-.01566	-.03165	.81377	.59320	1.37183
63.892	37.240	5.18661	1.22306	.06615	.02068	-.00940	-.01138	-.05429	.93366	.79260	1.17787
GRADIENT	.00809	.03867	-.00193	-.00030	-.00010	-.00030	-.00038	.00114	.02745	.02697	-.02218

RUN NO. 27/ 0 RM/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CC	L/D
227.027	15.413	5.18888	.37843	.10874	.02447	-.00260	.00056	-.07992	.33576	.25568	1.63244
224.789	20.863	5.19282	.58005	.09482	.01860	-.00371	-.00576	-.06935	.57824	.25520	1.72166
226.378	26.313	5.19495	.78558	.08517	.02104	-.00483	-.01359	-.05297	.66544	.42457	1.55949
228.379	31.730	5.19393	.98886	.07625	.01997	-.00702	-.01364	-.03150	.80947	.59117	1.37158
225.237	37.215	5.18252	1.22067	.06653	.02083	-.00972	-.01176	-.03150	.93188	.79125	1.17775
GRADIENT	-.00025	.03868	-.00189	-.00032	-.00011	-.00032	-.00060	.00135	.02746	.02697	-.02318

OA-73, UPLAT1943, ORB (219CTFSW619) (NADITEZ3) (NTR5)

(RPV010) ( 24 OCT 73 )

## REFERENCE DATA

SPOT = 67.1500 SQ. IN. XMRP = 12.1605 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.0150 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = -5.000 WACH = 4.670  
 RNUL = 1.720 ELEVTR = .000  
 ALUPOR = .000 EXPLAP = .000  
 RUOTLR = 40.000 FUDDER = .000

RUN NO. 11/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CT	CL	CD	L/D
315	15.839	-5.08271	.32379	.06636	-.01582	.00621	.00150	.05878	.29323	.15274	1.93292
315	20.983	-5.08562	.59424	.16263	-.01941	.00729	.00414	.05610	.44630	.23505	1.87581
315	26.212	-5.08633	.79514	.05974	-.02603	.00811	.00604	.05301	.67624	.36503	1.68789
315	31.460	-5.08858	.92448	.05649	-.03824	.00923	.00638	.05431	.75911	.55567	1.43247
315	36.708	-5.09062	1.15733	.05287	-.05577	.01149	.00607	.05377	.89598	.73599	1.22070
GRADIENT		-.00036	.03990	-.00063	-.00189	.00024	.00022	-.00015	.02896	.02787	-.02583

RUN NO. 12/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CT	CL	CD	L/D
70.359	15.827	-5.08423	.32276	.06379	-.01535	.00608	.00137	.06041	.29258	.15132	1.93348
69.463	26.180	-5.08641	.70221	.05926	-.02551	.00811	.00603	.05306	.60397	.36357	1.66351
70.359	31.412	-5.08859	.91857	.05587	-.03516	.00890	.00634	.05440	.75483	.52643	1.43125
70.582	36.729	-5.09058	1.15412	.05212	-.05430	.01111	.00603	.05582	.83392	.73198	1.22111
GRADIENT		-.00030	.03957	-.00065	-.00177	.00023	.00023	-.00023	.02892	.02727	-.02474

RUN NO. 13/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CT	CL	CD	L/D
162.556	15.738	-5.08273	.32294	.06643	-.01531	.00615	.00150	.05882	.29281	.15153	1.93220
163.699	20.951	-5.08570	.59117	.06253	-.01889	.00735	.00413	.05615	.44558	.23760	1.87570
163.676	26.206	-5.08784	.69906	.05914	-.02512	.00760	.00586	.05466	.60109	.36177	1.66155
163.229	31.456	-5.08954	.91828	.05590	-.03525	.00860	.00619	.05596	.75416	.52689	1.43134
162.781	36.694	-5.09056	1.15096	.05203	-.05277	.01078	.00599	.05587	.89180	.72947	1.22253
GRADIENT		-.00038	.03955	-.00068	-.00174	.00020	.00021	-.00012	.02874	.02757	-.02557





DATE 28 OCT 73

TABULATED SOURCE DATA, LARC UPWT 1243 (0A-TD)

0A-TD, UPWT1243,ORB(219C7F5H0K19) (M07E23) (VTR5)

(BPV011) (24 OCT 73)

## REFERENCE DATA

SREF = 87.1500 SQ. IN. XREF = 12.5800 INCHES  
LREF = 7.1222 INCHES YREF = .0000 INCHES  
BREF = 14.0000 INCHES ZREF = 6.0000 INCHES  
SCALE = .0150

## PARAMETRIC DATA

BETA = .071 MULCH = 4.800  
RN/L = 1.72 ELEVTR = .000  
AULSON = .000 EDSFLAP = .000  
PUFLR = 40.000 FUDTER = .000

RUN NO. 14/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
317	15.745	-.00260	32087	.06896	-.01578	.00015	-.00041	.00335	.29090	.15063	1.93098
317	25.960	-.00513	.49917	.06245	-.01862	-.00021	-.00060	.00616	.44360	.23687	1.87929
.094	26.210	-.00624	.70332	.05937	-.02333	.00017	-.00122	.00866	.60479	.36189	1.66199
.094	34.461	-.00691	.92514	.05337	-.03777	.00004	-.00092	.00667	.76023	.51356	1.43419
317	36.743	-.01042	1.15800	.05184	-.05579	-.00023	-.00076	.01046	.89892	.73430	1.22247
GRADIENT		-.00033	.04051	-.00068	-.00189	-.00001	-.00002	.00032	.02911	.02763	-.03541

RUN NO. 15/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
73.137	15.737	-.00256	32094	.06615	-.01475	.00004	-.00042	.00332	.29097	.15071	1.93063
71.480	20.964	-.00456	.50224	.06243	-.01807	.00005	-.00084	.00617	.44666	.23798	1.87885
72.398	26.186	-.00623	.70043	.05921	-.02546	-.00005	-.00125	.00869	.60243	.36223	1.66359
64.766	31.447	-.00738	.92252	.05316	-.03735	-.00040	-.00071	.00864	.75824	.52835	1.43510
63.214	36.707	-.01144	1.15491	.05142	-.05532	-.00067	-.00030	.01039	.89517	.73153	1.22369
GRADIENT		-.00036	.03984	-.00070	-.00192	-.00004	-.00001	.00032	.02899	.02771	-.03541

RUN NO. 16/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
161.867	15.738	-.00205	32083	.06600	-.01474	.00010	-.00067	.00340	.29086	.15064	1.93092
163.453	20.961	-.00442	.49947	.06215	-.01768	-.00055	-.00089	.00614	.44419	.23671	1.87848
164.372	26.190	-.00762	.70319	.05915	-.02497	-.00029	-.00138	.01020	.60490	.36343	1.66343
164.372	31.423	-.00896	.92256	.05328	-.03739	-.00064	-.00085	.01023	.75841	.52816	1.43589
164.796	36.744	-.01134	1.15761	.05116	-.05483	-.00089	-.00031	.01032	.89700	.73353	1.22285
GRADIENT		-.00044	.03999	-.00070	-.00191	-.00004	-.00001	.00034	.02911	.02781	-.03542

RUN NO. 17/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CBL	CYN	CY	CL	CD	L/D
397.752	15.765	-.00331	32088	.06593	-.01479	-.00009	-.00081	.00497	.29090	.15063	1.93121
413.416	20.961	-.00588	.49937	.06212	-.01773	-.00074	-.00103	.00771	.44410	.23585	1.87861
423.577	26.199	-.00701	.70027	.05880	-.02350	-.00072	-.00168	.01029	.60237	.36192	1.66438
447.546	31.399	-.00931	.91656	.05482	-.03649	-.00136	-.00066	.01025	.75379	.52430	1.43770
446.983	36.730	-.01286	1.15191	.05084	-.05399	-.00140	-.00049	.01203	.89255	.72936	1.22275
GRADIENT		-.00042	.03968	-.00072	-.00185	-.00006	-.00002	.00032	.02897	.02762	-.03545

DATE 09 OCT 73

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TABULATED SOURCE DATA, LARC UPST 1043 (0A-70)

(PPV012) ( 24 OCT 73 )

0A-70, UPST1043, ORB (B19CTF3M6H19) (N07E23) (VTR2)

## PARAMETRIC DATA

BETA = 5.000 MACH = 4.600  
 RV/L = 1.720 ELEVTR = 1.000  
 ALURON = .000 EXFLAP = .000  
 FUFLR = 40.000 FUSSEN = .000

## REFERENCE DATA

SPEP = 87.1500 SQ. IN. XWRP = 12.5800 INCHES  
 LPEP = 7.1222 INCHES WWRP = .0000 INCHES  
 SPEP = 14.0500 INCHES ZWRP = 6.0000 INCHES  
 SCALE = .0120

RUN NO. 18/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CD	L/D
317	15.774	5.08931	.32199	.06684	-.01691	-.00374	-.00262	-.05202	.29189	.19185	1.92748
317	20.948	5.08634	.50261	.06313	-.02097	-.00707	-.00543	-.04337	.44495	.23794	1.87003
317	26.229	5.08632	.70158	.06015	-.02700	-.00807	-.00748	-.03910	.67276	.36403	1.65580
094	31.455	5.08410	.92891	.05664	-.03866	-.00970	-.00750	-.03707	.75652	.52888	1.42948
094	36.800	5.08192	1.14728	.05205	-.04592	-.01117	-.00831	-.03445	.86748	.72893	1.21750
GRADIENT		-.00037	.03941	-.00069	-.00183	-.00026	-.00025	.00083	.02859	.02751	-.03517

RUN NO. 19/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CD	L/D
58.277	15.724	5.08901	.31911	.06605	-.01633	-.00397	-.00259	-.05208	.28913	.19054	1.92067
60.515	20.950	5.08639	.50733	.06279	-.01995	-.00723	-.00544	-.04339	.44497	.23759	1.87283
63.424	26.183	5.08635	.69927	.05995	-.02655	-.00835	-.00751	-.03911	.60111	.36226	1.65535
66.333	31.442	5.08306	.91775	.05640	-.02815	-.00972	-.00789	-.03534	.75358	.52684	1.43016
67.900	36.698	5.08039	1.14747	.05235	-.03586	-.01113	-.00843	-.03283	.88875	.72770	1.22132
GRADIENT		-.00043	.03955	-.00066	-.00186	-.00024	-.00027	.00093	.02875	.02753	-.03512

RUN NO. 20/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CD	L/D
183.231	15.734	5.08914	.31915	.06650	-.01535	-.00639	-.00261	-.05215	.28916	.19055	1.92064
182.113	20.960	5.08693	.50244	.06236	-.02091	-.00747	-.00558	-.04383	.44487	.23760	1.87237
184.350	26.183	5.08436	.69889	.05947	-.02659	-.00858	-.00740	-.03758	.60093	.36175	1.66120
186.588	31.428	5.08151	.91328	.05620	-.03769	-.00980	-.00803	-.03371	.75170	.52521	1.43125
187.483	36.664	5.08043	1.14796	.05224	-.03588	-.01130	-.00845	-.03283	.88964	.72758	1.22307
GRADIENT		-.00044	.03962	-.00067	-.00189	-.00024	-.00027	.00093	.02882	.02756	-.03510



DATE 04 OCT 73

TABULATED SOURCE DATA, LARC UPWT 1243 (OM-73)

OM-73, UPWT1243,068 (B19CTTSMON19) (M07E23) (VTR5)

(RPV013) ( 04 OCT 73 )

## REFERENCE DATA

SPT = 87.1800 SQ-IN. XMRP = 12.5800 INCHES  
UPWT = 7.1222 INCHES YMRP = .0000 INCHES  
BPT = 14.9900 INCHES ZMRP = 6.0000 INCHES  
SCALE = .0125

## PARAMETRIC DATA

BETA = -5.0000 MACH = 4.8700  
RN/L = 1.720 ELEVTR = -271.0000  
AIRLON = .0000 BDFLAP = -14.2500  
RUEFLR = 40.0000 RUOTER = .0000

RUN NO. 44/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/D
042	15.792	-5.06246	.29091	.06813	.00935	.00480	.00065	.05912	.26331	.14927	1.81254
-162	23.954	-5.06403	.45942	.06399	.01600	.05606	.00385	.05501	.40611	.22412	1.81201
-161	26.231	-5.06699	.64901	.06036	.01865	.00665	.00559	.05371	.55507	.34071	1.83092
.042	31.426	-5.06794	.85337	.05561	.02000	.00761	.00601	.05371	.69919	.49240	1.41999
.042	36.605	-5.06859	1.07069	.04985	.01852	.00985	.00579	.05378	.82884	.67982	1.21355
GRADIENT		-.00033	.03732	-.00086	.00043	.00022	.00023	-.00023	.02726	.02560	-.00022

RUN NO. 45/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/D
09.424	15.743	-5.06810	.29312	.06815	.00934	.00484	.00073	.06073	.26363	.14912	1.81862
71.692	23.957	-5.06521	.46245	.06400	.01549	.00610	.00374	.05634	.40597	.22517	1.81603
72.993	26.212	-5.06606	.64891	.05996	.01863	.00634	.00558	.05369	.55570	.34041	1.83246
69.862	31.469	-5.06899	.85041	.05506	.02147	.00727	.00610	.05527	.69660	.49081	1.41300
68.967	36.702	-5.06978	1.06460	.04945	.01944	.00950	.00588	.05542	.82339	.67591	1.21308
GRADIENT		-.00033	.03683	-.00088	.00050	.00020	.00024	-.00023	.02686	.02532	-.00038

RUN NO. 46/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/D
163.401	15.735	-5.06318	.29013	.06791	.00881	.00489	.00073	.06080	.26386	.14402	1.81125
164.968	23.957	-5.06521	.46240	.06400	.01549	.00610	.00374	.05634	.40892	.22515	1.81525
164.744	26.178	-5.06601	.64594	.05954	.02014	.00621	.00555	.05370	.55342	.33940	1.81541
164.072	31.455	-5.06847	.84730	.05450	.02194	.00705	.00582	.05538	.69435	.48853	1.42102
162.954	36.702	-5.06974	1.06443	.04911	.01940	.00928	.00586	.05542	.82406	.67553	1.21385
GRADIENT		-.00031	.03687	-.00090	.00053	.00019	.00024	-.00023	.02692	.02530	-.00010

ON-70, UFWT1543, ORR (B19C7F3M8N19) (A19TE23) (V705)

(BPV014) (24 OCT 73)

## REFERENCE DATA

SREF = 87.1500 SQ. IN. XREF = 12.5000 INCHES  
 LREF = 7.1222 INCHES YREF = .0000 INCHES  
 BREF = 14.0500 INCHES ZREF = 6.0000 INCHES  
 SCALE = .0150

## PARAMETRIC DATA

BETA = .000 MACN = 4.800  
 RNUL = 1.700 ELEVR = -20.000  
 AILPON = .000 EOP LAP = -14.250  
 EUDFLP = 40.000 FUCER = .000

RUN NO. 47/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CD	L/C
.042	15.759	-.00034	.29094	.06740	.00806	-.00004	-.00087	.00290	.26170	.14398	1.81339
.042	20.954	-.00289	.46039	.06333	.01486	-.00047	-.00108	.00519	.40726	.22376	1.82702
.042	26.199	-.00479	.64697	.05934	.01894	-.00029	-.00148	.00792	.55431	.27358	1.83172
.042	31.445	-.00672	.85151	.05410	.01901	-.00000	-.00073	.00814	.63824	.43037	1.83249
.042	36.690	-.00947	1.06561	.04879	.01756	-.00111	-.00024	.00853	.82559	.67548	1.83202
GRADIENT		-.00037	.03759	-.00091	.00044	-.00003	.00003	.00029	.02710	.02541	-.00019

RUN NO. 48/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CD	L/C
66.281	15.759	-.00034	.29094	.06740	.00806	-.00004	-.00087	.00290	.26170	.14398	1.81339
69.881	20.954	-.00289	.46039	.06333	.01486	-.00047	-.00108	.00519	.40726	.22376	1.82702
66.904	26.199	-.00479	.64697	.05934	.01894	-.00029	-.00148	.00792	.55431	.27358	1.83172
68.985	31.445	-.00672	.85151	.05410	.01901	-.00000	-.00073	.00814	.63824	.43037	1.83249
70.532	36.690	-.00947	1.06561	.04879	.01756	-.00111	-.00024	.00853	.82559	.67548	1.83202
GRADIENT		-.00042	.03707	-.00090	.00044	-.00003	.00003	.00029	.02710	.02541	-.00019

RUN NO. 49/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CD	L/C
163.651	15.759	-.00032	.29093	.06739	.00805	-.00010	-.00087	.00229	.26169	.14387	1.81330
164.745	20.984	-.00288	.45736	.06317	.01637	-.00063	-.00110	.00522	.40441	.22276	1.81943
164.299	26.196	-.00481	.64416	.05880	.01844	-.00040	-.00150	.00756	.55210	.27372	1.82919
164.075	31.420	-.00725	.84845	.05361	.01949	-.00092	-.00050	.00813	.69809	.48875	1.82629
163.628	36.690	-.00999	1.06566	.04905	.01756	-.00117	-.00001	.00847	.82582	.67525	1.82370
GRADIENT		-.00045	.03711	-.00082	.00042	-.00003	.00003	.00029	.02715	.02540	-.00024



TABULATED SOURCE DATA, LARC UPME 1243 (OA-75)

DATE 09 OCT 75

OA-75, UPMT1243, OFB (B19CTF346N19) (RADTE23) (VTR5)

(RPV015) ( 04 OCT 75 )

PARAMETRIC DATA

REFERENCE DATA

REF = 87.1501 50. IN. XMRP = 12.5800 INCHES  
LREF = 7.1222 INCHES YMRP = .0000 INCHES  
BREF = 14.0500 INCHES ZMRP = 6.0000 INCHES  
SCALE = .0155

BETA = 5.000 MACH = 4.600  
PNL = 1.720 ELEVTR = -27.000  
ATLUPON = .000 BOFLAP = -14.250  
FUELR = 40.000 FUELER = .000

RUN NO. 50/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/D
268	15.763	5.11032	.28907	.06772	.00603	-.00478	-.00232	-.05458	.25980	.4371	1.80781
.244	20.983	5.10963	.45863	.06348	.01433	-.00620	-.00520	-.04778	.40549	.22350	1.81424
.244	26.182	5.10743	.64244	.05976	.01757	-.00737	-.00729	-.04128	.55016	.33708	1.63212
.244	31.438	5.10493	.84398	.05513	.02000	-.00878	-.00732	-.03832	.69133	.48723	1.41890
.244	36.683	5.10206	1.05812	.04937	.01835	-.01008	-.00788	-.03625	.81907	.67169	1.21941
GRADIENT		-.00241	.03679	-.00086	.00055	-.00025	-.00025	.00087	.02686	.02524	-.00302

RUN NO. 51/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/D
79.088	15.744	5.11042	.28811	.06756	.00821	-.00518	-.00235	-.05459	.25704	.14286	1.80179
72.102	20.948	5.10870	.45575	.06338	.01371	-.00655	-.00361	-.04611	.40297	.22213	1.81413
66.732	26.204	5.10544	.64245	.05961	.01751	-.00761	-.00718	-.03977	.55010	.33716	1.63159
68.969	31.437	5.10394	.84389	.05510	.01893	-.00879	-.00770	-.03725	.69123	.48712	1.41902
79.983	36.681	5.10107	1.05791	.04935	.01728	-.01009	-.00826	-.03458	.81894	.57133	1.21950
GRADIENT		-.00245	.03689	-.00085	.00045	-.00023	-.00027	.00093	.02697	.02526	-.00300

RUN NO. 52/ 0 RN/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/D
181.814	15.736	5.10902	.28620	.06763	.00917	-.00550	-.00249	-.05307	.25713	.14271	1.80175
164.076	20.948	5.10870	.45572	.06337	.01371	-.00655	-.00361	-.04610	.40294	.22211	1.81413
162.286	26.182	5.10547	.64248	.05941	.01750	-.00772	-.00718	-.03978	.55039	.33679	1.63411
188.498	31.431	5.10387	.84105	.05470	.01946	-.00874	-.00771	-.03719	.69237	.48489	1.42171
162.266	36.669	5.10104	1.05489	.04913	.01882	-.01014	-.00827	-.03452	.81679	.69319	1.22022
GRADIENT		-.00240	.03675	-.00087	.00048	-.00022	-.00026	.00088	.02687	.02516	-.00304

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TABULATED SOURCE DATA, LARC UPNT 1043 (0A-70)

0A-70, UPNT1043,0FB (0A19CTFSM0A119) (0A10T23) (VFR3)

(8P4718) ( 24 OCT 73 )

## REFERENCE DATA

SPEF = 87.1593 CAL-IN. XMRP = 12.5800 INCHES  
 LREF = 7.1222 INCHES YMRP = .0000 INCHES  
 BREF = 14.0500 INCHES ZMRP = 6.0000 INCHES  
 SCALE = .0150

BETA = -5.0000 MACH = 4.8000  
 RV/L = 1.7200 ELEVTR = -40.0000  
 ATURON = .0000 EFLAP = -14.2500  
 FUFLR = 40.0000 RUDDER = .0000

## PARAMETRIC DATA

RUN NO. 31/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CC	L/D
.112	15.786	-5.06203	.28109	.07381	.01336	.00456	.00056	.00935	.25046	.14741	1.59312
.112	20.961	-5.06468	.45045	.06879	.02154	.00584	.00382	.05511	.39803	.22537	1.75723
.112	26.207	-5.06616	.63711	.06483	.02880	.00661	.00295	.05224	.54299	.33932	1.53328
.112	31.430	-5.06709	.83972	.06112	.02868	.00758	.00636	.02230	.68379	.48351	1.39659
.112	36.675	-5.06785	1.04973	.05679	.03021	.00960	.00512	.02245	.80798	.67251	1.20343
GRADIENT		-.00027	.03683	-.00080	.00070	.00023	.00026	-.00032	.02533	.02514	-.02594

RUN NO. 32/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CC	L/D
71.943	15.724	-5.06297	.27833	.07348	.01484	.00453	.00055	.00943	.24800	.14616	1.59331
70.378	20.992	-5.06468	.45052	.06826	.02155	.00585	.00383	.05513	.39617	.22513	1.75373
72.614	26.182	-5.06618	.63434	.06437	.02631	.00650	.00594	.05230	.54786	.33783	1.67187
72.166	31.392	-5.06711	.83309	.06009	.02965	.00720	.00632	.05242	.67395	.48525	1.47103
71.495	36.674	-5.06883	1.04733	.05631	.03170	.00931	.00571	.05418	.82672	.67167	1.20157
GRADIENT		-.00031	.03671	-.00081	.00080	.00021	.00024	-.00025	.02676	.02503	-.02578

RUN NO. 33/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	CN	CA	CLM	CBL	CYN	CY	CL	CC	L/D
165.035	15.786	-5.06355	.28125	.07356	.01537	.00457	.00044	.00997	.25070	.14717	1.70351
164.587	20.931	-5.06468	.45073	.06836	.02158	.00585	.00383	.05513	.39656	.22487	1.76353
164.353	26.172	-5.06617	.63431	.06409	.02630	.00645	.00593	.05228	.54101	.33729	1.60398
163.915	31.426	-5.06913	.83303	.05989	.03066	.00712	.00645	.05393	.67951	.48545	1.39393
163.468	36.674	-5.06927	1.04717	.05618	.03163	.00887	.00592	.05413	.80633	.67049	1.20259
GRADIENT		-.00030	.03658	-.00083	.00079	.00019	.00026	-.00028	.02664	.02493	-.02512

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TABULATED SOURCE DATA, LARC UPMT 1243 (0A-75)

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0A-75, UPMT1043,0FB (B59C7F5M6N19: N107E23) (V7R3)

(PPV317) ( 04 OCT 75 )

## REFERENCE DATA

SPDF = 07.1500 SQ. IN. XMPF = 12.5000 INCHES  
 LDEF = 7.1222 INCHES YMPF = .0000 INCHES  
 BDEF = 14.0000 INCHES ZMPF = 6.0000 INCHES  
 SCALE = .0125

BETA = .000 MACH = 4.000  
 RNUL = 1.720 ELEVTR = -40.900  
 AIRDRN = .000 BDFLAP = -14.250  
 RUFLR = 43.000 RUDESR = .000

## PARAMETRIC DATA

RUN NO. 34/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CY	CL	CD	L/D
-117	15.751	.01910	.27619	.07371	.01574	.00024	-.00023	.00086	.24581	.14592	1.68461
.107	20.959	.01656	.44861	.06890	.02259	-.00002	-.00043	.00371	.39442	.22444	1.73737
.106	26.196	.01470	.63954	.06426	.02723	.00011	-.00084	.00641	.54190	.33822	1.61223
.106	31.441	.01377	.83703	.05932	.02989	-.00013	-.00058	.00650	.68700	.48757	1.47023
-118	36.677	.01102	1.03109	.05602	.03052	-.00037	-.00007	.00714	.87953	.67273	1.27532
GRADIENT		-.00036	.03704	-.00084	.00071	-.00003	.00000	.00030	.02756	.02517	-.02532

RUN NO. 35/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CY	CL	CD	L/D
71.938	15.729	.01910	.27615	.07355	.01574	.00024	-.00023	.00086	.24589	.14583	1.68845
73.371	20.959	.01654	.44867	.06785	.02212	.00004	-.00043	.00372	.39471	.22365	1.76327
69.924	26.181	.01467	.63249	.06369	.02667	.00000	-.00085	.00646	.53950	.33622	1.67461
69.924	31.393	.01325	.83414	.05968	.02932	-.00036	-.00035	.00680	.68794	.48545	1.47271
69.924	36.639	.01045	1.04597	.05517	.03043	-.00071	.00015	.00721	.80542	.68822	1.27532
GRADIENT		-.00039	.03677	-.00086	.00070	-.00004	.00002	.00030	.02687	.02499	-.02537

RUN NO. 36/ 0 RNUL = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEB	CYN	CY	CL	CD	L/D
163.029	15.751	.01908	.27618	.07338	.01576	.00000	-.00023	.00087	.24580	.14559	1.68891
164.356	20.930	.01654	.44871	.06772	.02213	.00004	-.00043	.00372	.39491	.22354	1.76865
163.666	26.181	.01369	.63264	.06384	.02770	-.00002	-.00025	.00612	.53968	.33822	1.67513
163.462	31.428	.01282	.83689	.05942	.02876	-.00003	-.00011	.00665	.68312	.48709	1.47048
163.015	36.674	.00998	1.04808	.05539	.03097	-.00076	.00040	.00718	.87752	.67704	1.27432
GRADIENT		-.00042	.03691	-.00095	.00071	-.00005	.00003	.00029	.02696	.02510	-.02550

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TABULATED SOURCE DATA, LARC UPLT 1543 (0A-73)

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CA-73, UPLT 1543, ORB (819C77FJW4N19) (4207E23) (N7P3)

(RPW318) (24 OCT 73)

## REFERENCE DATA

SREF = 87.1500 50.1N. YREF = 12.5800 INCHES  
 LREF = 7.1222 INCHES YREF = .0700 INCHES  
 DREF = 14.0570 INCHES ZREF = 6.0700 INCHES  
 SCALE = .0150

BETA = 5.000 MACH = 4.800  
 FVL = 1.700 ELEV = -40.000  
 ATLSON = .000 DFLAP = -14.250  
 RUOTUP = 40.000 FLUCOR = .000

## PARAMETRIC DATA

RUN NO. 37/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/C
328	15.749	5.11136	.27720	.07005	.01297	-.00409	-.00128	-.05769	.24897	.14934	1.69048
102	25.946	5.10972	.44697	.06861	.02056	-.00599	-.00464	-.04926	.39291	.22386	1.75518
386	26.168	5.10736	.63582	.06439	.02484	-.00681	-.00663	-.04278	.53777	.33597	1.67769
101	31.433	5.10532	.82560	.06173	.02989	-.00800	-.00769	-.04033	.67618	.48447	1.59371
101	36.686	5.10204	1.03782	.05988	.03133	-.00949	-.00720	-.03787	.79380	.66491	1.47019
GRADIENT		-.00044	.03636	-.00380	.00068	-.00024	-.00027	.00093	.02649	.02483	-.02582

RUN NO. 39/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/C
69.471	15.722	5.10989	.27739	.07291	.01398	-.00422	-.00141	-.05810	.24717	.14932	1.70078
70.142	25.946	5.10824	.44692	.06823	.02053	-.00572	-.00468	-.04767	.39299	.22391	1.75528
69.919	26.177	5.10607	.63079	.06431	.02483	-.00683	-.00677	-.04119	.53772	.33599	1.67743
69.693	31.419	5.10447	.82657	.06052	.03038	-.00796	-.00729	-.03861	.67593	.48253	1.59448
69.693	36.686	5.10106	1.03773	.05830	.03129	-.00920	-.00758	-.03601	.79854	.66511	1.47741
GRADIENT		-.00041	.03627	-.00378	.00085	-.00023	-.00029	.00094	.02640	.02479	-.02571

RUN NO. 40/ 0 RV/L = 1.72 GRADIENT INTERVAL = -5.00/ 5.00

PO-JET	ALPHA	BETA	ON	CA	CLM	CEL	CYN	CY	CL	CD	L/C
163.010	15.741	5.10990	.27724	.07244	.01399	-.00427	-.00141	-.05611	.24719	.14894	1.70548
163.450	25.943	5.10821	.44403	.06798	.02103	-.00577	-.00469	-.04763	.39042	.22221	1.75598
163.906	26.188	5.10610	.63088	.06425	.02479	-.00705	-.00679	-.04121	.53777	.33507	1.67715
164.352	31.429	5.10294	.82664	.06011	.02932	-.00798	-.00742	-.03751	.67402	.48233	1.59741
164.576	36.650	5.09897	1.03497	.05654	.03177	-.00928	-.00746	-.03442	.79560	.66316	1.47019
GRADIENT		-.00032	.03629	-.00376	.00084	-.00023	-.00026	.00103	.02642	.02479	-.02517

NASA-MSEC-WAF